

# Center for Space Telemetry and Telecommunications Research

## Program Review

NASA Grant: NAG 5-1491

### Prepared by:

Stephen Horan  
William Ryan  
James LeBlanc  
Phillip De Leon  
Sheila Horan  
Tom Shay

New Mexico State University  
Klipsch Department of Electrical and Computer Engineering  
Box 30001 - Dept. 3-0  
Las Cruces, New Mexico 88003

0154

001 352

# Program Review: Telemetry and Telecommunications Research

Stephen Horan

Director, Telemetry and  
Telecommunications Program

March 11, 1997

Program Overview

# Program Overview

- Topics
  - NMSU Background
  - Telemetry and Telecommunications Program
  - Grant History
  - Faculty & Staff
  - Facilities
  - Review Program

# NMSU Background

- NMSU is the Land Grant University and Space Grant University
- Federally-designated Minority University
- Carnegie-I Research University
- Statistics (Fall 1996):
  - enrollment = 14748 students
  - ABET-accredited College of Engineering

# NMSU Background

- Student Ethnicity
  - African American 2%
  - American Indian 3%
  - American Oriental 1%
  - Hispanic 34%
  - Other 60%

# Telemetry & Telecommunications Program

- Senior-level courses in Analog & Digital Communications
- Graduate-Level courses in Communications Theory, Digital Communications, Coding (Channel & Source), Satellite Communications, Telemetry Systems
- MS EE and Ph.D. degree programs

# Telemetry & Telecommunications Program

- Full-time & part-time students, off-campus programs at HAFB, KAFB, NTU
- Average 5-6 MSSEE degrees and 2 Ph.D. degrees awarded each year
- Research Programs with
  - NASA (Telemetry & Telecommunications, NASA Space Grant, ACTS Propagation Experiment)

# Telemetry & Telecommunications Program

- Research Programs (cont.)
  - NSF (tape recording technology)
  - Rome Labs (signal processing)
- Chaired Professorship in Telemetry and Telecommunications funded by IFT, State of New Mexico, and industry
- Designated Center of Excellence in Telemetry Systems by the IFT



# Grant History

- Major research funding comes from NASA NAG 5-1491
  - Continuous since 1990
  - Frank Carden and William Osborne were previous PIs
- Related funding from
  - NSF, Sandia, IFT, Rome Labs

# Faculty & Staff

- Faculty
  - Stephen Horan, Director
  - William Ryan, Associate Director
  - Sheila Horan
  - Phillip DeLeon
  - James LeBlanc
  - Thomas Shay

# Faculty & Staff

- Staff
  - Janice Apodaca, Secretary
  - Lawrence Alvarez, Technician
- Students
  - 4 Undergraduates
  - 10 Graduate Students

# Facilities

- Faculty Offices in Thomas & Brown Hall and Goddard Hall
- Student Offices in Thomas & Brown Hall and Goddard Hall
- Telemetering Center is a central suite
  - Director's Office
  - Research Laboratory
  - Secretary
  - Technician

# Facilities

- Laboratory
  - hardware development and testing area
  - software simulation area
- Future (hopefully)
  - Telemetry and Telecommunications Program to take over the majority of Goddard Hall (present Engineering Technology Department)

# Facilities

- Presently developing IFT-funded academic communications and signal processing laboratory
  - 5 PCs
  - Labview
  - Satellite Tool Kit
  - Matlab
  - radio equipment

# Review Program

- 8:30-9:00 Introductions
- 9:00-9:30 - Program Overview
- 9:30-10:30 - Small Satellite & Related Work
- 10:30-10:45 - Break
- 10:45-12:00 - Bandwidth-Efficient Modulation Techniques
- 12:00-1:30 - Lunch
- 1:45-2:30 - Coding Techniques
- 2:30-3:00 - DAMA Development
- 3:00-3:15 - Break
- 3:15-4:00 - Tours & Demonstrations
- 4:00-5:00 - Wrap-up Review
- 5:00- Adjourn

# Small Satellite Technology

Stephen Horan, Phillip DeLeon, and  
Thomas Shay

March 11, 1997

Small Satellite



# Small Satellite Access of the Space Network

- Topics
  - Purpose
  - Activities
  - Current Year Highlights
  - Plans for Next Year
  - Future Plans

# Purpose

- To assist the small satellite community in utilizing the SN for communications services rather than proprietary ground stations
  - reduce costs associated with communications system design
  - assist users in gaining access when high-priority users are being supported

# Activities

- **Orbital Access**
  - predicting when a fixed antenna system on a user satellite can access a TDRS
  - evaluation of communications performance
- **Spacecraft Testing**
  - testing communications concepts with on-orbit spacecraft

# Activities

- DAMA
  - Develop concept for how on-demand services might be scheduled
  - Develop concept for Doppler tracking using advanced DSP concepts
- RF Testbed
  - Develop RF satellite simulation testbed from NMSU

# Current Year Highlights

- *Orbital Access*
  - looked at 600 km - 1200 km orbits over a 30-day period using Satellite Tool Kit to improve methodology from “home-brew” simulations of the previous years
  - Added TDRS-Z to coverage predictions
  - Use SNUG Rev. 7 data to evaluate data throughput

# Orbital Access

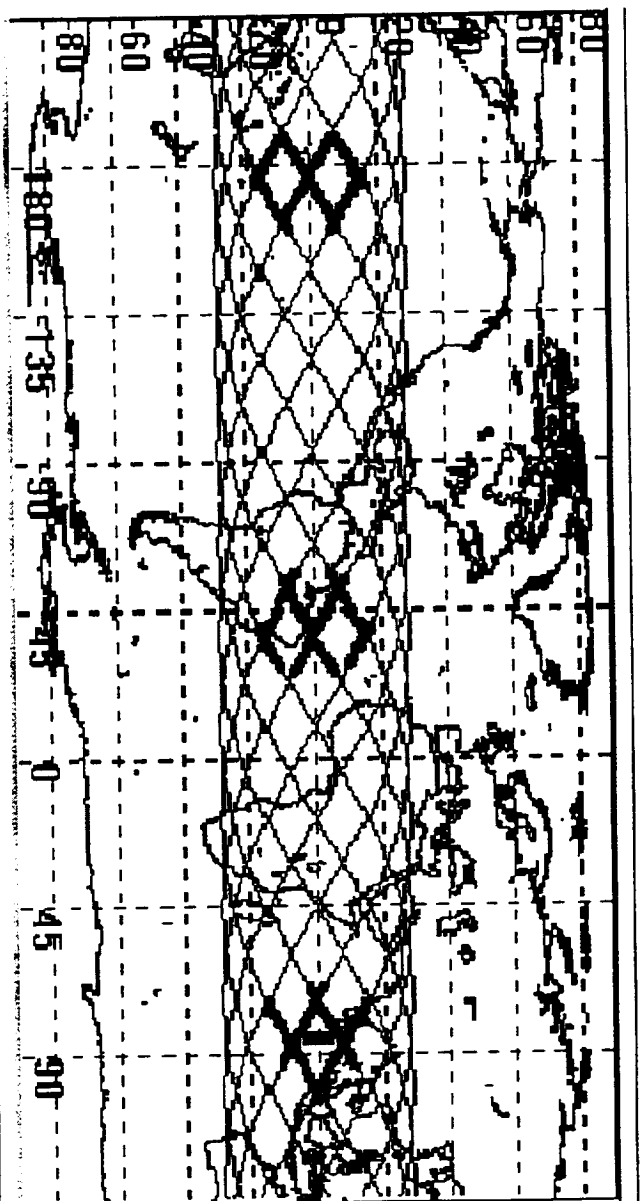
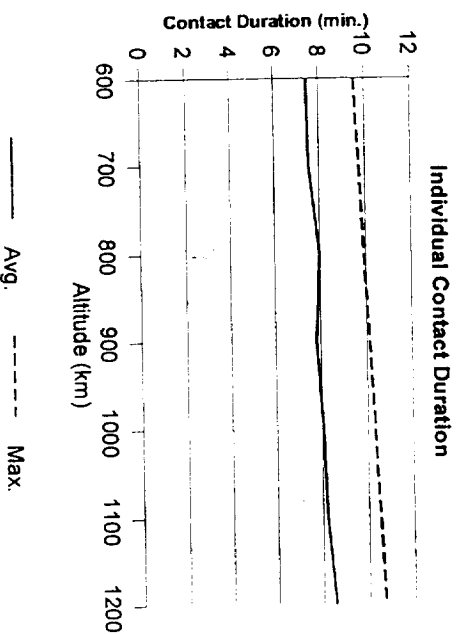
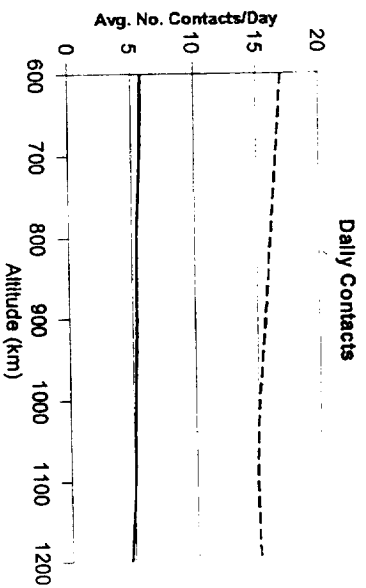
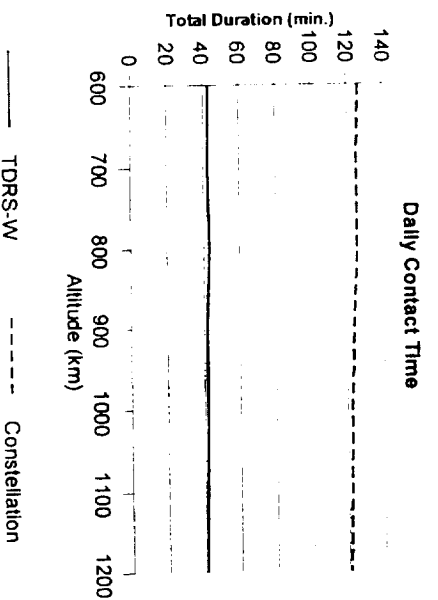


Figure 1 - Simulated 24-hour ground track for spinning-satellite contact with the Space Net-work. Highlighted areas along the ground track near the TDRS positions at -174°, -41°, and +85° longitude show times when contact is possible.

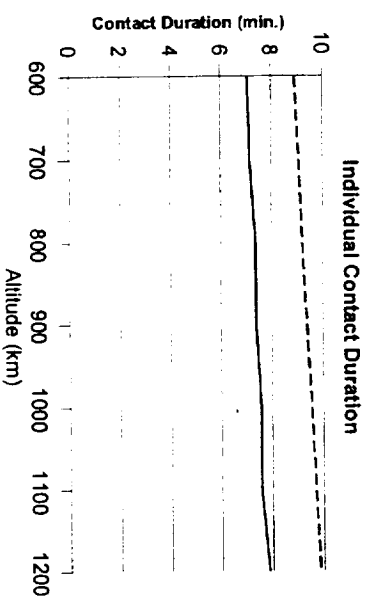
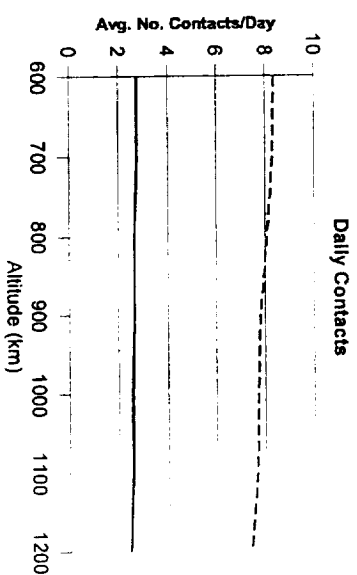
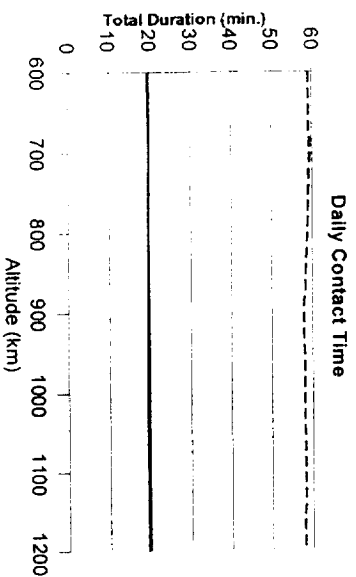
# Orbital Access - 28.5 deg.



March 11, 1997

Small Satellite

# Orbital Access - sun synch.



— Avg.    - - - - - Max.

— TDRS-W    - - - - - Constellation

— TDRS-W    - - - - - Constellation

March 11, 1997

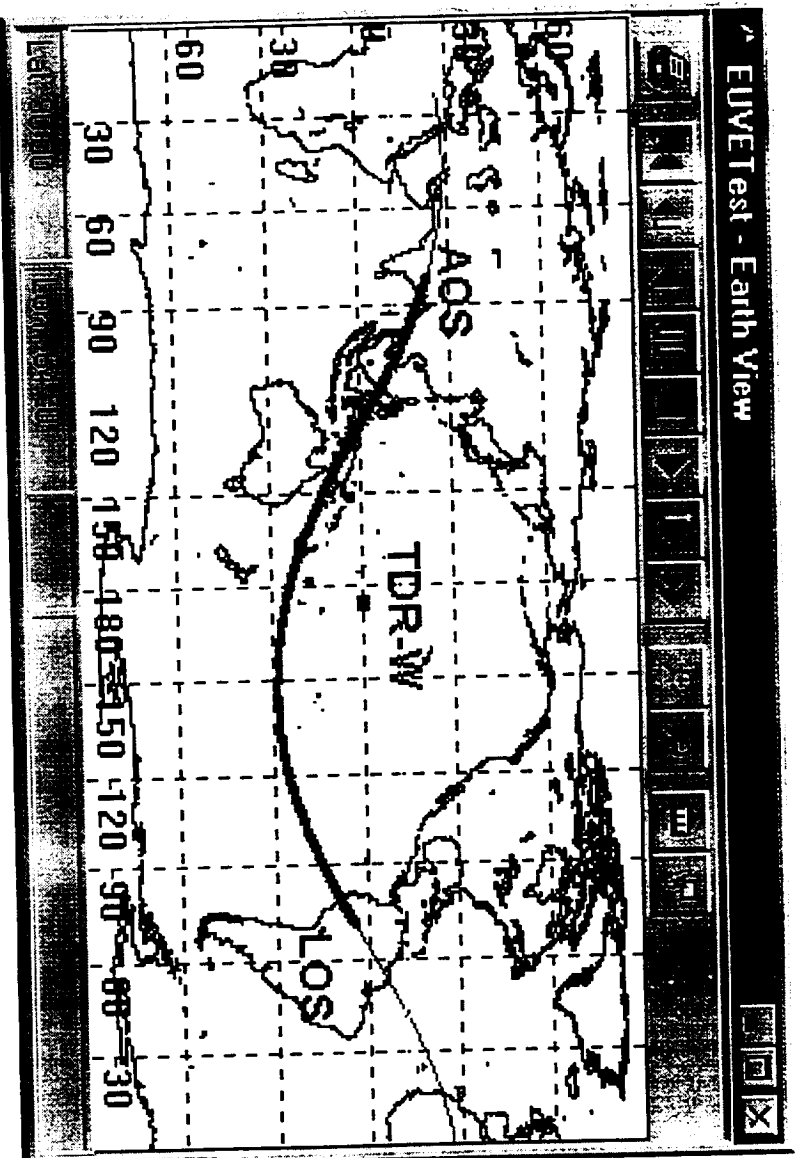
Small Satellite



# Current Year Highlights

- EUVE Testing
  - 6 passes used in May 1996
  - found inertially-pointed satellites can remain pointed at a TDRS for up to 50 minutes (30 minutes were observed, 50 minutes simulated)
  - Used STK to simulate passes
    - emulated orbital contact
    - predicted antenna performance due to space loss and pattern gain

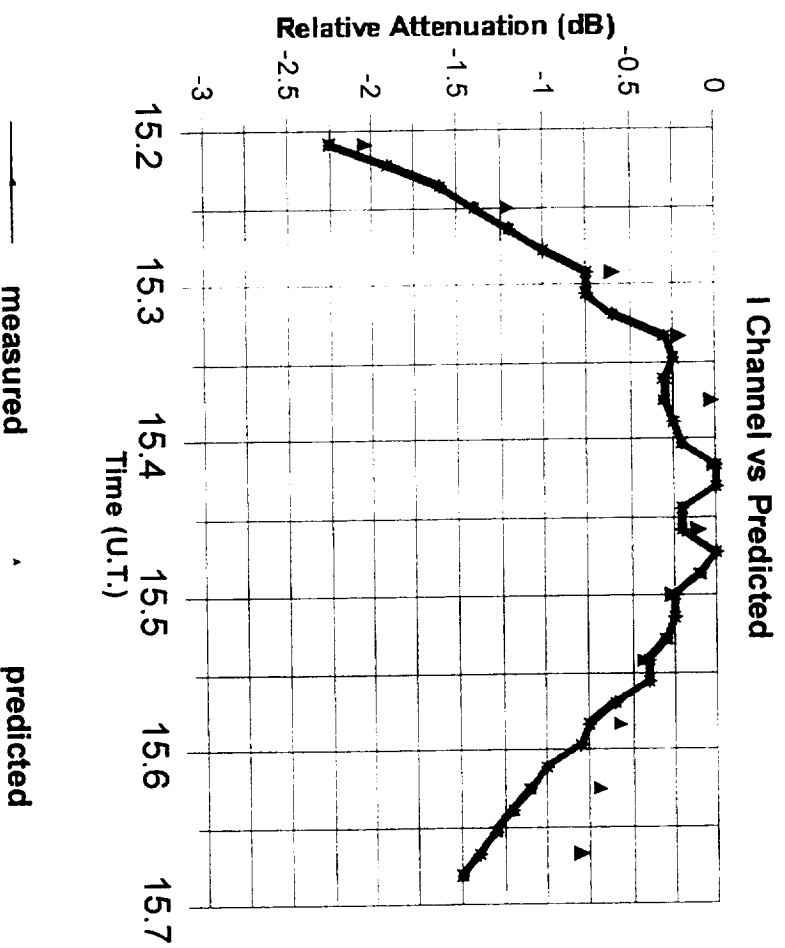
# EUVE Testing



March 11, 1997

Small Satellite

# EUVE Testing



# Current Year Highlights

- DAMA Approach
  - Presented baseline approach at the International Telemetry Conference
  - Began work on Doppler tracking methodology in the carrier recovery subsystem
  - Began investigation of how the interface to the WSC beam-forming subsystem might be accomplished

# Current Year Highlights

- RF Testbed
  - began development of requirements for NMSU portion of the system
    - S-Band Transmission
    - helical antenna
    - DG1 Mode 2 transmission
  - began discussions with engineering staff at WSC to insure compatibility of the concept and design

# Plans for Next Year

- Follow-on Spacecraft Testing
  - would like to re-try fixed-antenna pointing scenario with orbiting satellite
  - working with NASA to identify candidate satellite
  - if found, conduct experiment EUVE-type again

# Plans for Next Year

- DAMA Development
  - Complete development of Doppler tracking algorithm
  - Demonstrate its use in the laboratory
  - Complete development of interface requirements to beam forming subsystem at the WSC

# Plans for Next Year

- RF Testbed
  - complete requirements definition and review
  - design with engineers who can critically review
  - begin acquisition and installation of equipment
  - prepare for testing



# Plans for Next Year

- Hitchhiker Proposal
  - Develop payload concept in detail
  - Select suite of experiments from current NMSU technology development efforts
  - Complete Customer Payload Requirements documentation and submit to NASA

# Hitchhiker Payload

- Plan to develop a suite of experiments to test communications concepts
- Will need to develop requirements for each
  - non-gimbaled antenna pointing
  - DAMA request message
  - passive telemetry via laser communications
  - Turbo Code test

# Future Plans

- DAMA Testing
  - Develop hardware for WSC interface
  - Develop transmitter with “Doppler” offset to carrier
  - Run test signals from NMSU through TDRS to WSC to verify concept

# Future Plans

- RF Testbed
  - Develop capability to test
    - turbo codes
    - bandwidth-efficient modulation
    - small satellite communications
    - packet communications
  - Look for potential industry partners for technology development

# Future Plans

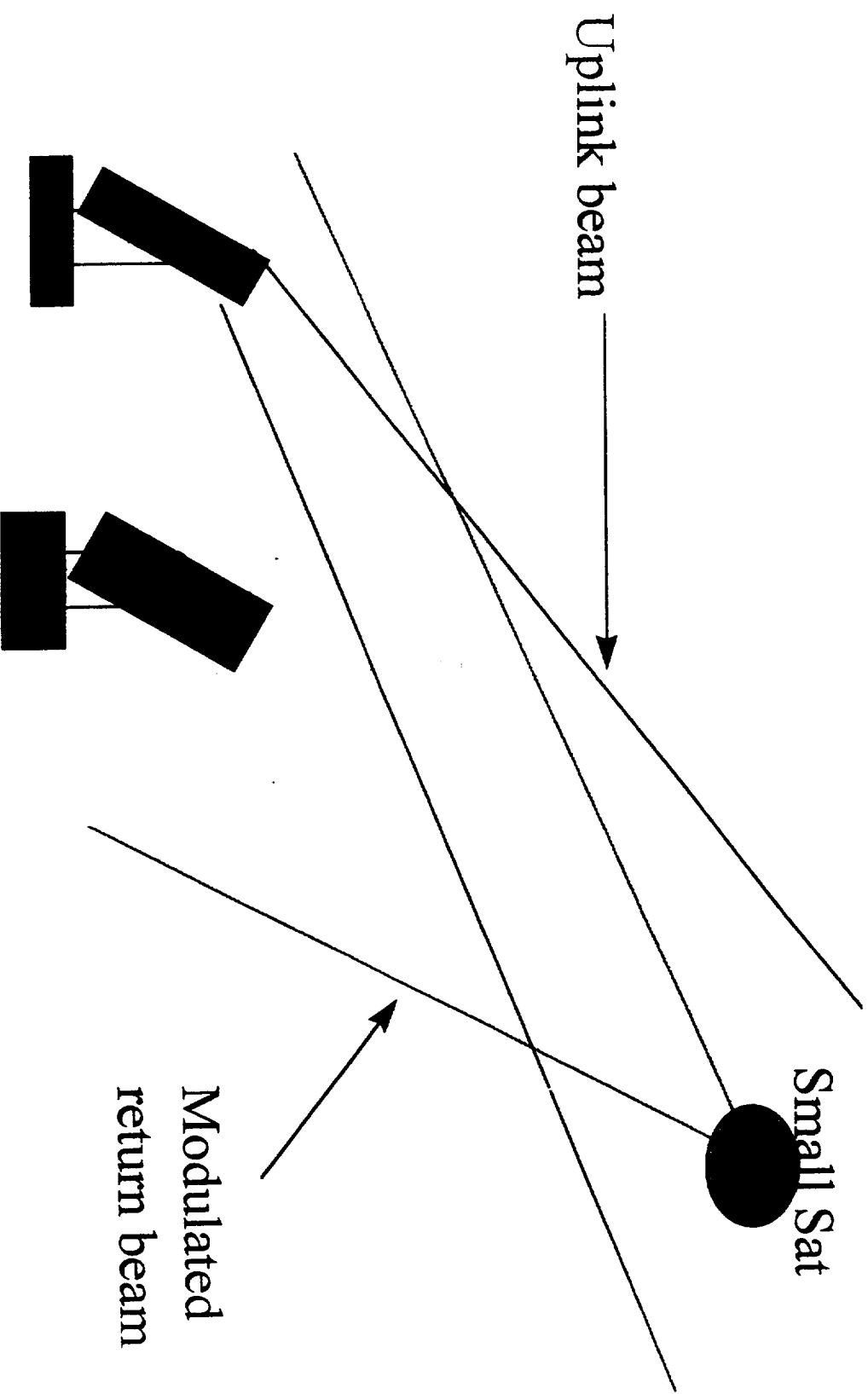
- Hitchhiker Program
  - have proposed project accepted by NASA
  - develop suite of experiments
  - develop partners for technology demonstrations; so far, we are looking at
    - Phillips Laboratory
    - WSMR

# Laser Com Activity

by

T. M. Shay

# LIGHT-WEIGHT LOW DATA RATE OPTICAL COM



# ADVANTAGES

- Very Lightweight and very low power consumption.
- Operation in full daylight will be possible using a FADOE on the ground based receiver
- Relatively low cost ground station



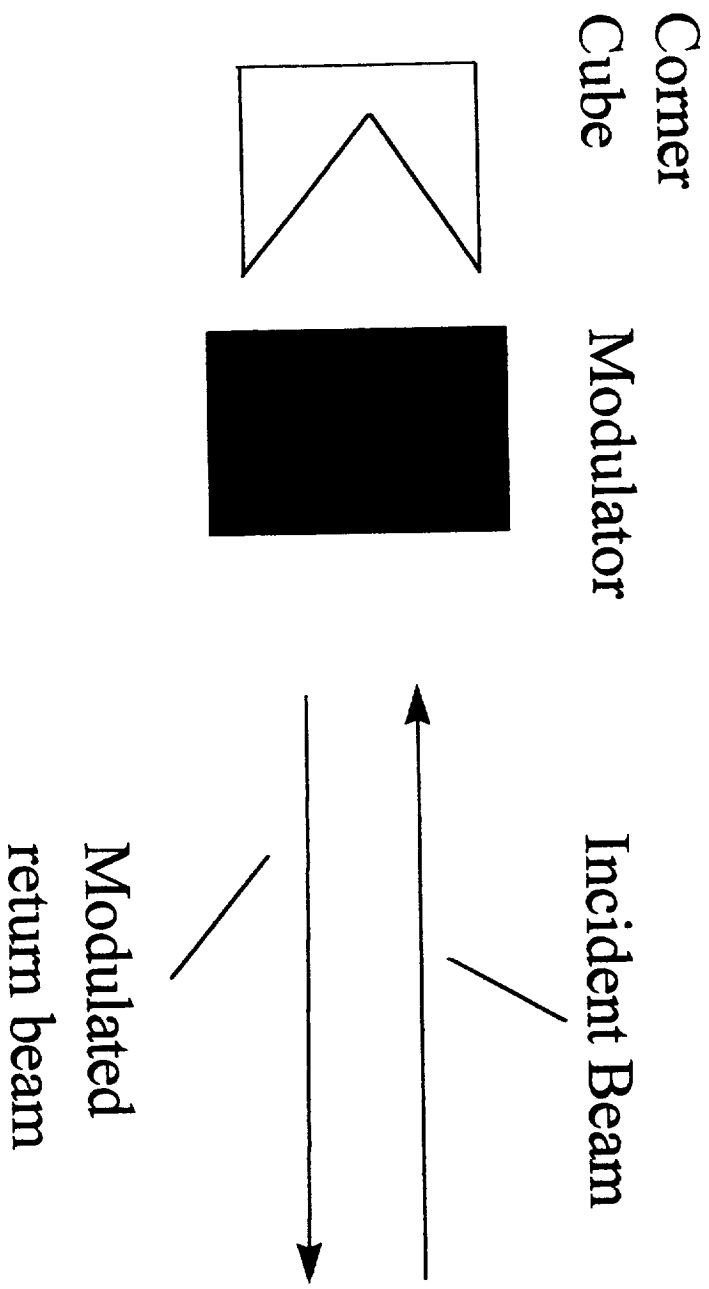
# PLANS FOR NEXT YEAR

- Simple model to estimate system performance and tradeoffs. (Report Sept, 97)
- Design laboratory demonstration unit. (Report April, 98)
- Build and test laboratory demonstration unit. (Report July 1, 98)

# LIGHTWEIGHT LOW DATA RATE OPTICAL COM POSSIBILITIES

Data Rate kb/s	Receiver Diameter	Modulator Size	Laser Power mW	Comments
10	1.8 m	4" x 4"	150	Demonstration System Using Existing Tracking Scopes
1	1.8 m	4" x 4"	45	
10	16 in	8" x 10"	100	Potential Fielded System
1	16 in	8" x 10"	40	

# Satellite Components



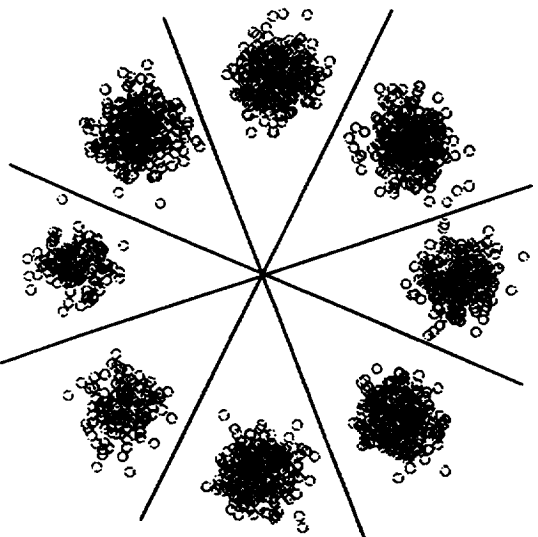
# LONG TERM PLAN

- Year I. Model performance and lab test.
- Year II. Perform field experiment (1/2 km)
- Year III. Build and test satellite. (Needs additional funding)
- Year IV. Fly satellite and test satellite laser com. (Needs additional funding)

# PROGRESS THIS YEAR

- Fast optical logic model.
  - Quantum Mechanical Theory Development on schedule.
- Stark Anomalous Dispersion Optical Filter.
  - Signal Oscillator Designed and testing has begun. This is also on schedule.
- This work will be completed this year.

# 8PSK SIGNALING OVER NON-LINEAR SATELLITE CHANNELS



Sheila Horan

Rubén Caballero, Jianping Tao, Josua Purba

(supported by a grant from NASA #NAG 5-1491)

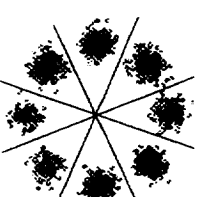
New Mexico State University

Department of Electrical and Computer Engineering  
Space & Telecommunications Center

3/10/97

# ***PRESENTATION***

---



8PSK OVER  
NON-LINEAR  
SATELLITE CHANNELS

## 1- Introduction

## 2- Approach

## 3- SPW Simulations

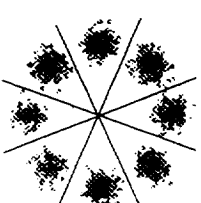
- End-to-End Functional System Diagram and Description

## 4- Results for 8PSK Modulation

- Power Containment and Spurious Emissions
- End-to-End System Performance
- Non-Constant Envelope

## 5- Conclusions & Further Work

## *(2)-APPROACH (cont.)*



8PSK OVER  
NON-LINEAR  
SATELLITE CHANNELS

- Software used: - SPW on Sun Spare 10 Unix Station
  - Matlab on PC (Windows) and Unix
- Block Diagram Created on SPW to perform simulations
- Measurements:
  - 1 - Measure and Plot: Power Spectral Density PSD  
( calculate Utilization Ratio, at -50dB point)
  - 2 - Measure Symbol Error Rate (BER)
    - BER plotted as function of symbol SNR ( $E_b/N_0$ )
    - Baseband Filter Optimization:  
BT product selected such that filter ISI Loss < 0.4dB  
at  $10^{-3}$
  - 3 - Non-Constant Envelope Measurements
    - plot Average Symbol Variance vs. Bandwidth



# ***(1) - INTRODUCTION***

---



•SFCG-12 (Australia, Nov 92) requested the study and comparison (CCSDS RF) of various modulation schemes in:

- BW needed;
- power efficiency;
- spurious emissions; and
- interference susceptibility.

•Why this study?

- frequency bands are becoming more and more congested; and
- space agencies are under constant pressure to reduce costs;
  - power efficiency.

•3 Phases

Phase I : BW utilization of various modulation schemes

Phase II : Effect of Spectrum Shaping

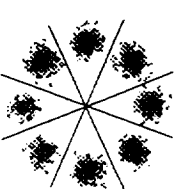
Phase III: End-to-End Performance with different  
Modulation Schemes

3/10/97

### ***(3) - SPW SIMULATIONS***

#### ***8PSK Over Non-Linear Channel***

---



8PSK OVER  
NON-LINEAR  
SATELLITE CHANNELS

#### SPW Simulation Variables:

- Bit rate =  $R_B = 1$  bps ( $R_S =$  Symbol Rate =  $(1/3)$  symbol/s ;
- Sample Rate =  $f_s = 16$  samples/sec;
- Data: NRZ-L;
- Carrier Frequency = 0 Hz (Baseband Simulations);

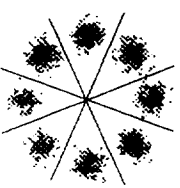
#### Transmitting System:

- Data Generator: Ideal and Non-Ideal Data;
  - data asymmetry = 2%;
  - data imbalance = 0.45;
- Baseband Filters do not include resistive and reactive losses;
- Power Amplifier based on:
  - ESA 10-watt Solid State Amplifier (SSPA);

### **(3) - SPW SIMULATIONS**

#### ***8PSK Over Non-Linear Channel (cont.)***

---



8PSK OVER  
NON-LINEAR  
SATELLITE CHANNELS

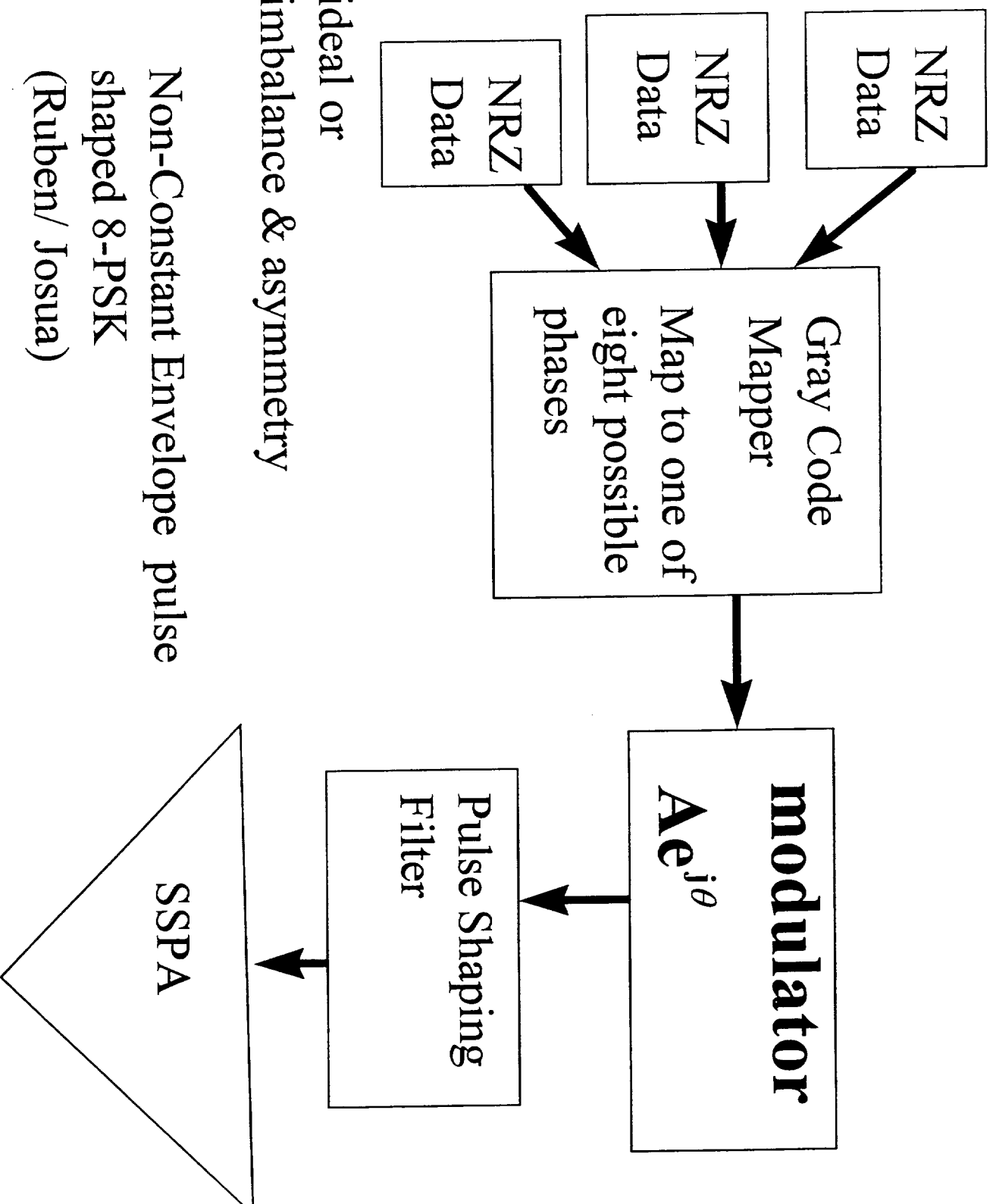
#### **Receiving System:**

- Matched Filter: Sliding Integrator;  
SRRC matched filter used for non-constant envelope
- Delay and Phase Meter (Synchronization); and
- Error Rate Estimator.

#### **Baseband Filters:**

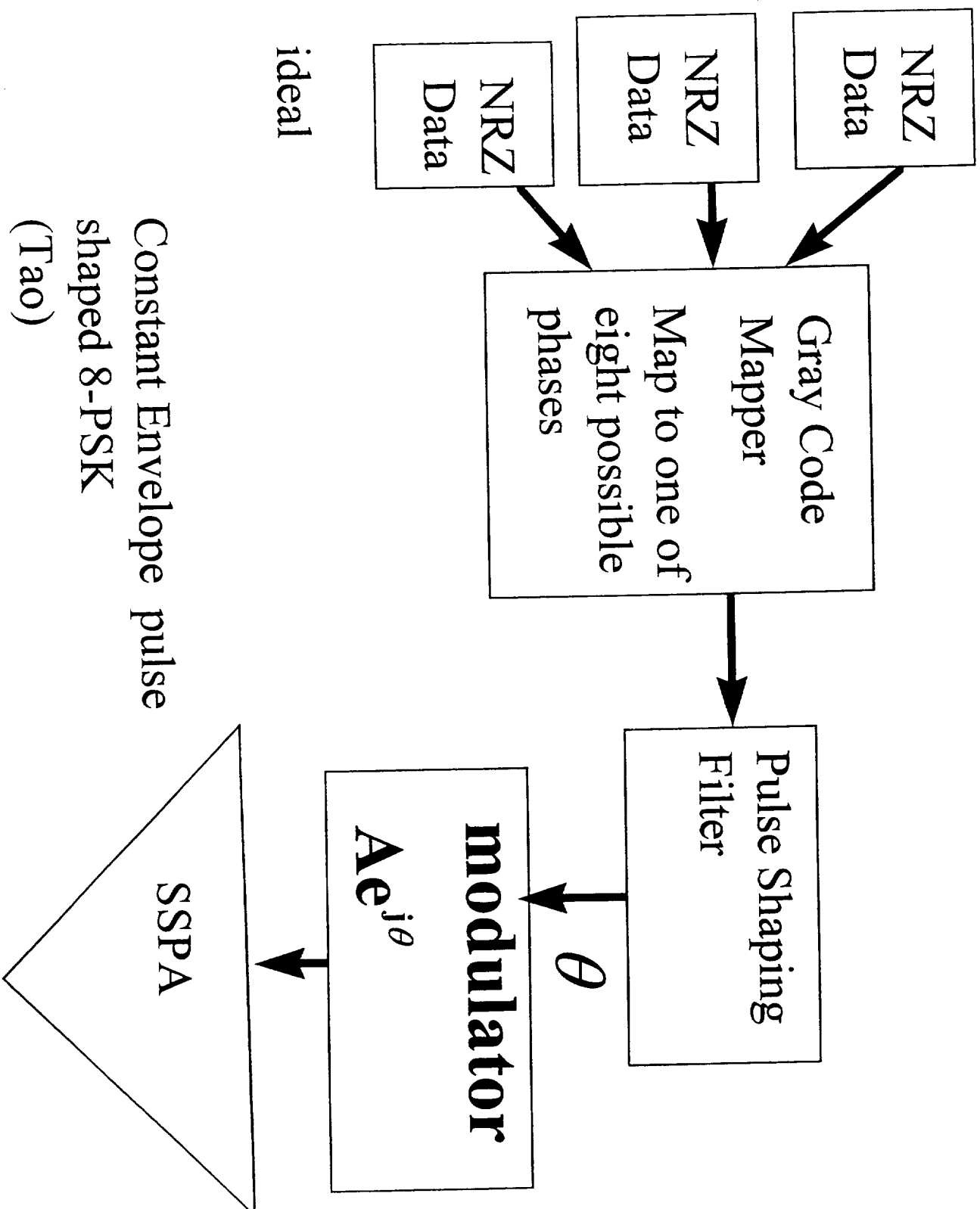
From Phase II :

- Butterworth 5th Order (BT=1,2,3);
- Bessel 3rd Order (BT=1,2,3);
- Square Root Raised Cosine (SRRC)  
 $\alpha = 1$  and 256 taps;

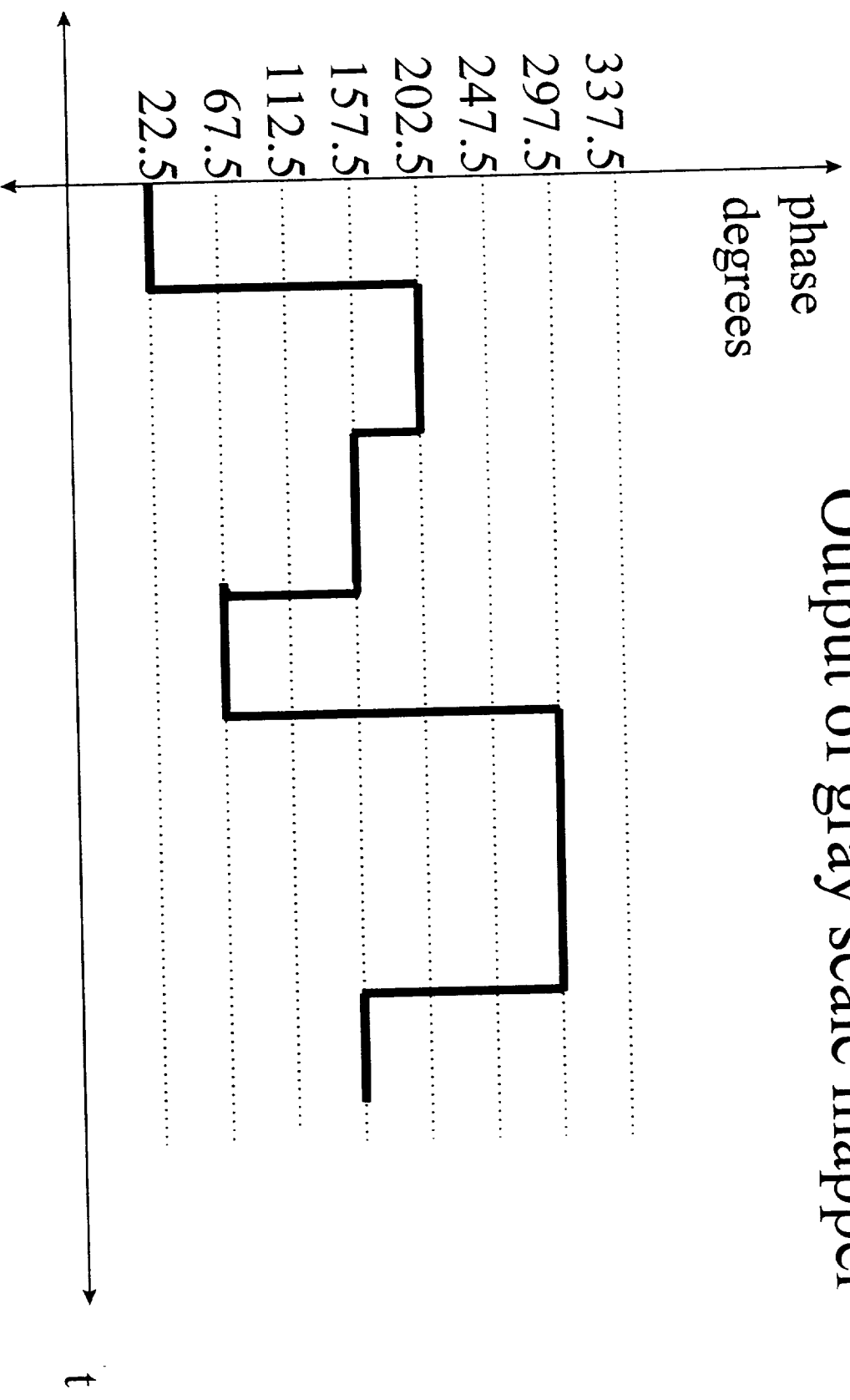


ideal or  
imbalance & asymmetry

Non-Constant Envelope pulse  
shaped 8-PSK  
(Ruben/ Josua)

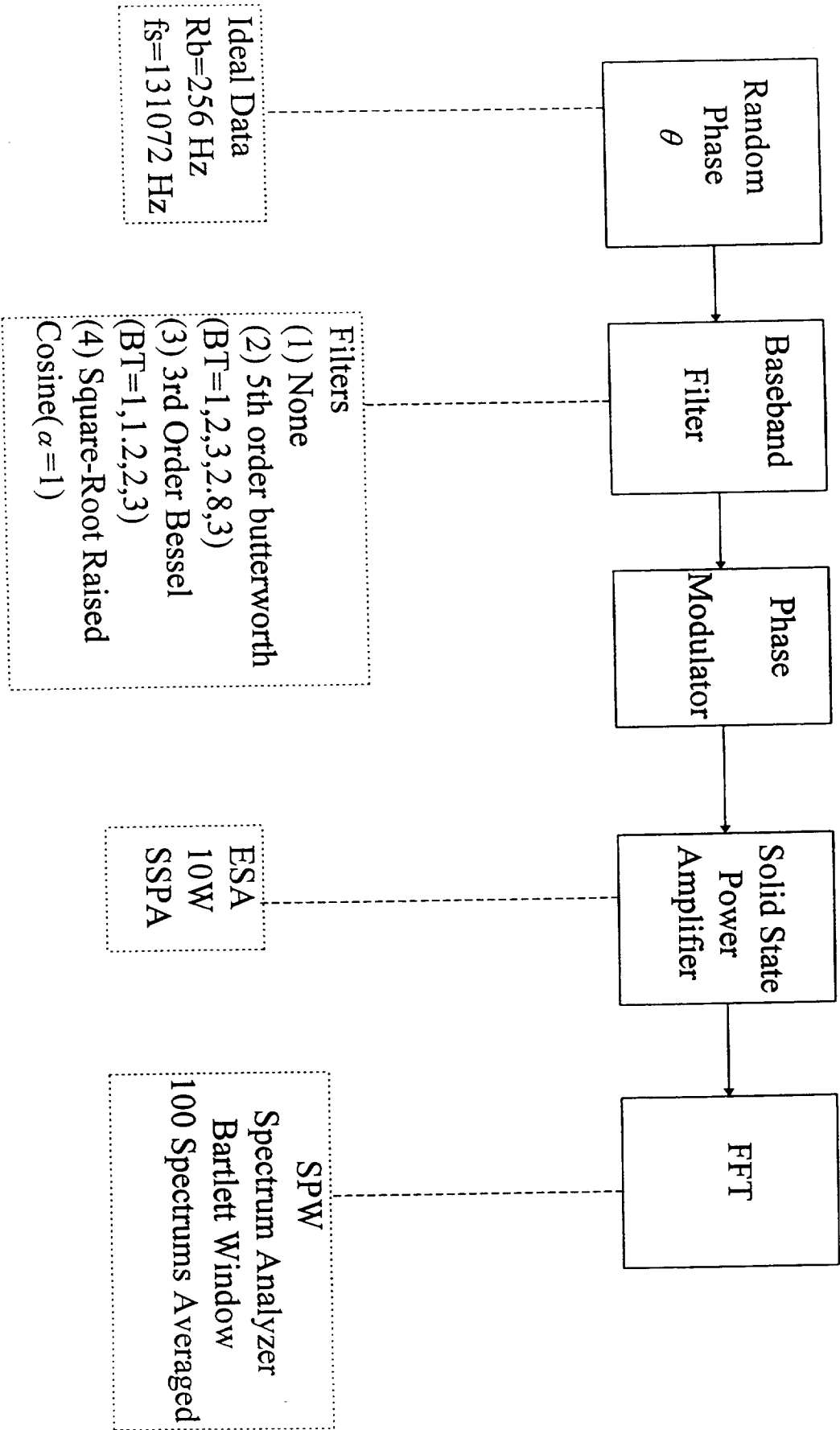


# Output of gray scale mapper

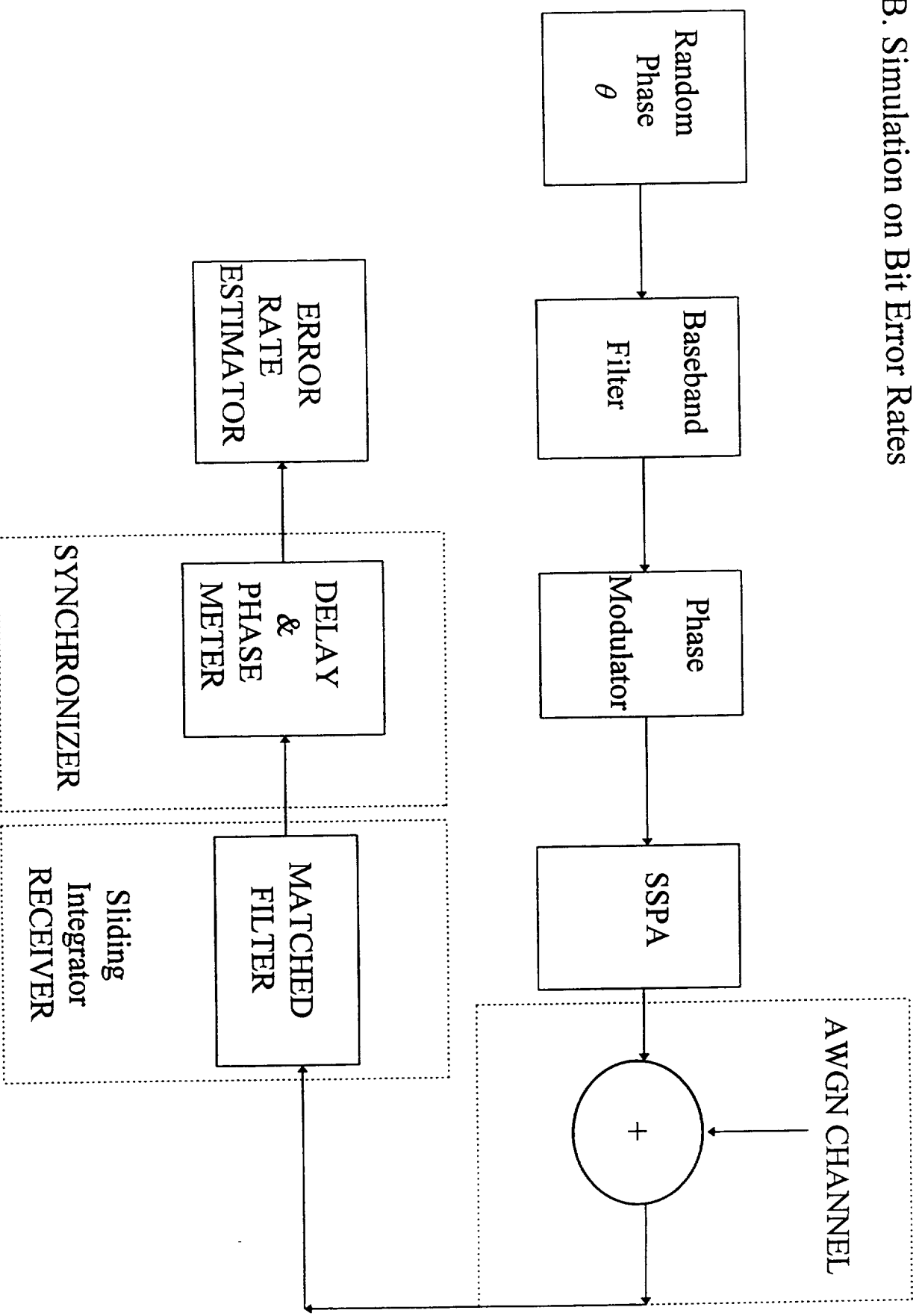


# . 8PSK EFFICIENT MODULATION STUDY OVERVIEW

## A. Spectrum Analysis

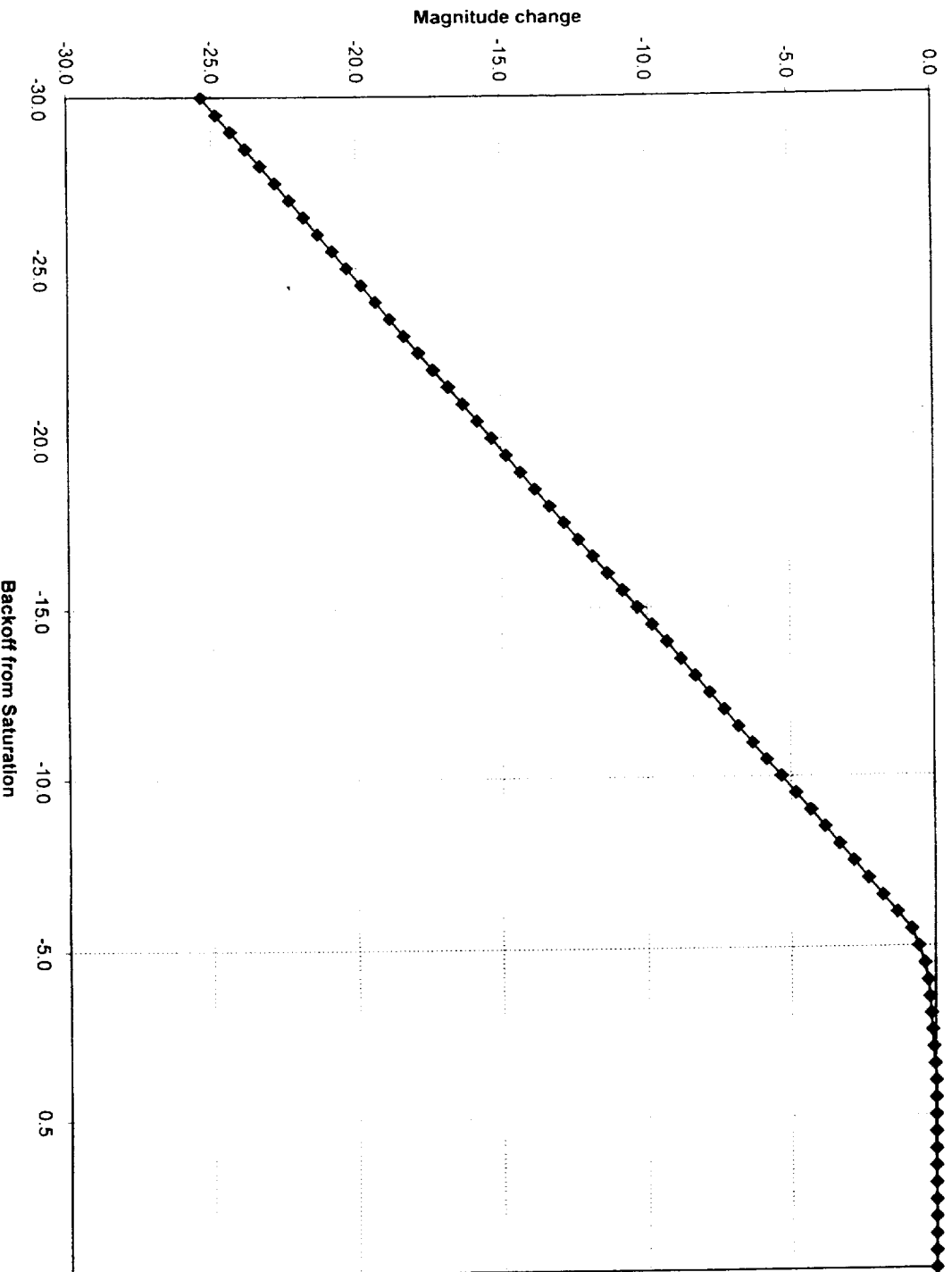


## B. Simulation on Bit Error Rates

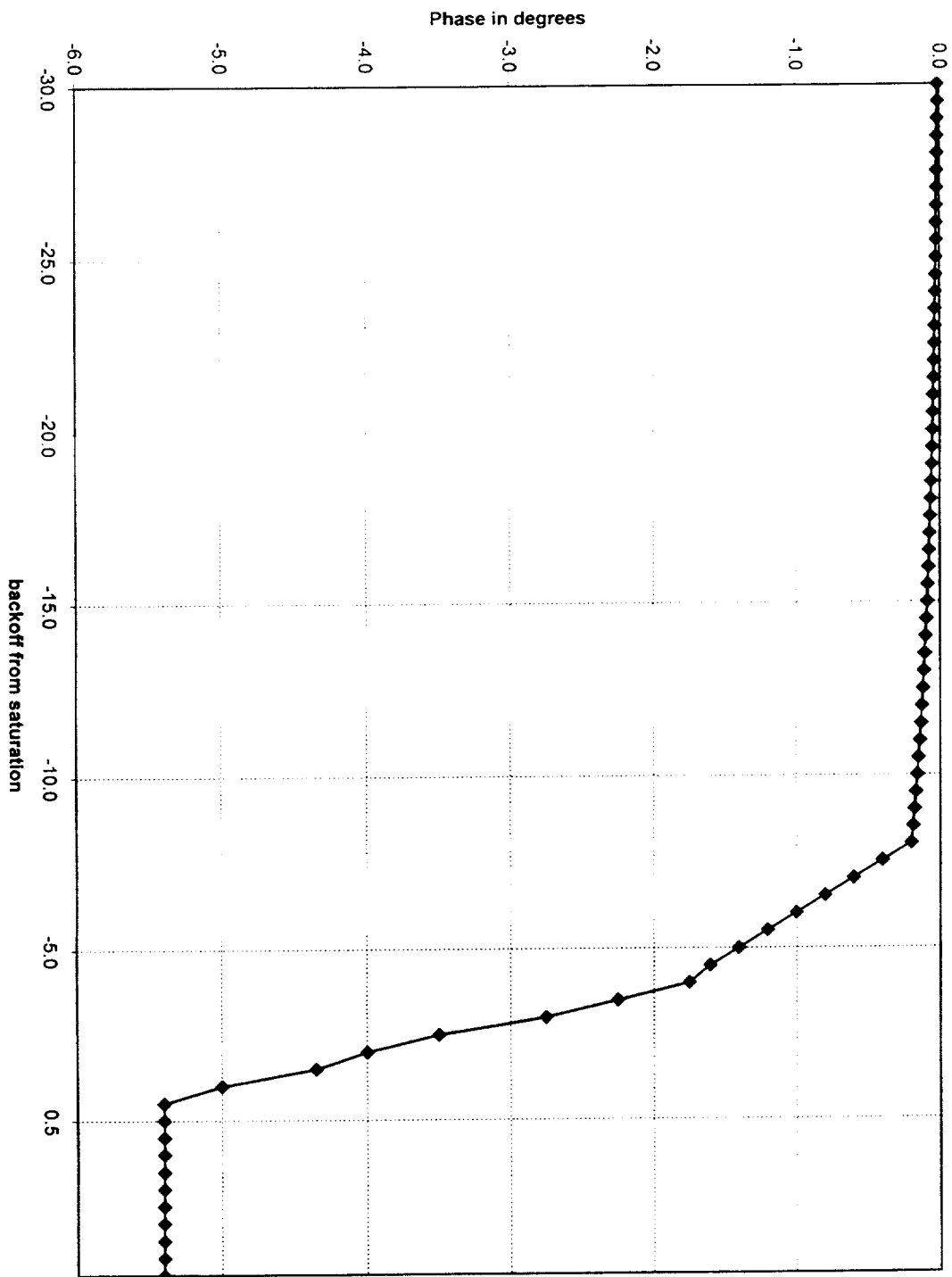


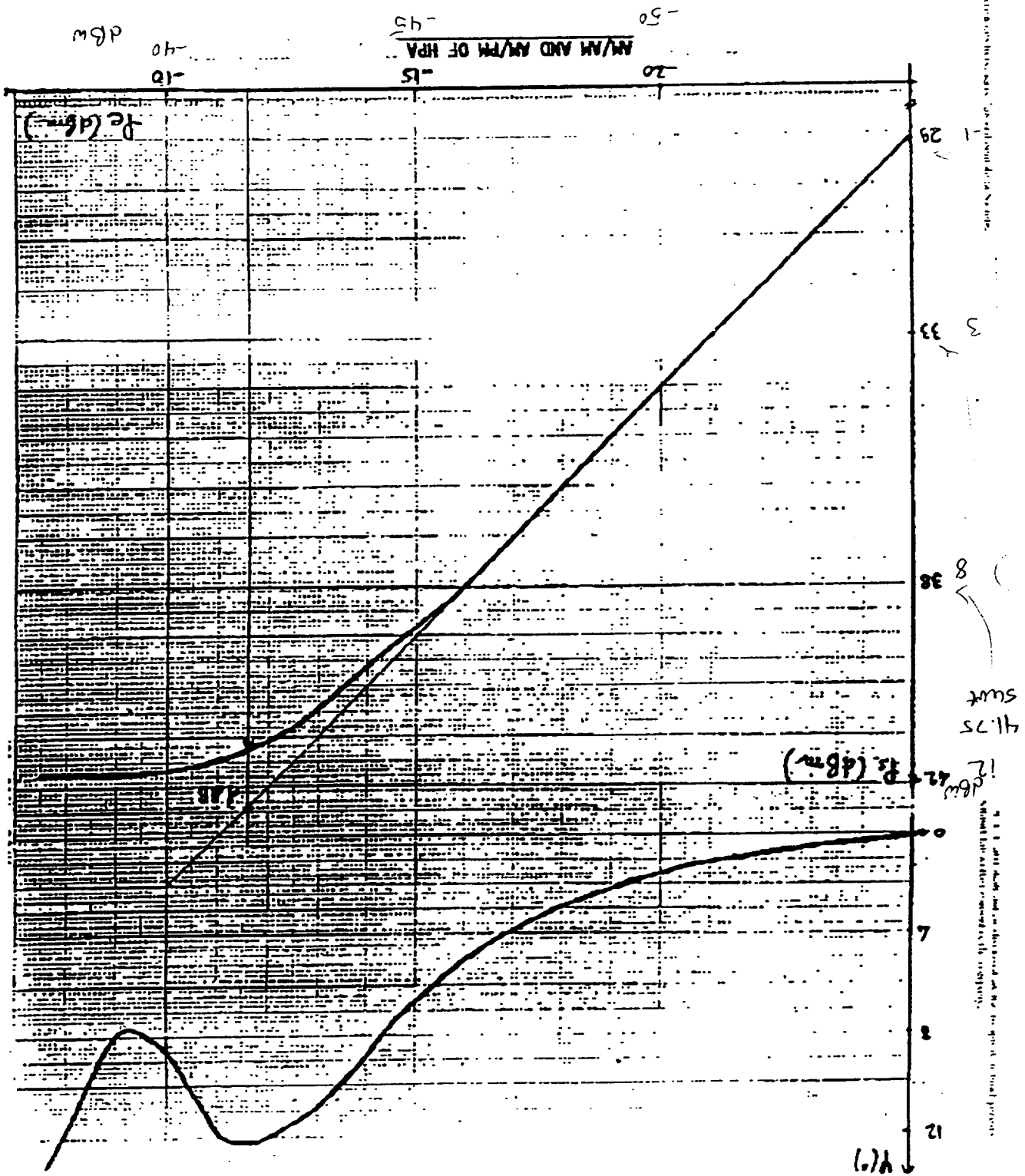


SSPA from JPL (Magnitude)



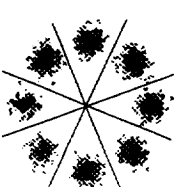
JPL SSPA Phase output





## **(4) - RESULTS FOR 8PSK**

### ***8PSK Over Non-Linear Channel***



8PSK OVER  
NON-LINEAR  
SATELLITE CHANNELS

#### Power Containment and Spurious Emissions

- PSD plots
- Band Utilization Ratio
  - . establishes user spacing based upon modulation characteristics
  - . sideband attenuation avoids interference to adjacent users

$$\rho = \frac{\text{number of s/c with filtering accommodated in freq. band}}{\text{number of s/c without filtering accommodated in freq. band}}$$

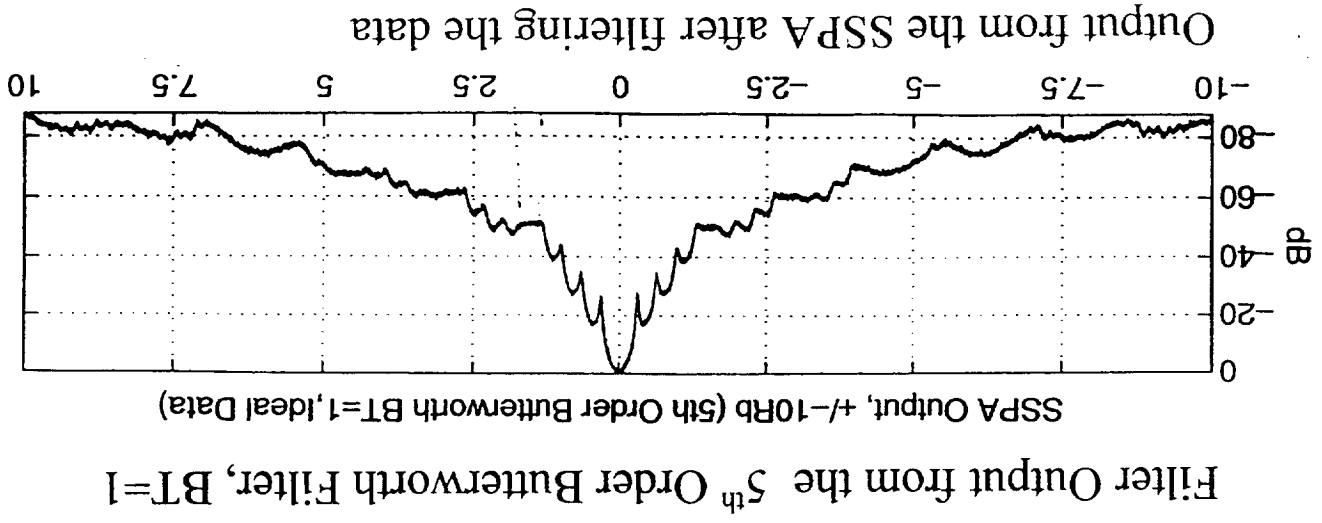
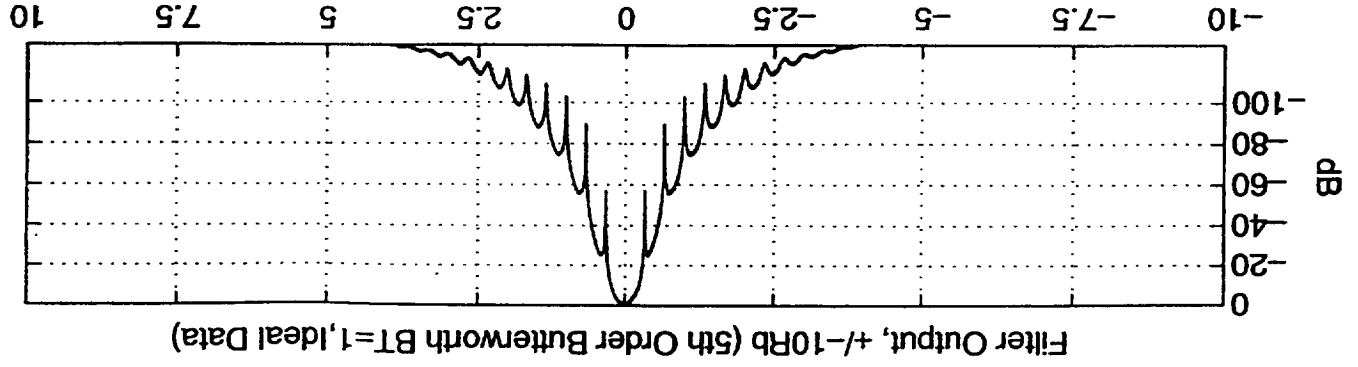
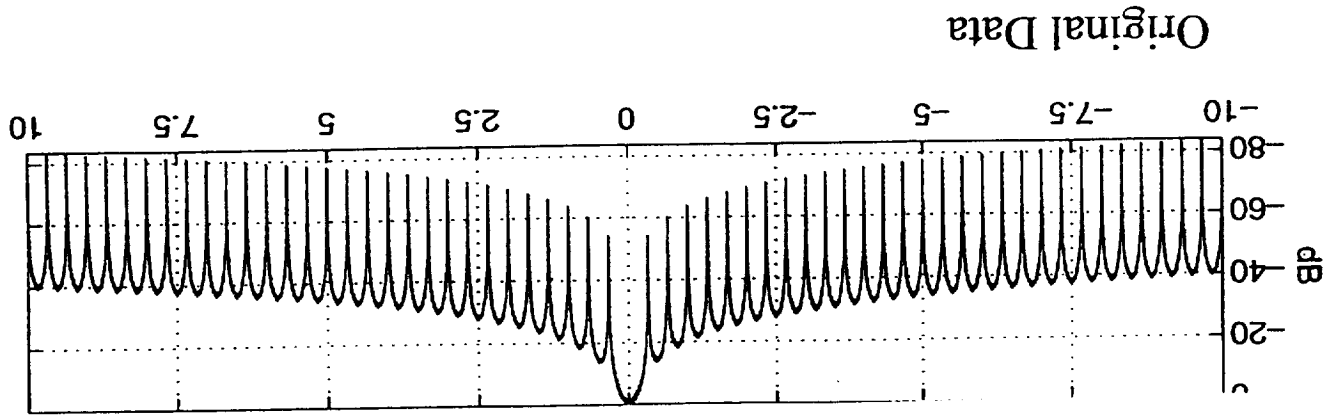
#### End-to-End Performance:

- BER plotted as function of symbol SNR ( $E_b/N_0$ )

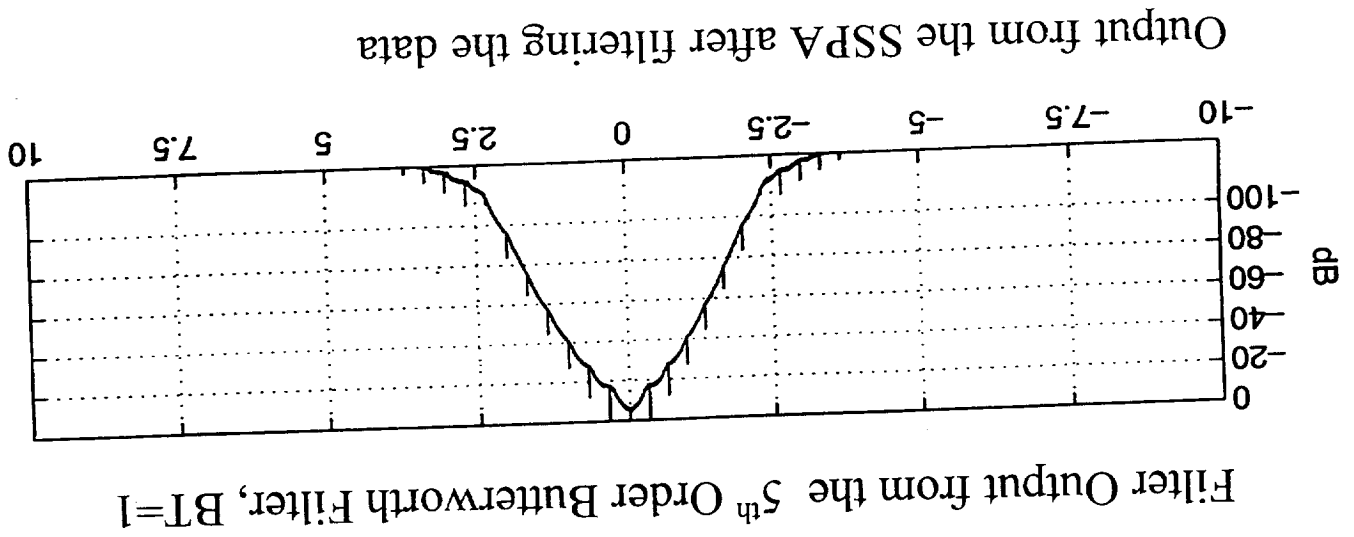
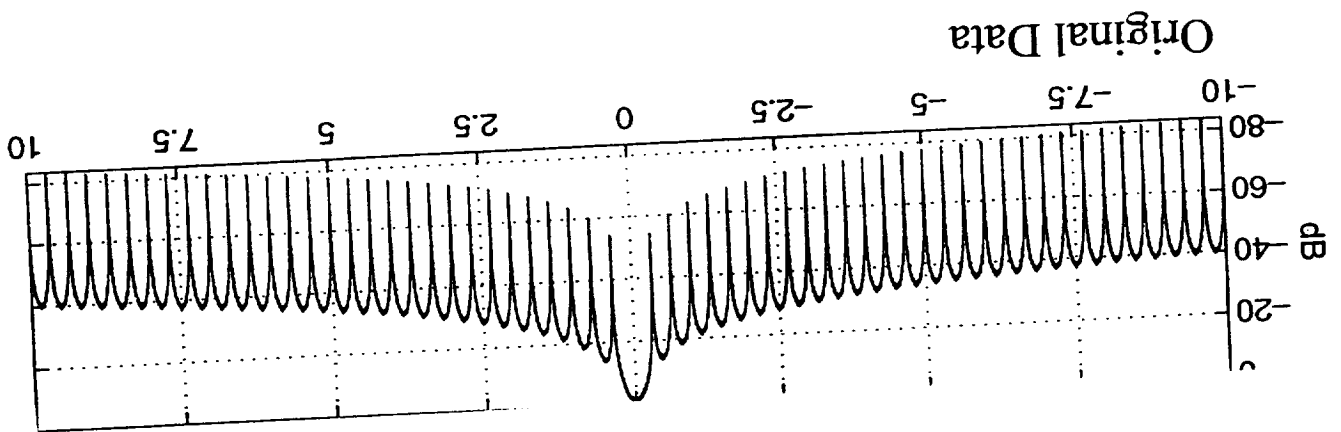
#### Non-Constant Envelope

- Average Symbol Variance vs. Bandwidth

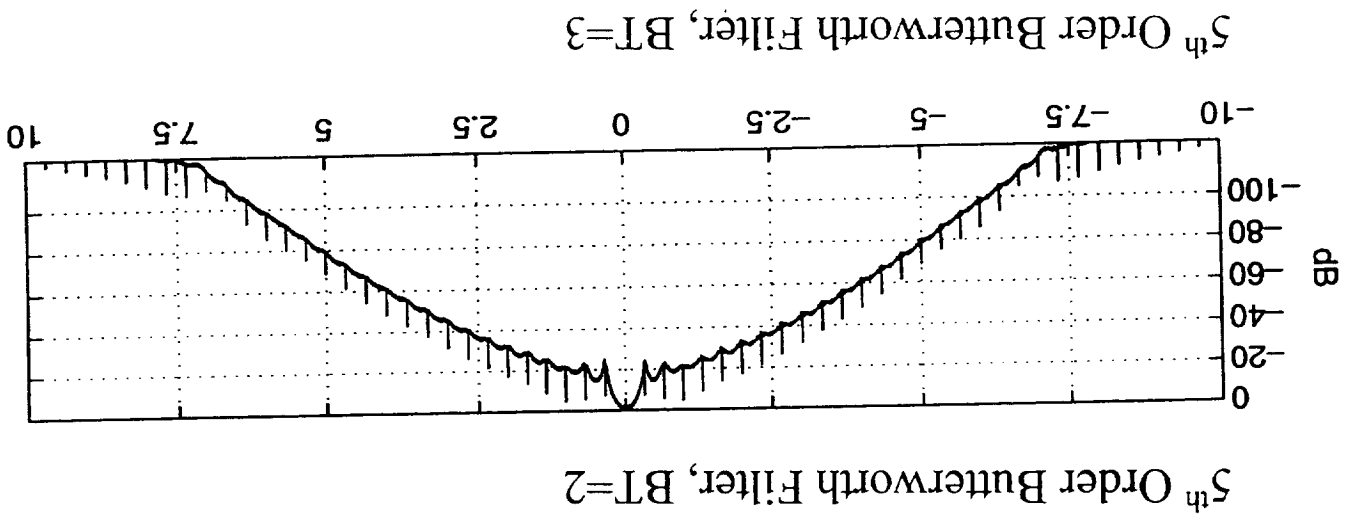
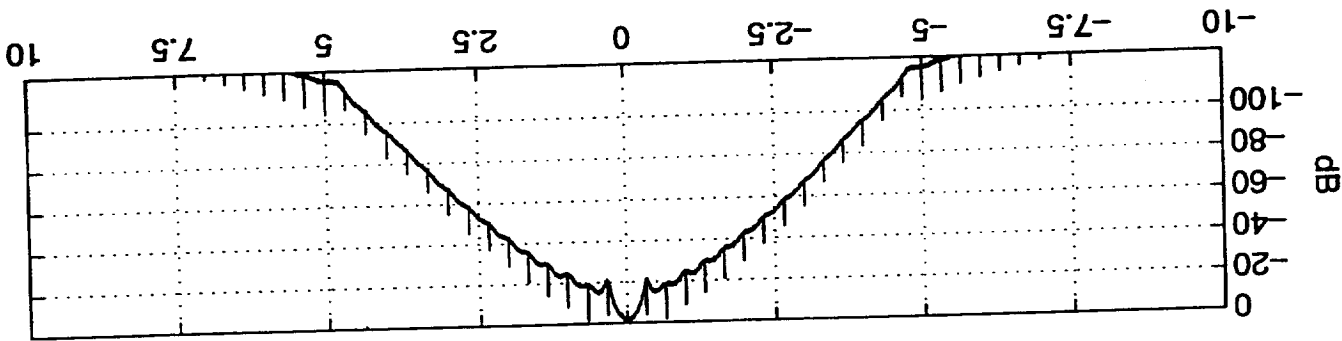
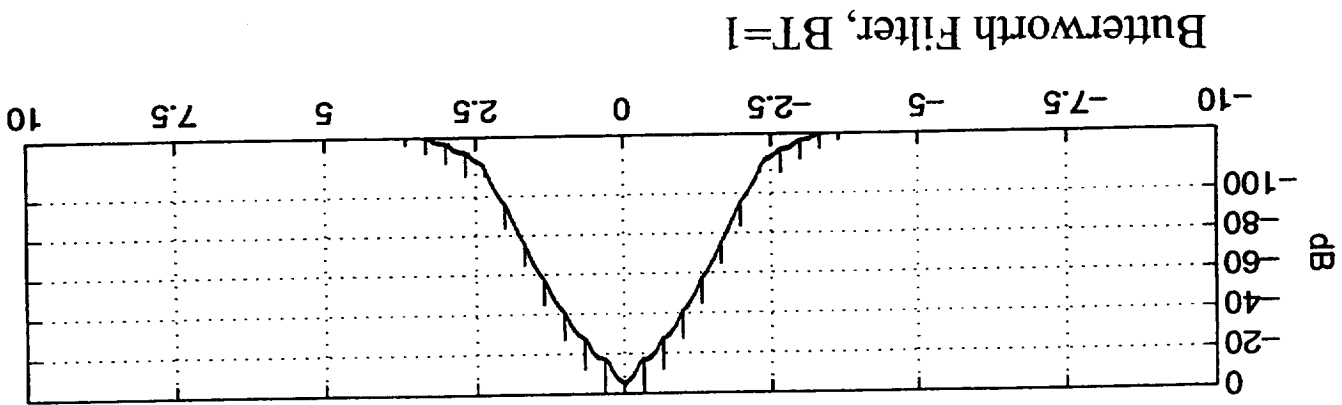
# Power Spectra Plots Non-constant envelope 8 PSK



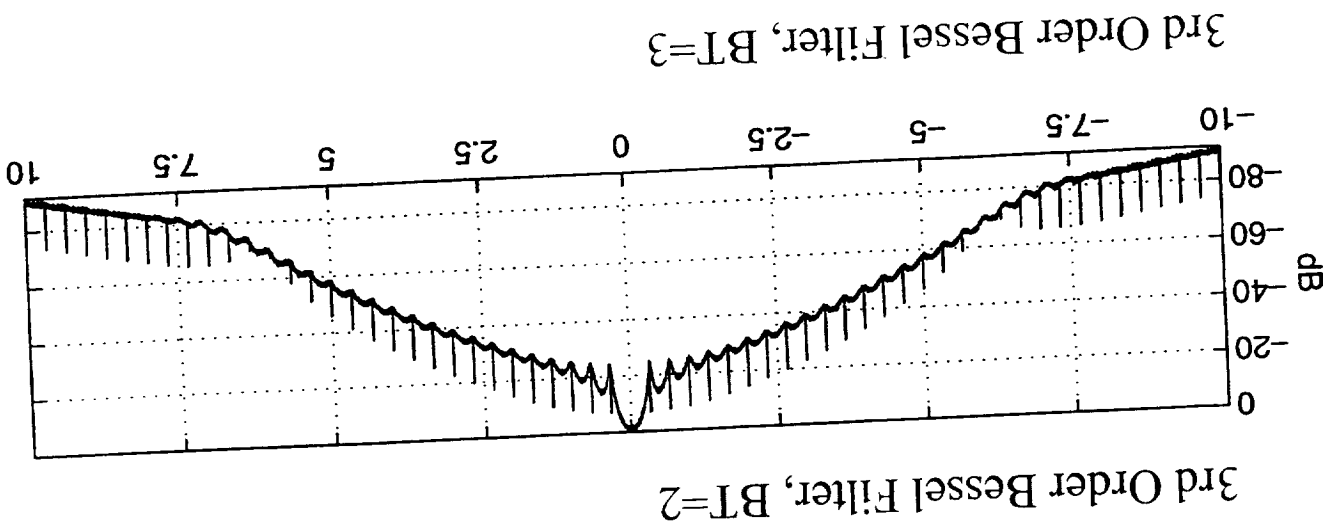
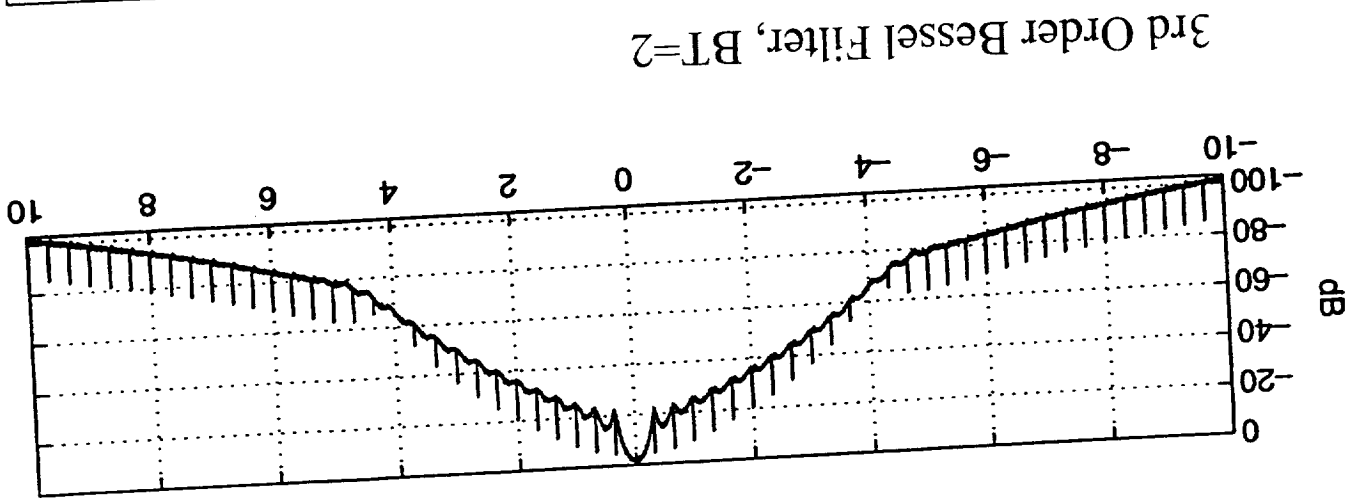
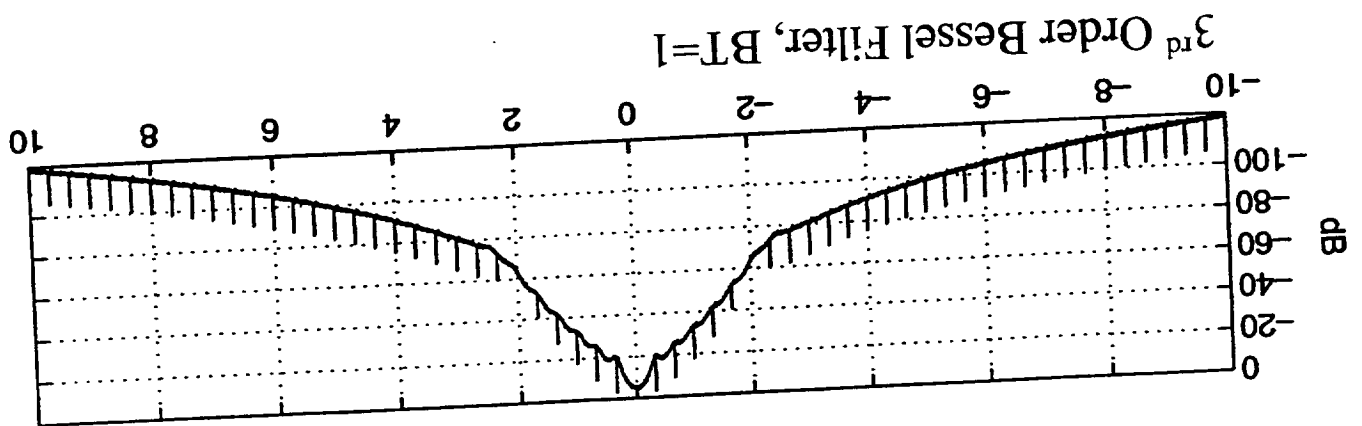
# Power Spectra Plots Constant envelope 8 PSK



# SSPA Power Spectra Plots Constant envelope 8 PSK

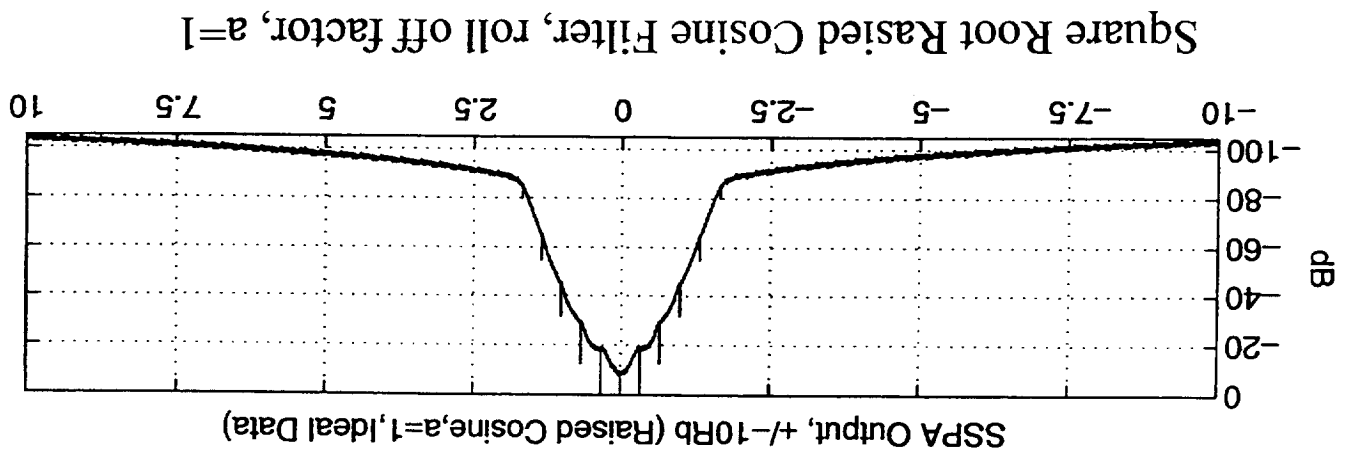


# SSPA Power Spectra Plots Constant envelope 8 PSK

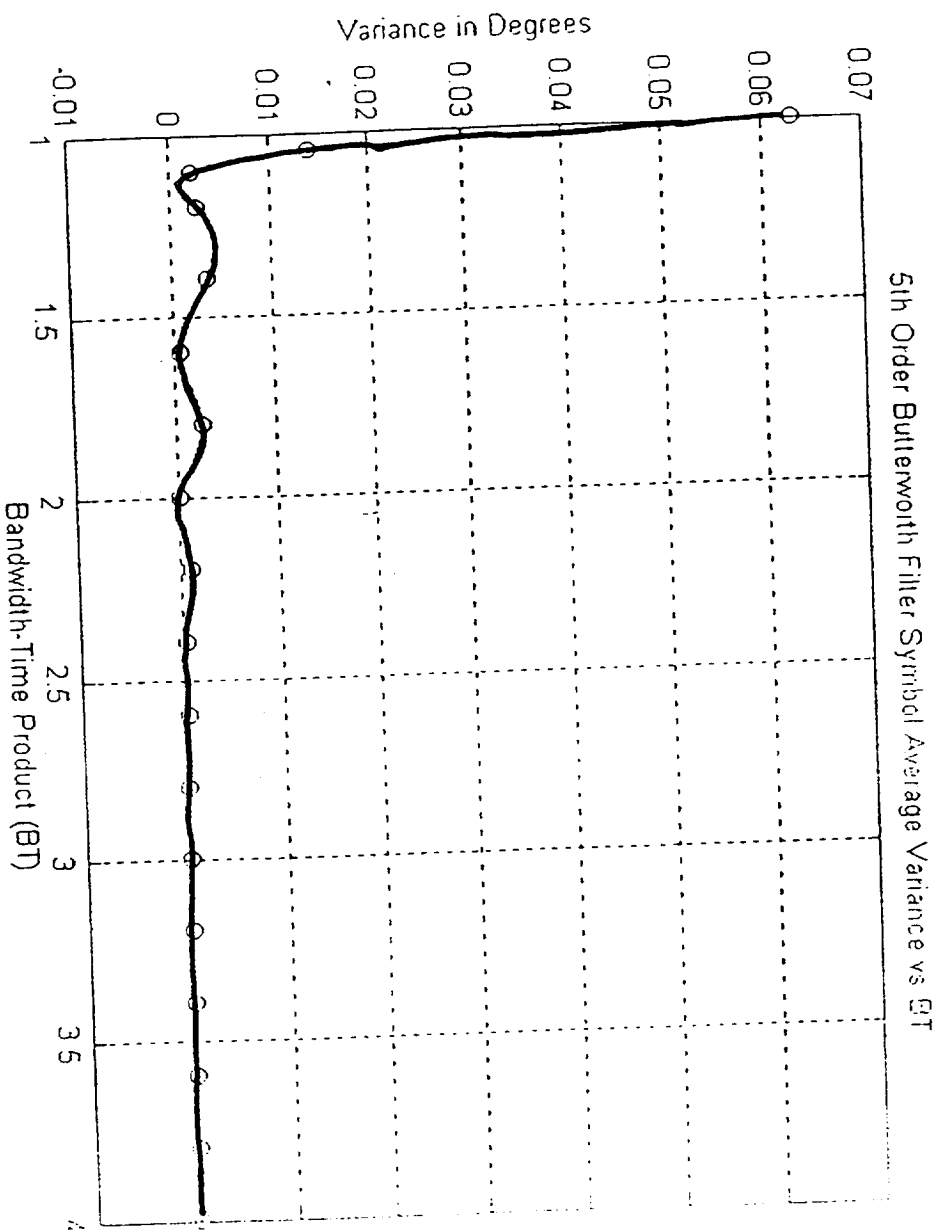




# SSPA Power Spectra Plots Constant envelope 8 PSK

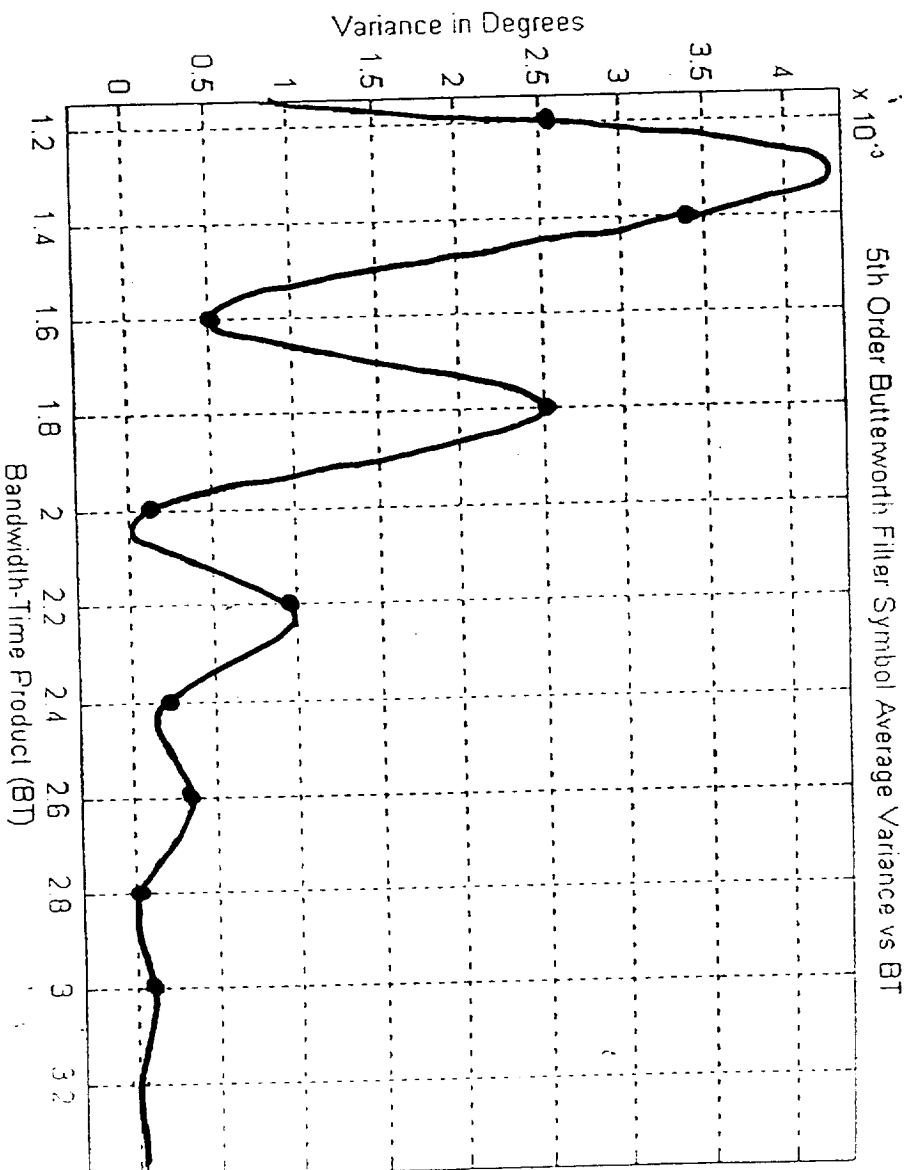


# ***RESULTS FOR 8PSK Over Non-Linear Channel***

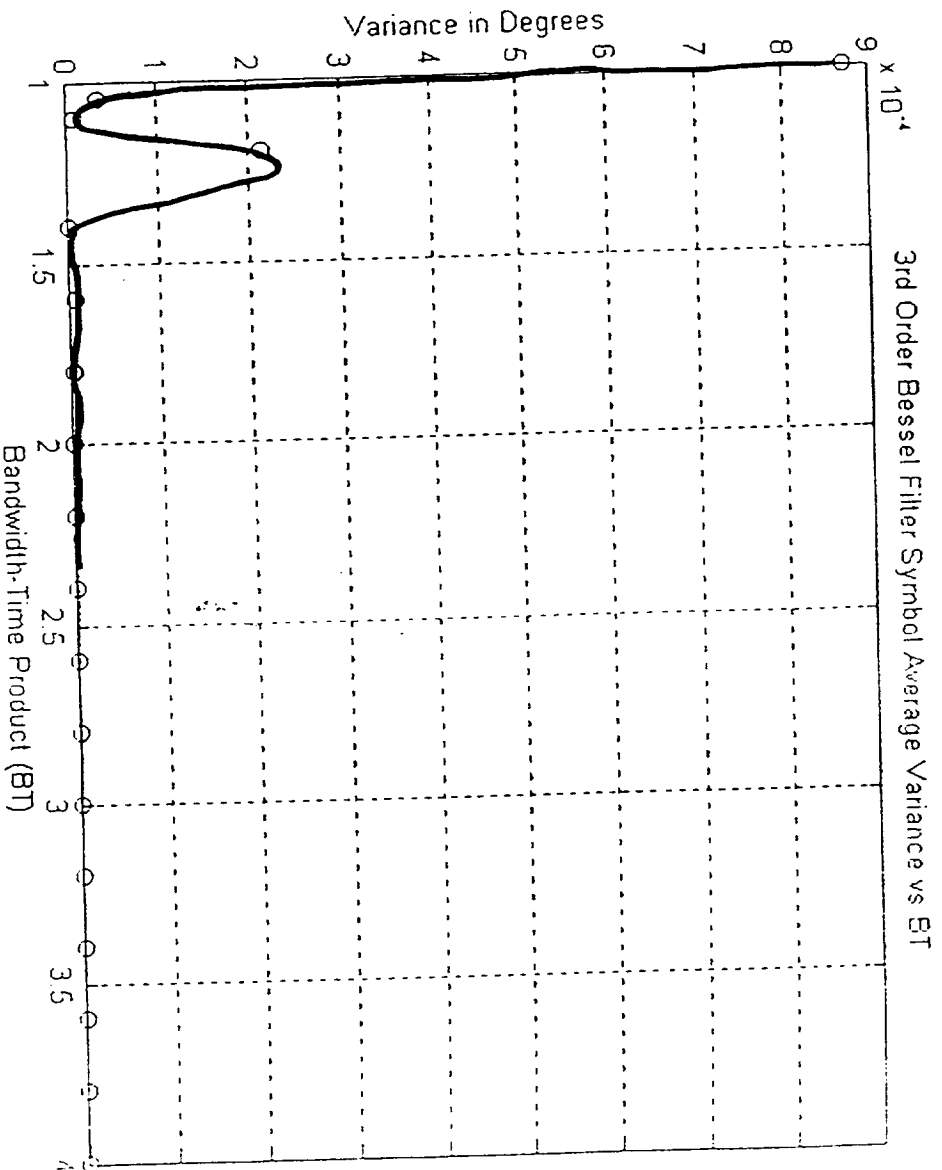


# ***RESULTS FOR 8PSK***

## ***8PSK Over Non-Linear Channel***



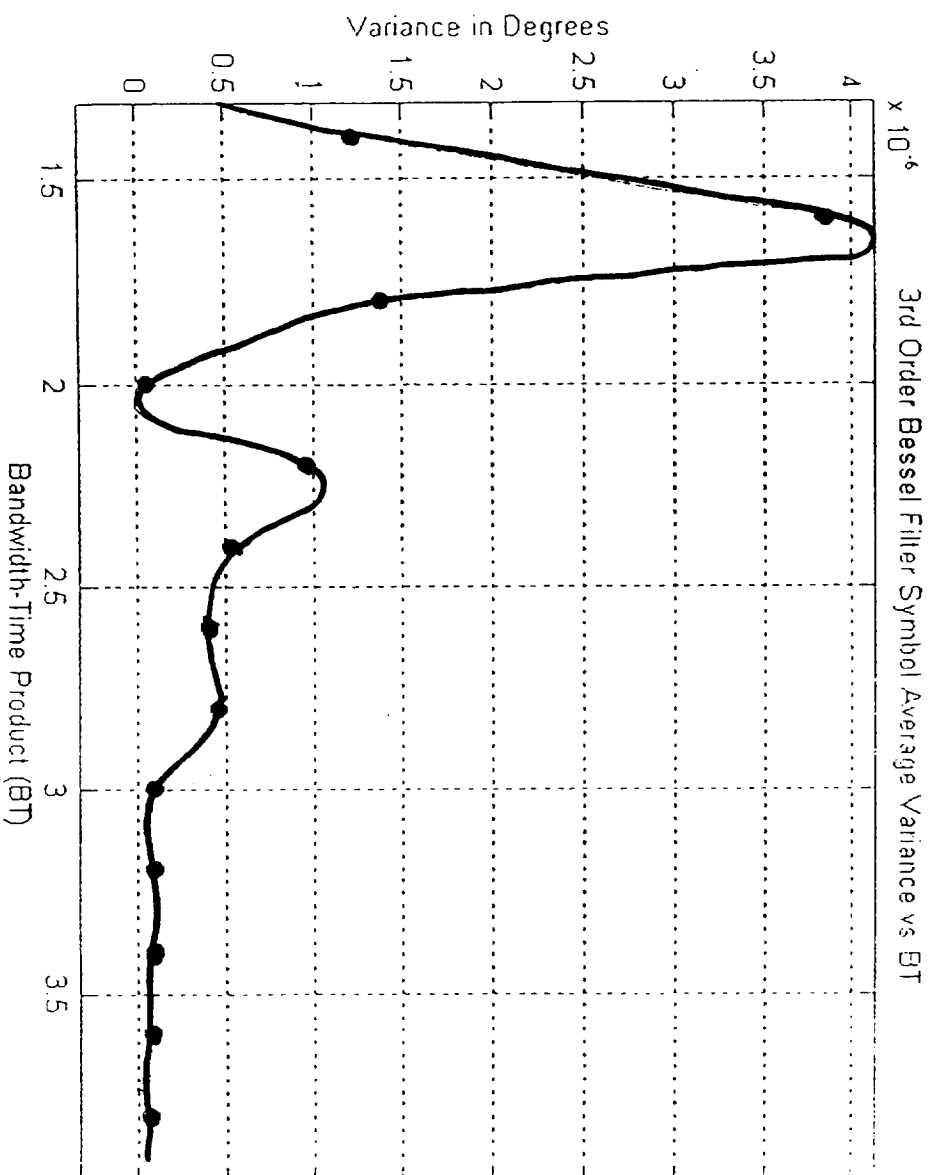
# ***RESULTS FOR 8PSK 8PSK Over Non-Linear Channel***



3rd Order Bessel Filter: Average Symbol Variance vs BT

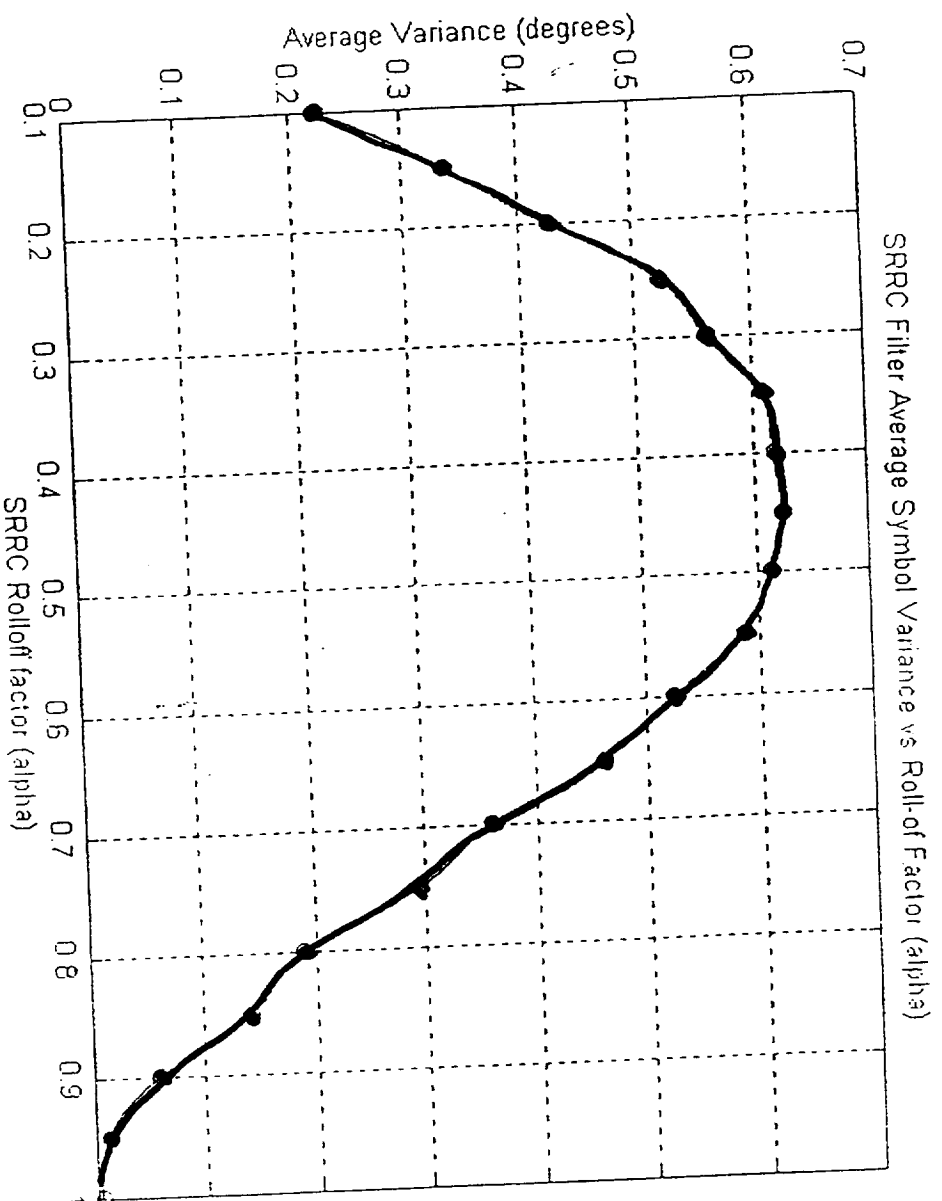
# *RESULTS FOR 8PSK*

## *8PSK Over Non-Linear Channel*

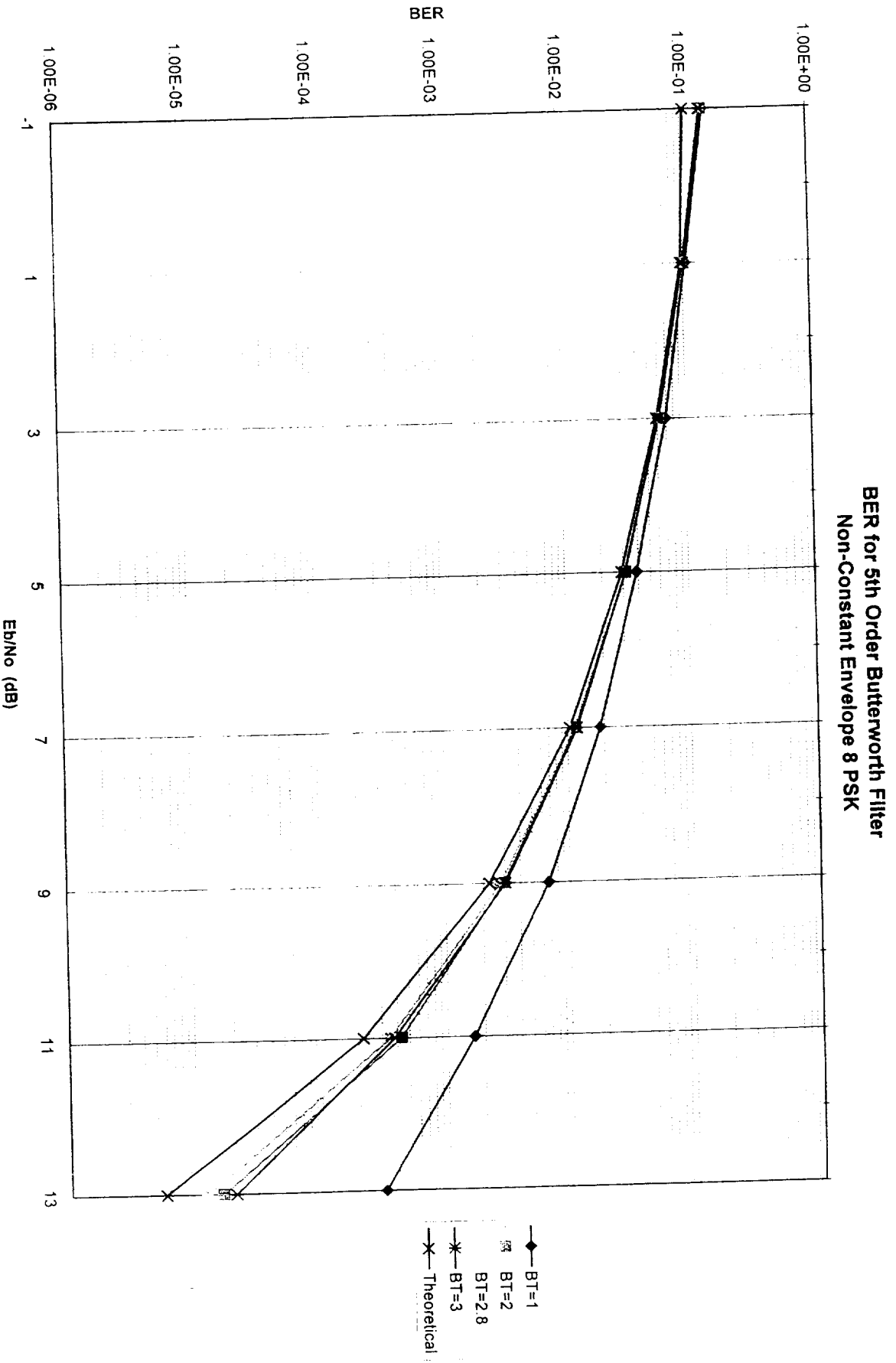


Zoom of Average Symbol Variance vs BT for 3rd Order Bessel Filter

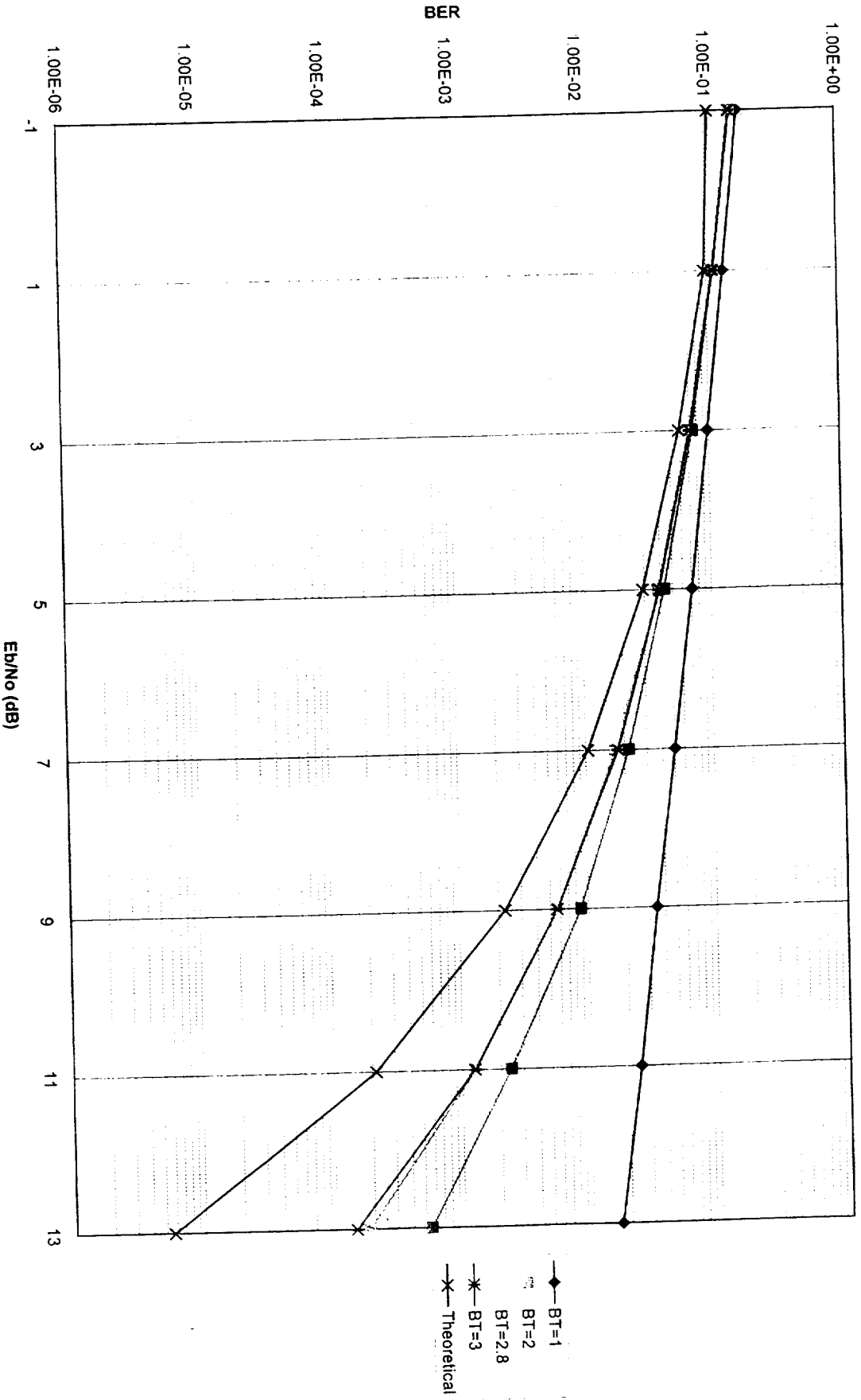
# *RESULTS FOR 8PSK 8PSK Over Non-Linear Channel*



SRRC: Average Symbol Variance vs Roll-off factor ( $\alpha$ )

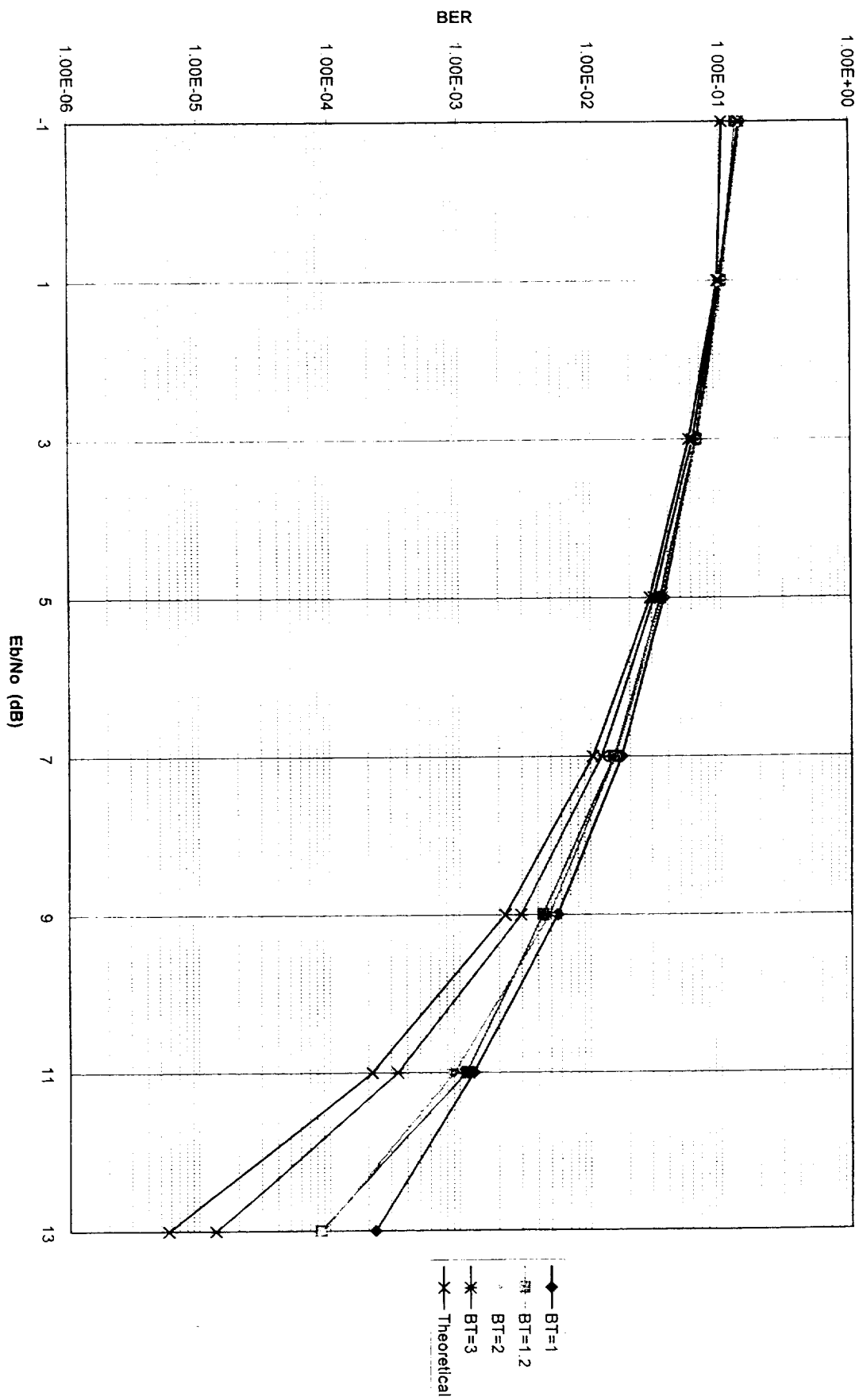


BER for 5th Order Butterworth Filter  
Constant Envelope 8 PSK

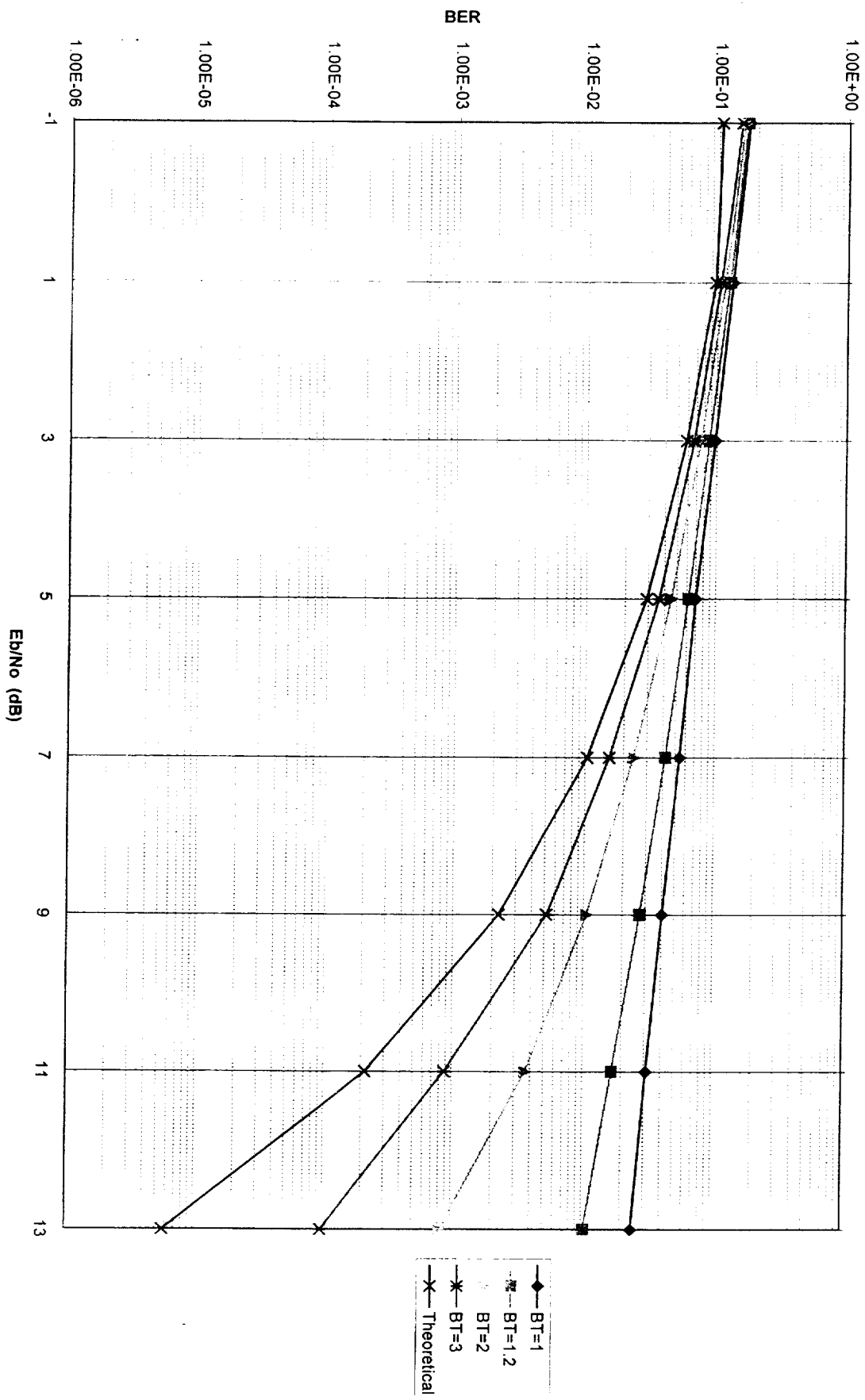




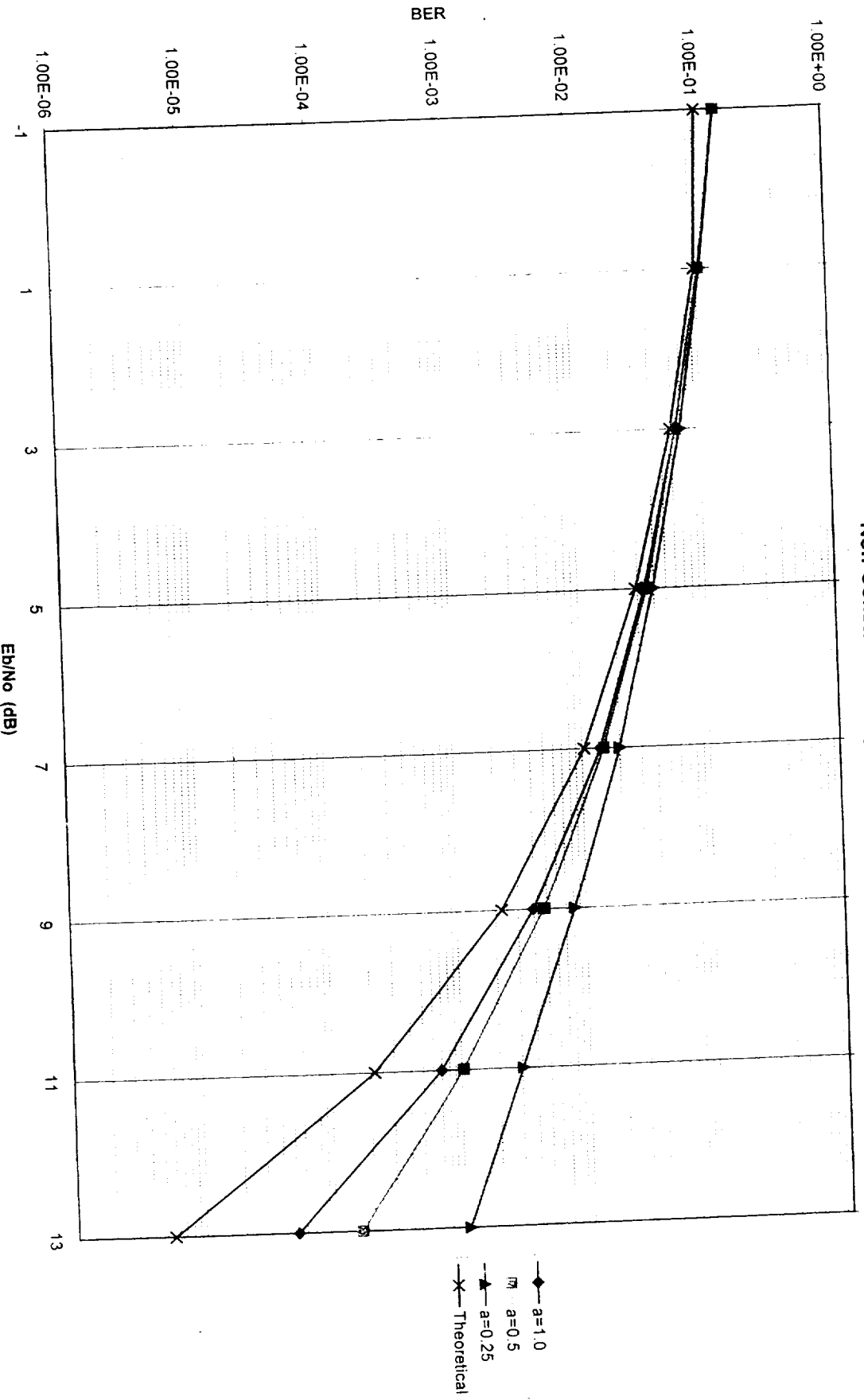
BER for 3rd Order Bessel Filter  
Non-Constant Envelope 8 PSK



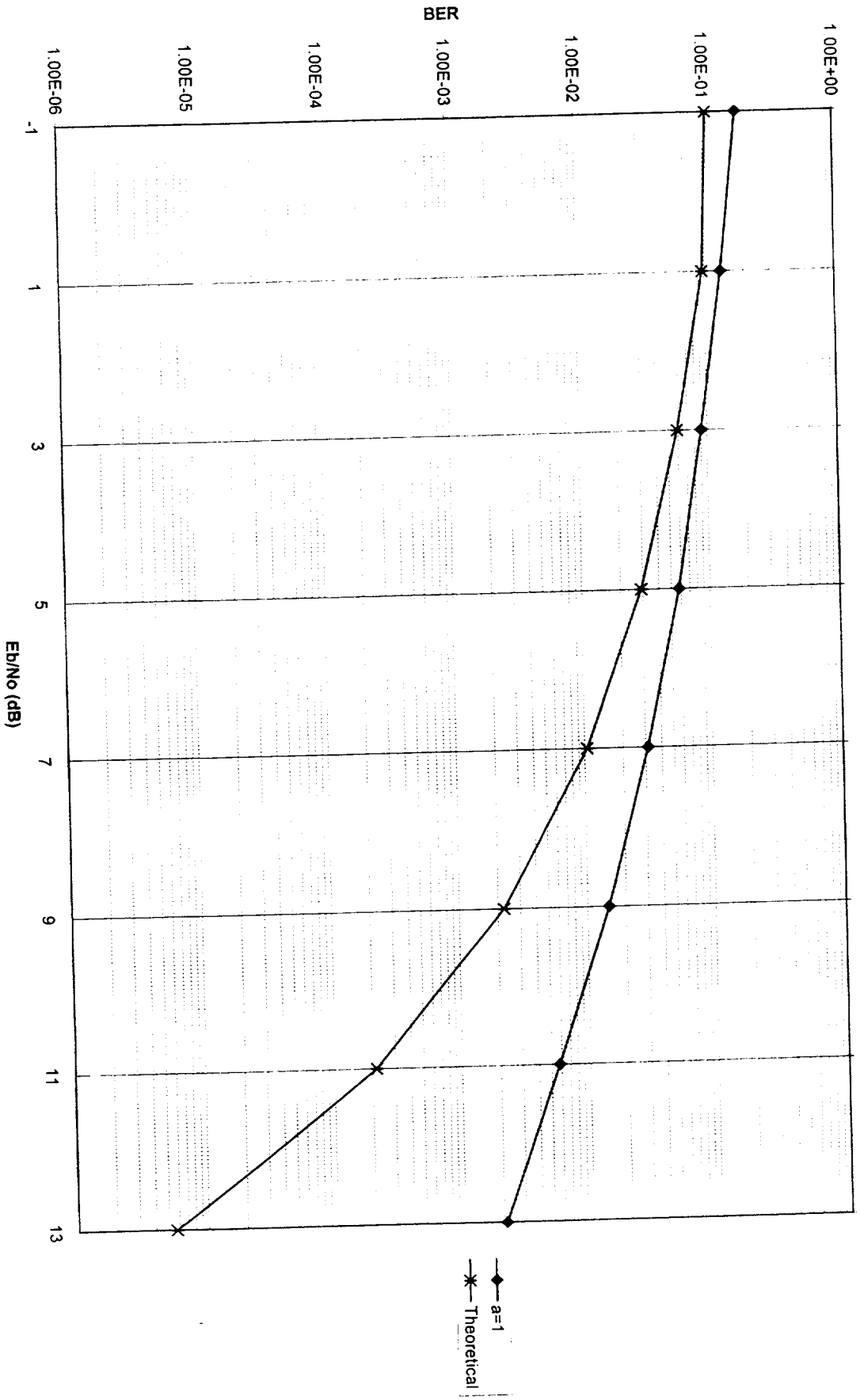
BER for 3rd Order Bessel  
Constant Envelope 8 PSK



BER for SRRC Filter  
Non-Constant Envelope 8 PSK



BER for SRRC Filter  
Constant Envelope 8 PSK

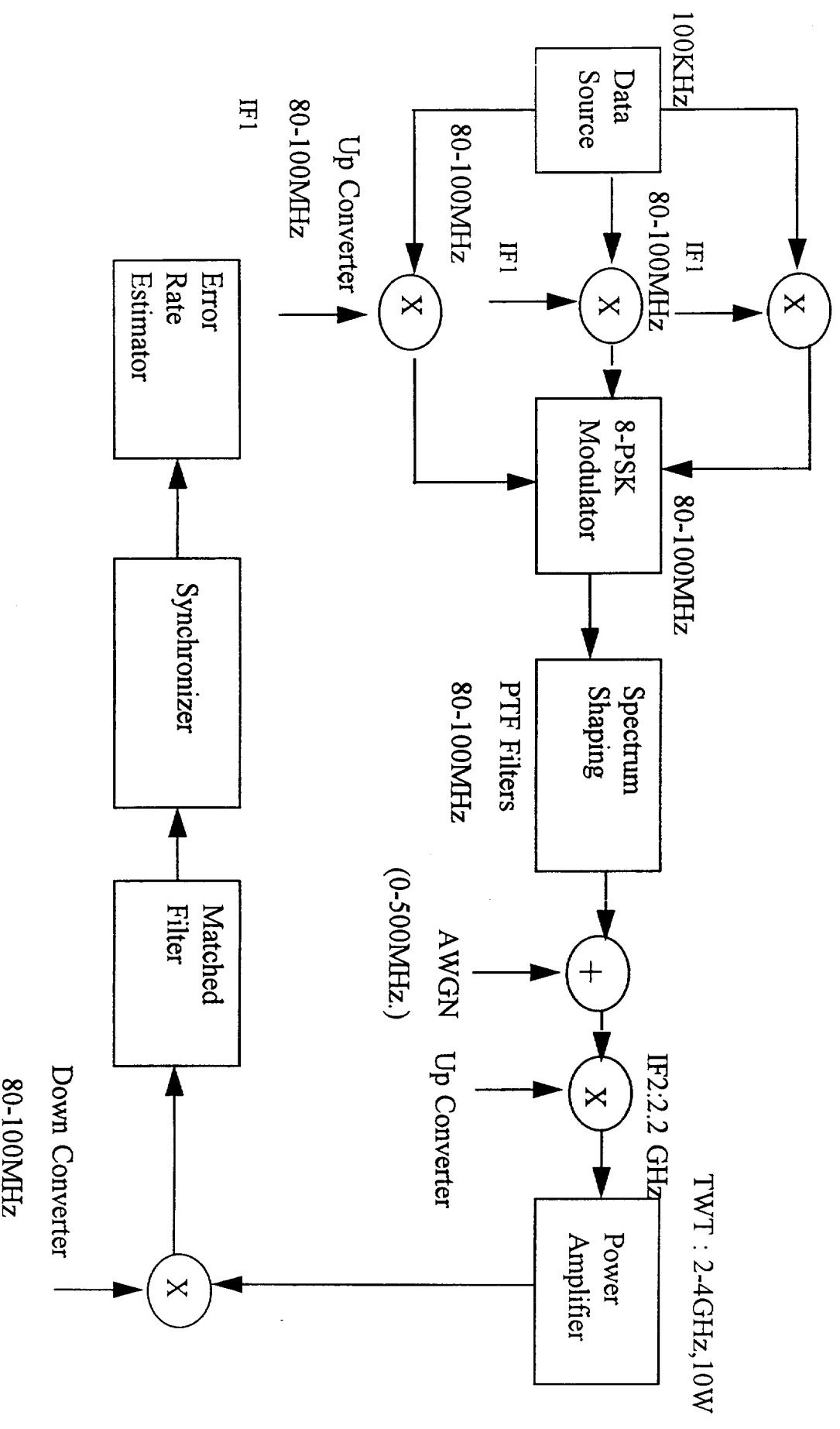


## ISI Losses and Utilization Ratio

Filter	Non-Const Envelope		Constant Envelope	
Type	Utilization	ISI Loss	Utilization	ISI Loss
None	1.00		1.00	
BW, BT=1	14.00	2.00	27.00	
BW, BT=2	13.46	0.43	14.4	
BW BT=2.8	10.00	0.29	10.8	
BW BT=3	9.72	0.43	10.1	
Bessel, bt=1	11.86	1.58	20	
Bessel, bt=1.2	11.67	1.44	17.1	
Bessel, bt=2	10.94	1.15	10.8	
Bessel, bt=3	7.78	0.29	7.7	
SRRC a=1	14.00	0.86	28.5	

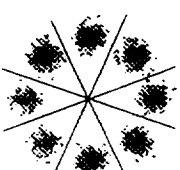
# The Implementation of Pulse Shaping on 8-PSK Signaling Over a Non-Linear Satellite Channels

## •Block Diagram



## ***(5) - CONCLUSIONS/FURTHER WORK***

---



8PSK OVER  
NON-LINEAR  
SATELLITE CHANNELS

### Conclusions:

- Improvement by a factor of 12 to 24 ( $BT=1$ ) with filtering 8PSK; Constant Envelope 8PSK gives 24 to 48 improvement at  $BT=1$ .
- BER still not good enough
- For non-constant envelope: there is a pattern!

### What is next:

- Obtain better models for High Power Amplifiers
- Simulations for higher order of BT;
- Implement in hardware;
- Find a close form equation for different filter orders.
- Use Equalization to improve BER
- Investigate spikes

# **CODING TALK OUTLINE**

1. INTRODUCTION TO TURBO CODES
2. ON TRANSPARENT TURBO CODES
3. TURBO CODES IN PULSED CW RFI
4. HIGH-RATE TURBO CODES
5. WSGT TURBO CODE TEST





# *INTRODUCTION TO TURBO CODES*

William E. Ryan, Asst. Prof., NMSU

March 11, 1997



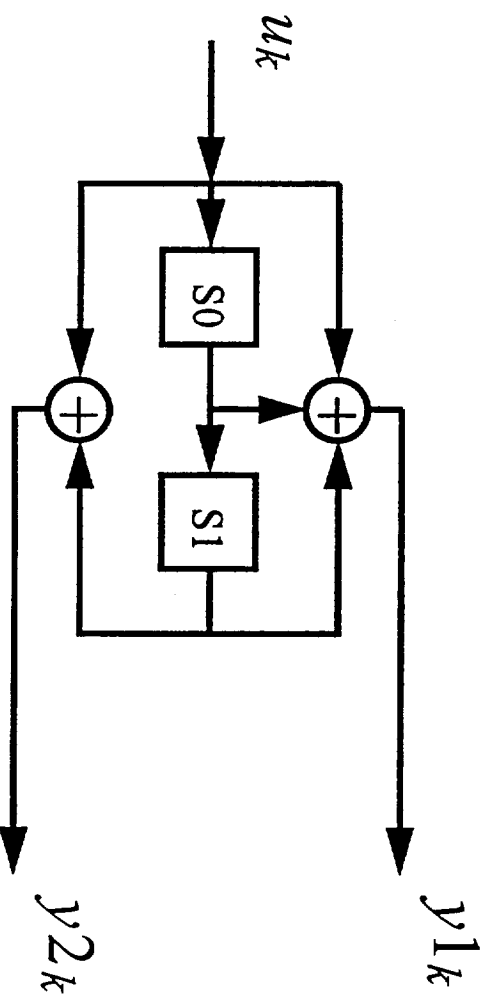
# NONRECURSIVE CONVOLUTIONAL CODES(CC)

**Example:** rate,  $r = 1/2$ .

$$G(D) = [g_1(D) \quad g_2(D)]$$

$$g_1 = 1 \ 1 \ 1 = 1 + D + D^2$$

$$g_2 = 1 \ 0 \ 1 = 1 + D^2$$

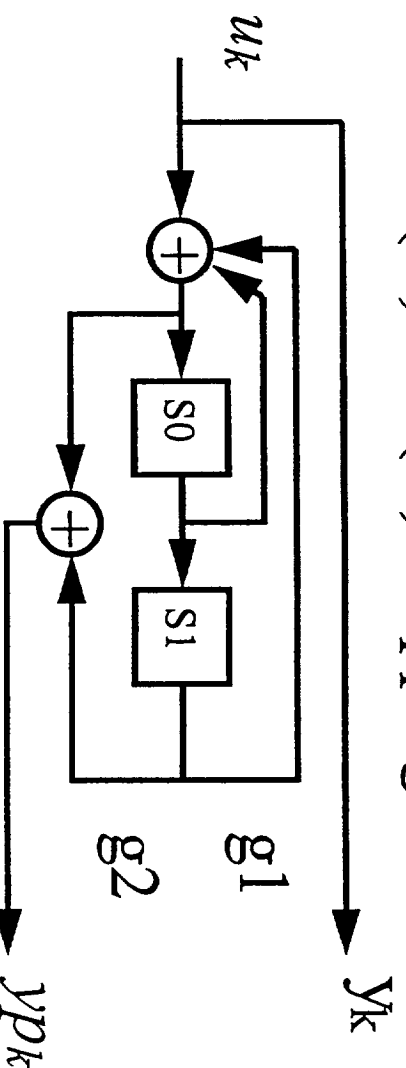


# RECURSIVE SYSTEMATIC CC (RSCC)

Remember  $G(D) = [g_1(D) \ g_2(D)]$  from non-recursive CC

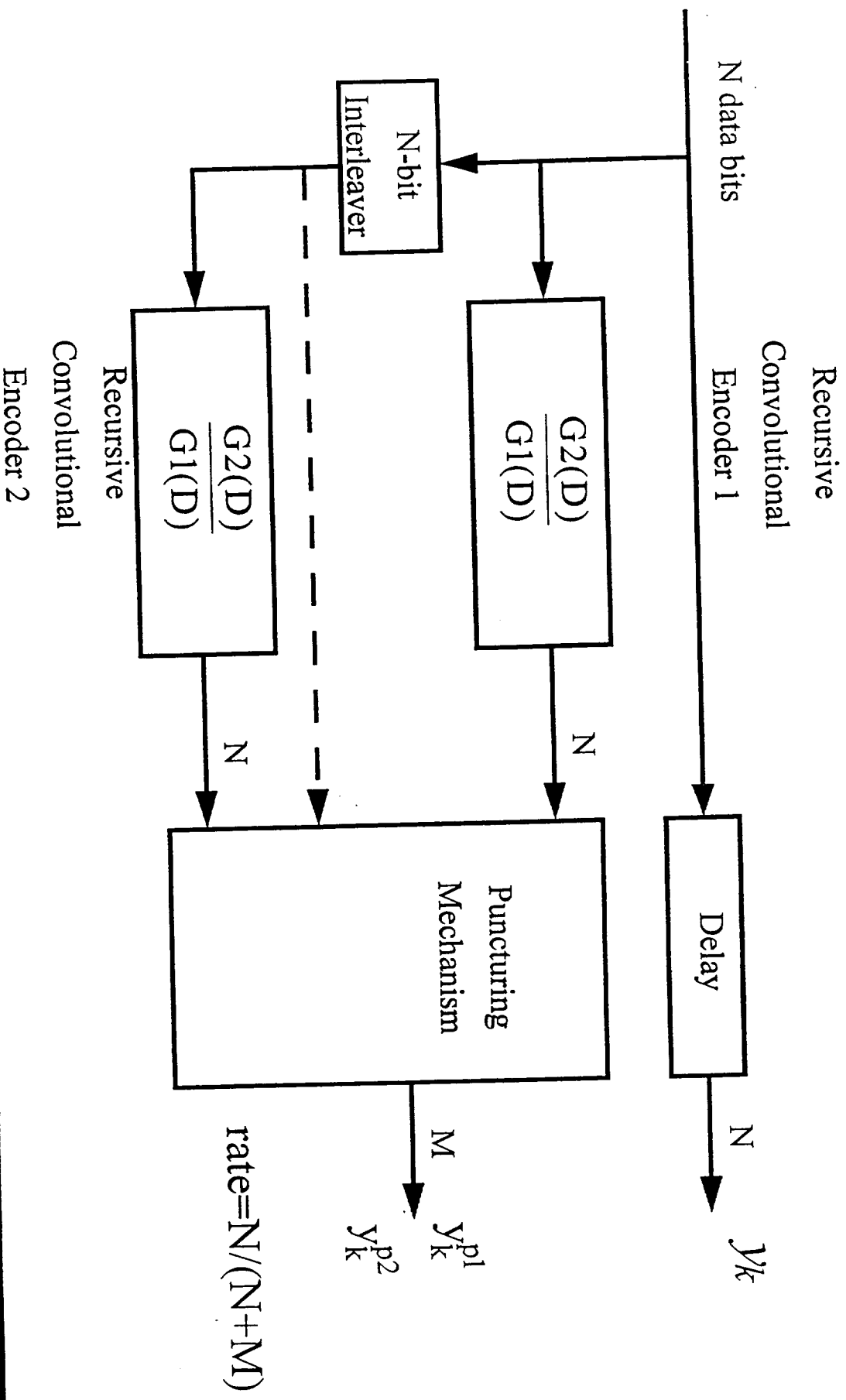
**Example:**  $G(D) = \begin{bmatrix} 1 & \frac{g_2(D)}{g_1(D)} \end{bmatrix}$  for recursive CC

- \* same set of code sequences  $c(D) = u(D) G(D)$
- \* different  $u(D) \leftrightarrow c(D)$  mapping



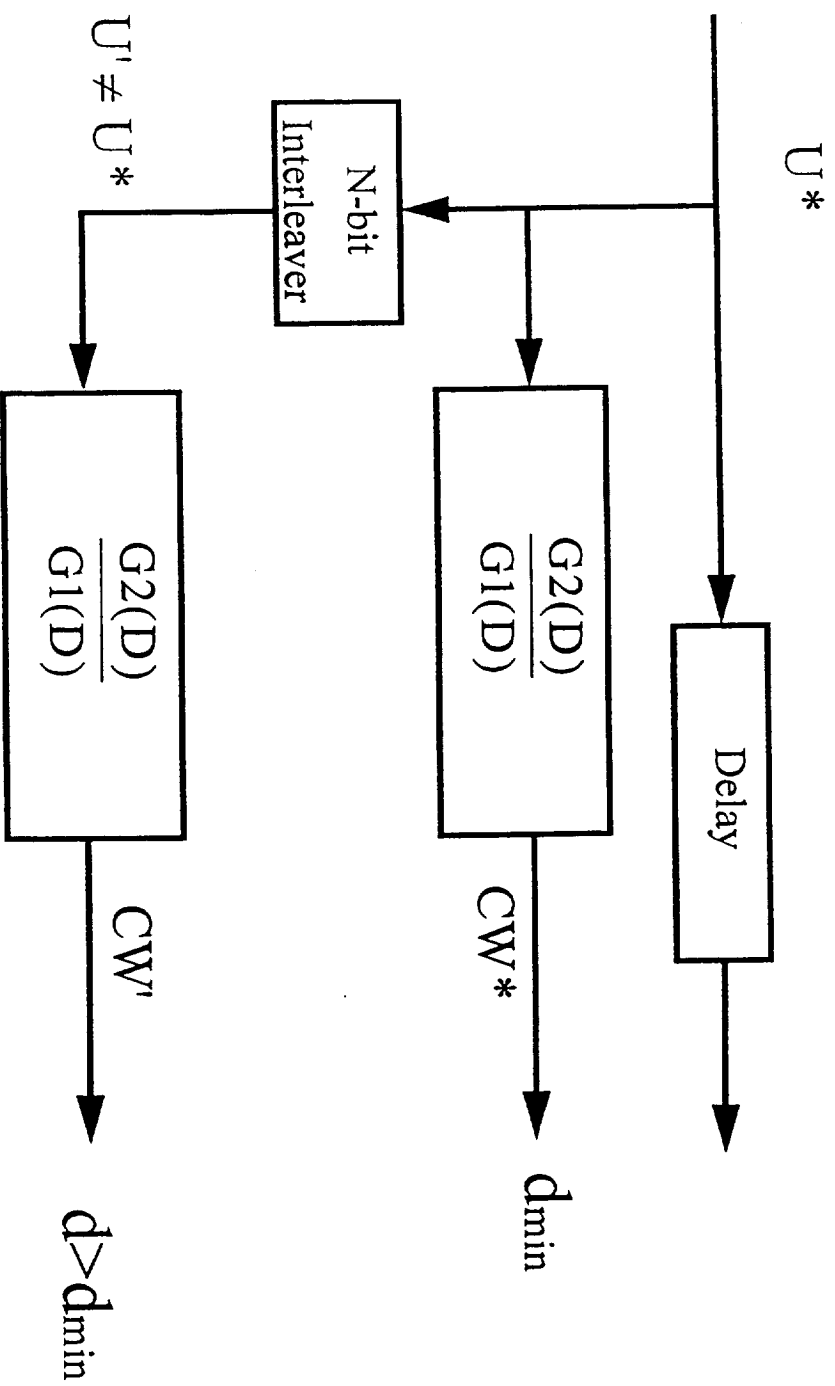


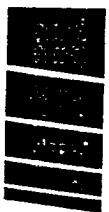
# TC ENCODER



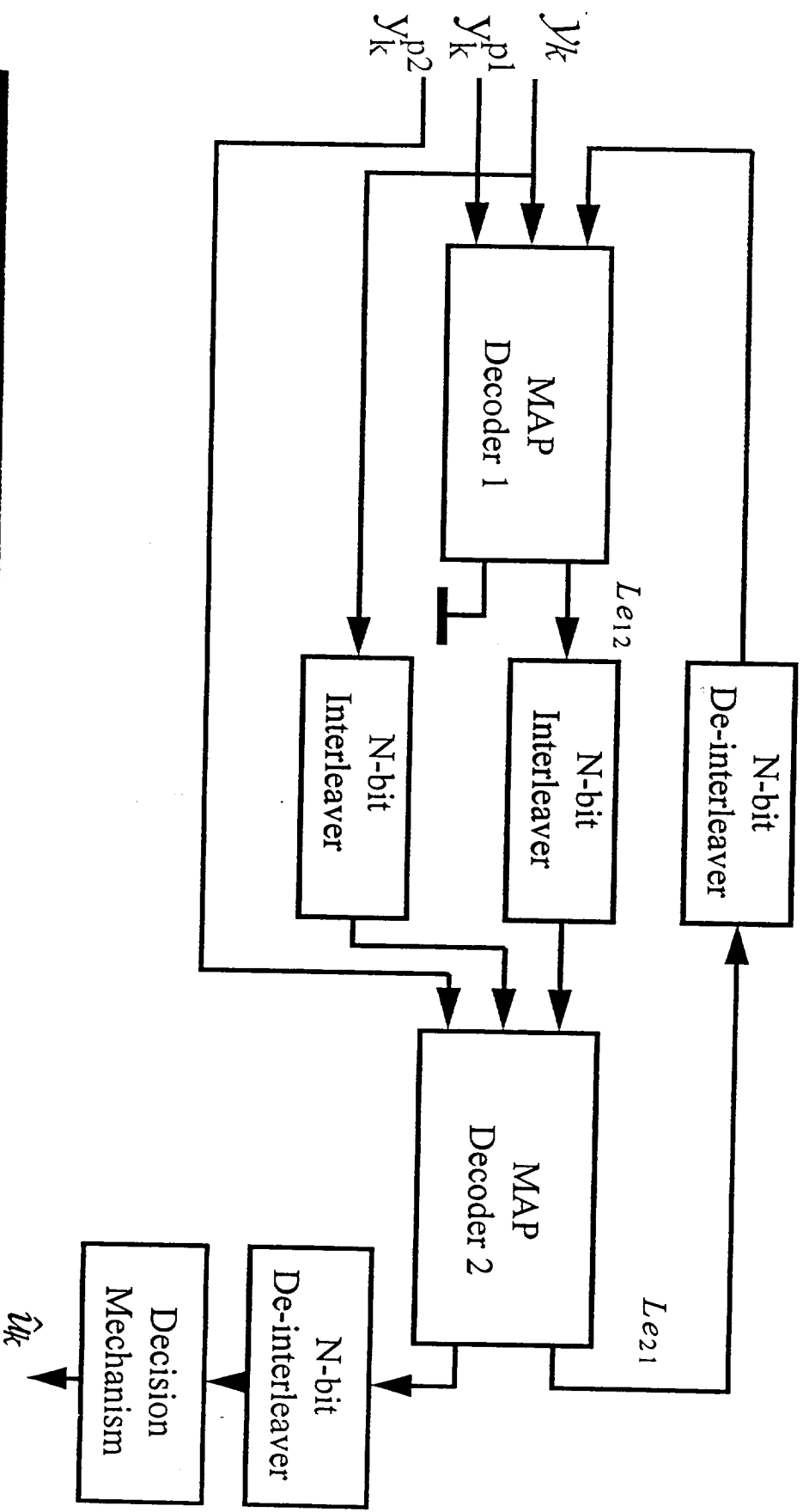


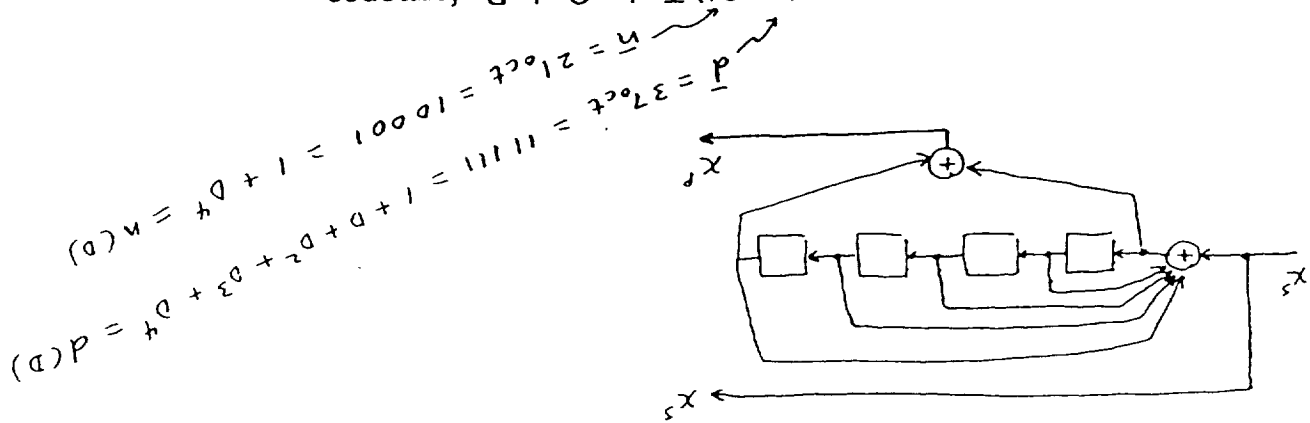
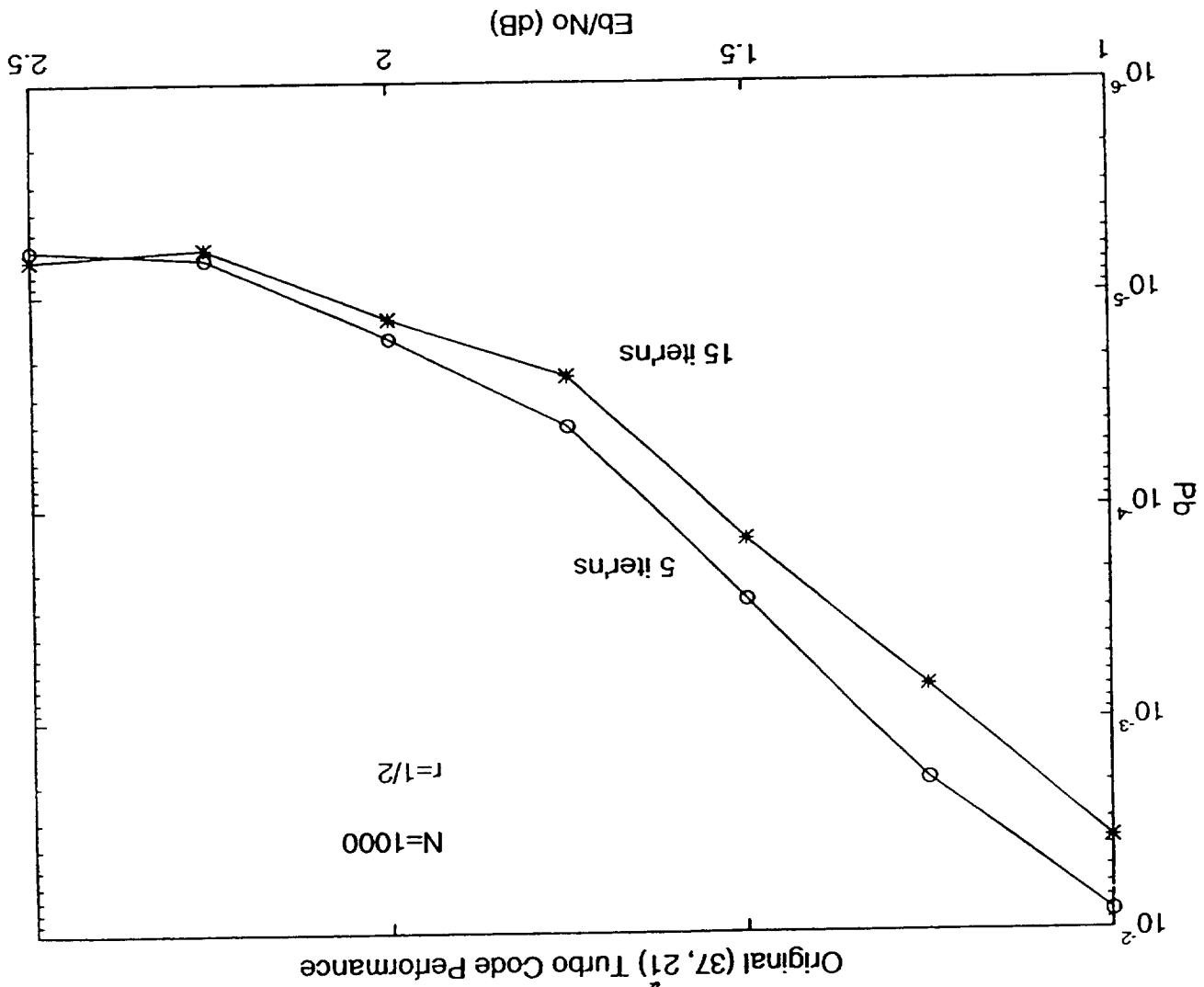
# TC ENCODER

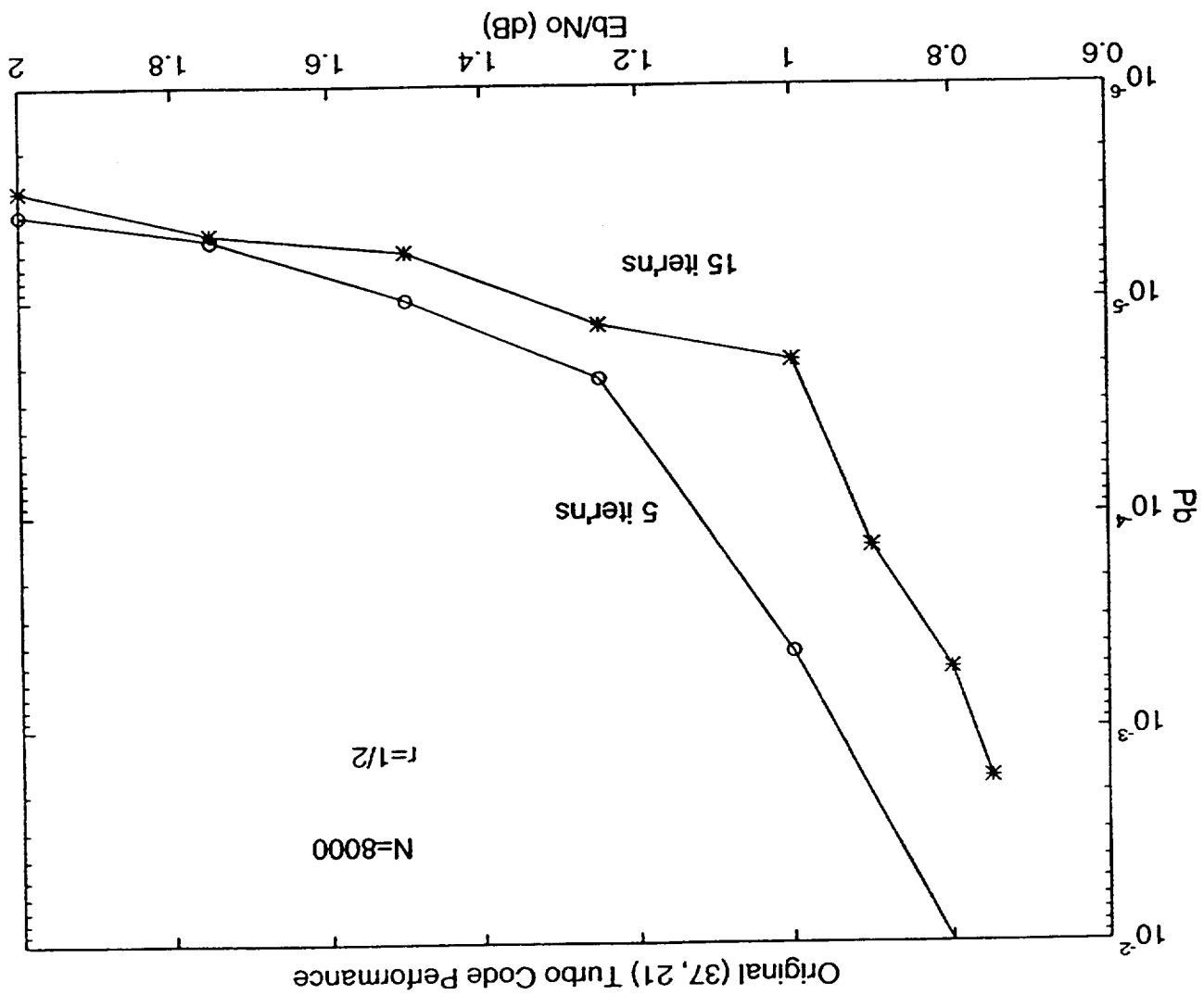




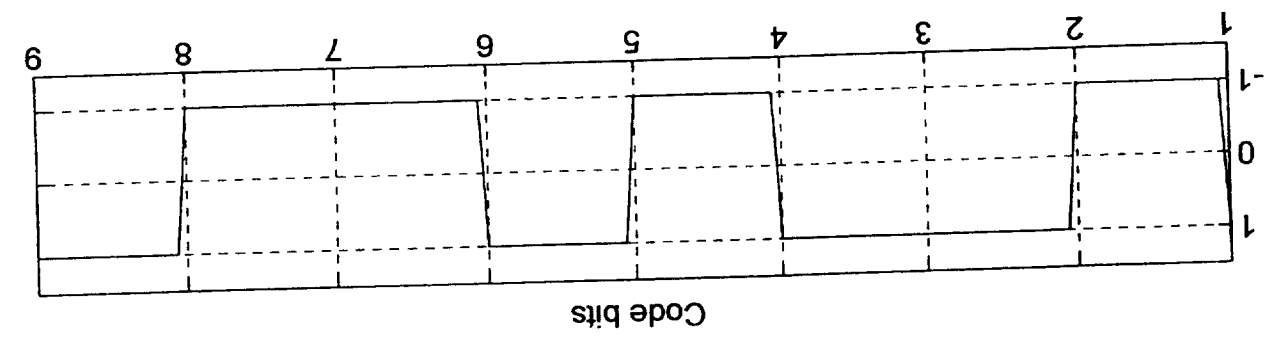
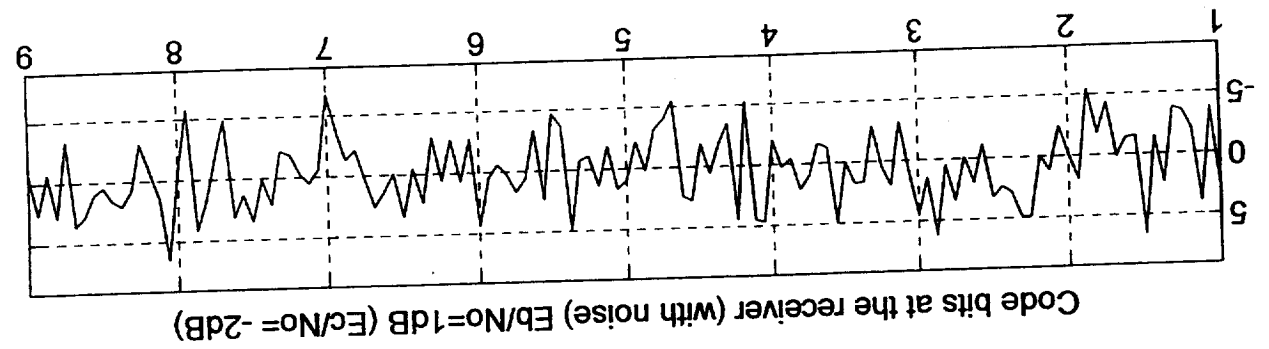
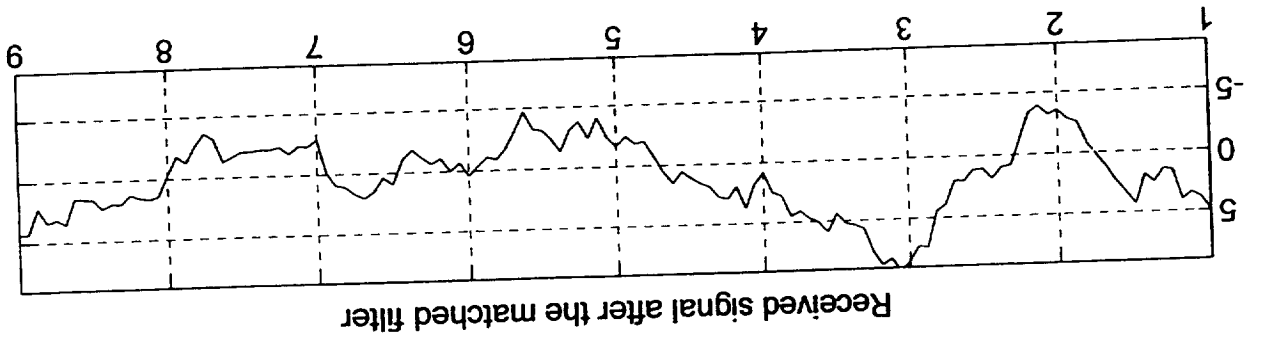
# ITERATIVE DECODER FOR TC









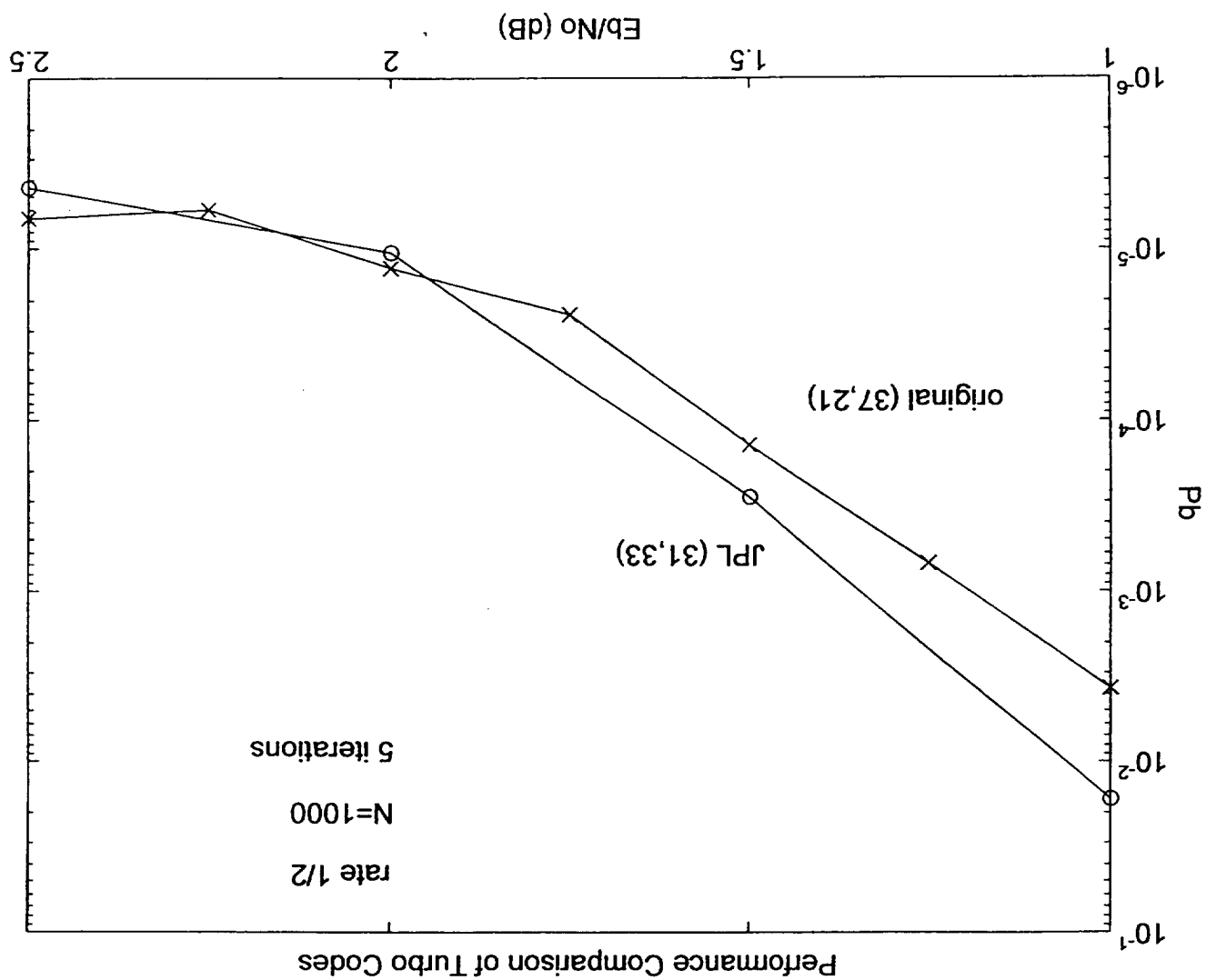


# **ON THE TRANSPARENCY OF TURBO CODES**

**William E. Ryan, Assistant Professor  
Omer Acikel, Ph.D. student**

**New Mexico State University**

**March 1997**



## **OUTLINE**

- I. Introduction to Transparent Codes
- II. Application to Turbo Codes
- III. Simulation Results

# I. INTRODUCTION TO TRANSPARENT CODES

## Notation

1. By  $u \leftrightarrow c$  we mean the message word  $u$  is mapped to the codeword  $c$  by the encoder.
2.  $u^c$  and  $c^c$  are the logical complements of  $u$  and  $c$  (binary codes are assumed).

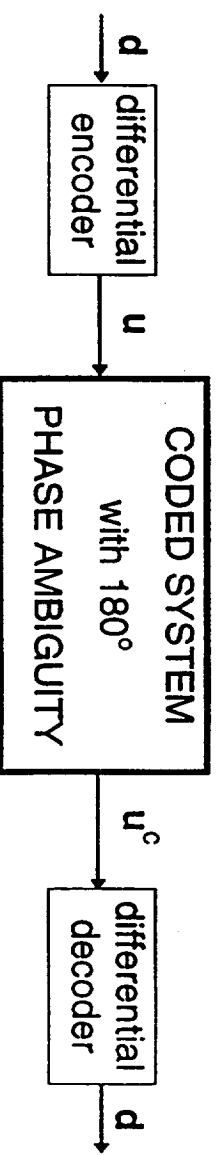
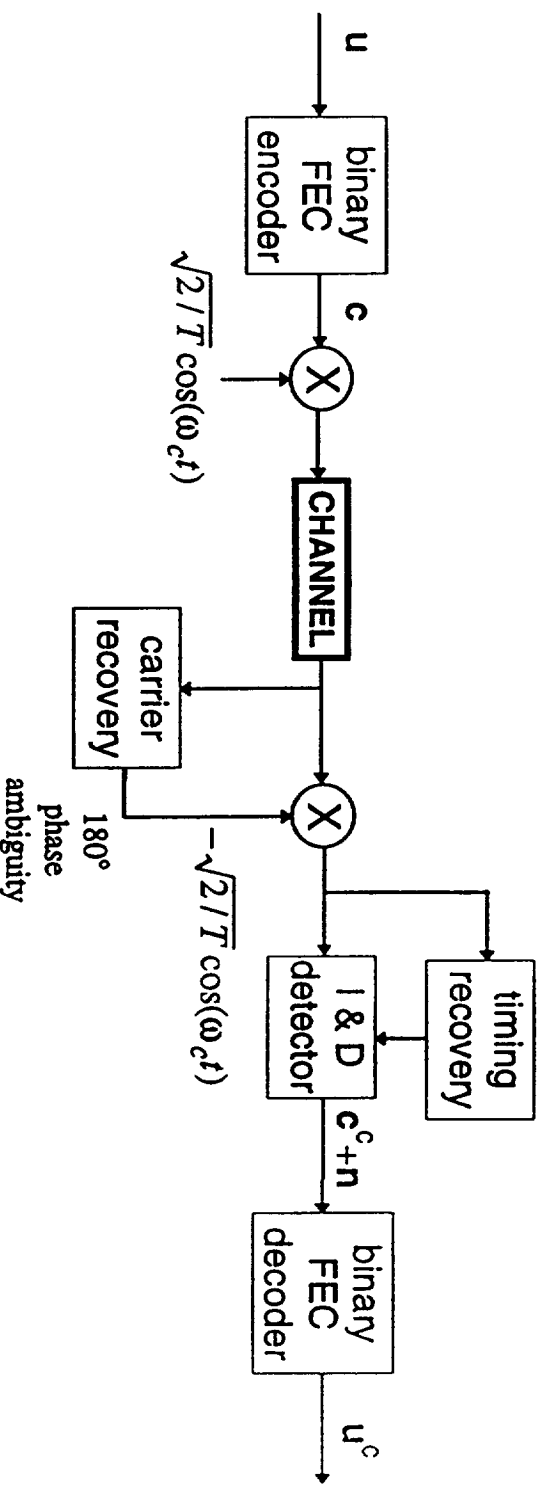
## Definition

A code is *transparent* if  $u \leftrightarrow c$  implies  $u^c \leftrightarrow c^c$ .

- The utility of transparent codes -- also called rotationally invariant codes -- arises as a result of the ambiguous lock phenomenon of carrier recovery circuits.

# I. INTRODUCTION TO TRANSPARENT CODES (cont'd)

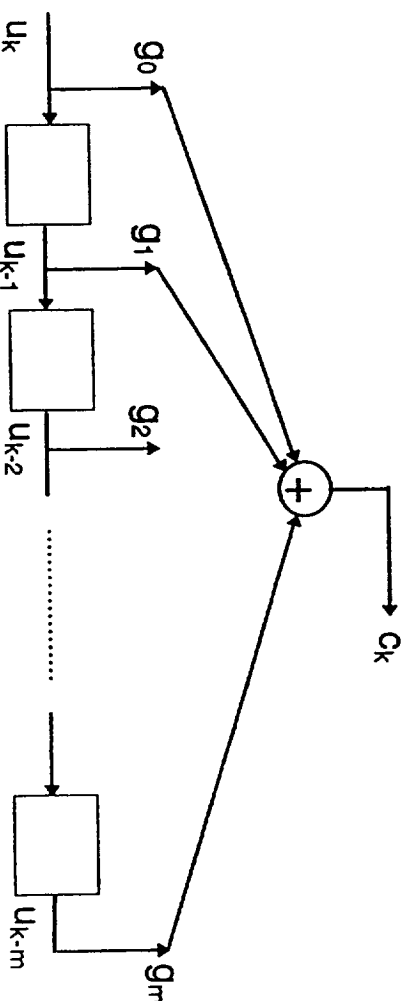
## Illustration



## II. APPLICATION TO TURBO CODES

(These results are due to Prof. Steve Wilson of the University of Virginia)

### Review of Transparent *Non-Recursive* Convolutional Codes



RESULT: A rate  $1/n$  non-recursive convolutional code is transparent if and only if the number of taps in each of its  $n$  generators is odd, i.e.,  $w_H[g_i(D)]$  are all odd.

## II. APPLICATION TO TURBO CODES (cont'd)

PROOF: When  $\mathbf{u}$  is input, the encoder output is

$$c_k = \sum_{j=0}^m u_{k-j} g_j .$$

When  $\mathbf{u}^c$  is input, the encoder output is

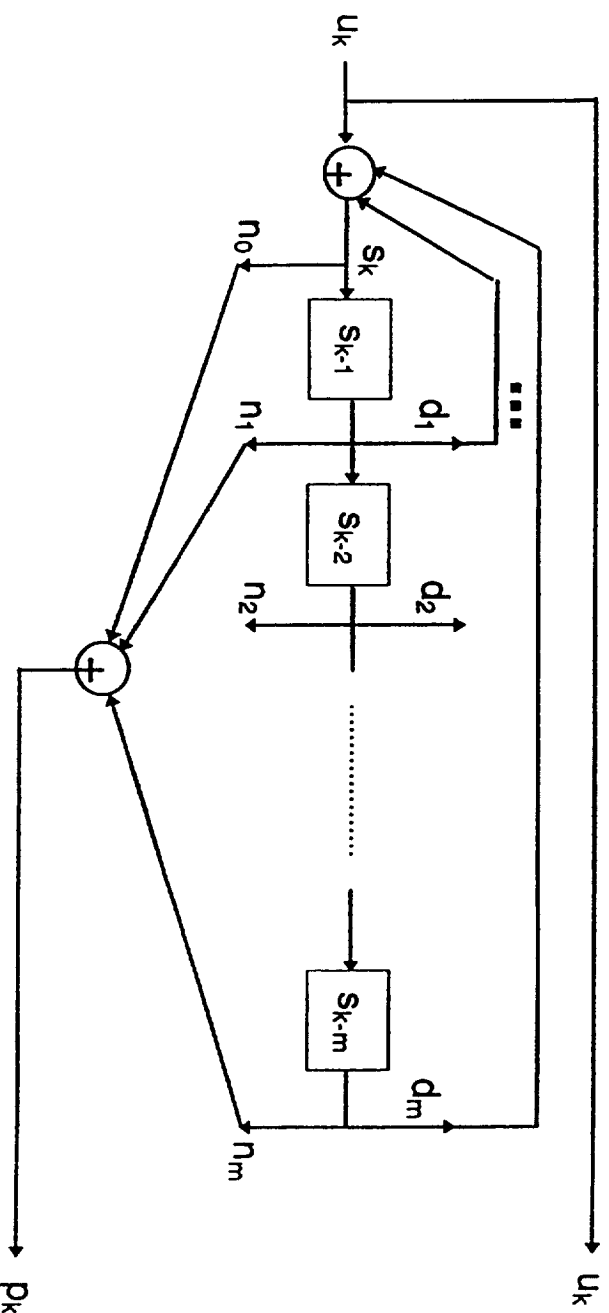
$$\begin{aligned} c'_k &= \sum_{j=0}^m u_{k-j}^c g_j \\ &= \sum_{j=0}^m (u_{k-j} + 1) g_j \\ &= \sum_{j=0}^m u_{k-j} g_j + \sum_{j=0}^m g_j \\ &= c_k + 1 \quad \text{iff the number of nonzero } g_j \text{ is odd} \\ &= c_k^c \quad \text{iff the number of nonzero } g_j \text{ is odd} \end{aligned}$$

■



## II. APPLICATION TO TURBO CODES (cont'd)

### The Result for *Recursive Convolutional Codes*



RESULT: A recursive convolutional code cannot be transparent, but can be “quasi-transparent” in the sense that  $(\mathbf{u}, \sigma_0) \leftrightarrow c$  implies  $(\mathbf{u}', \sigma_0') \leftrightarrow c'$  ( $\sigma_0$  is the initial state of the encoder). The code will be quasi-transparent whenever both  $w_H[n(D)]$  and  $w_H[d(D)]$  are odd.

## II. APPLICATION TO TURBO CODES (cont'd)

PROOF: We note that when  $\mathbf{u}'$  is input:

1. the systematic part of the codeword  $\mathbf{c} = [\mathbf{u} \mid \mathbf{p}]$  is automatically complemented.
  2. the parity is complemented if and only if  $(s_k, s_{k-1}, \dots, s_{k-m})$  is complemented and  $w_H[n(D)]$  is odd (from nonrecursive result).
- Thus, when  $w_H[n(D)]$  is odd,  $p_k$  is complemented if  $s_k$  and the state  $\sigma_k = (s_{k-1}, s_{k-2}, \dots, s_{k-m})$  are complemented, for all  $k \geq 0$ .
  - But,  $s_k$  and  $\sigma_k = (s_{k-1}, s_{k-2}, \dots, s_{k-m})$  will be complemented, for all  $k \geq 0$ , if:
    1. the encoder initialized to  $\sigma_0^c$
    2.  $w_H[d(D)]$  is odd

## II. APPLICATION TO TURBO CODES (cont'd)

*Proof of previous bullet:* When  $\mathbf{u}$  is input,

$$s_k = u_k + \sum_{j=1}^m s_{k-j} d_j .$$

When  $\mathbf{u}^c$  is input and the encoder is initialized to  $\sigma_0^c = (s_{-1}^c, s_{-2}^c, \dots, s_{-m}^c)$ ,

$$s'_0 = u_0^c + \sum_{j=1}^m s_{-j}^c d_j$$

$$= u_0 + 1 + \sum_{j=1}^m (s_{-j} + 1) d_j$$

$$= u_0 + \sum_{j=1}^m s_{-j} d_j + 1 + \sum_{j=1}^m d_j$$

$$= u_0 + \sum_{j=1}^m s_{-j} d_j + \sum_{j=0}^m d_j$$

## II. APPLICATION TO TURBO CODES (cont'd)

$$\begin{aligned}
 &= s_0 + 1 \quad \text{iff } w_H[d(D)] \text{ is odd} \\
 &= s_0^c \quad \text{iff } w_H[d(D)] \text{ is odd}
 \end{aligned}$$

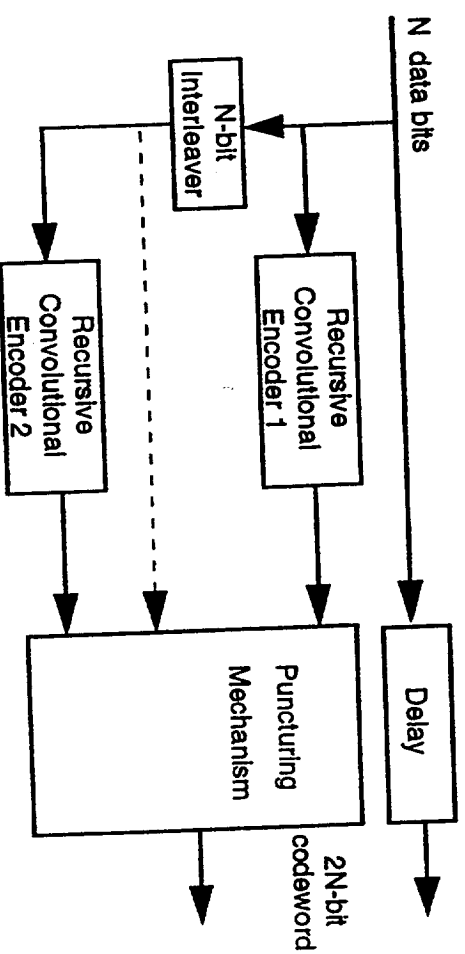
Thus,  $s_0, s_{-1}, \dots, s_{-m}$  will each be complemented so that  $\sigma_1 = (s_0, s_{-1}, \dots, s_{-m+1})$  will be complemented (relative to when  $\mathbf{u}$  is input).

Following, identical steps, can show (inductively) that  $s_k$  and  $\sigma_k = (s_{k-1}, s_{k-2}, \dots, s_{k-m})$  will be complemented, for all  $k \geq 0$ , provided the encoder initialized to  $\sigma_0^c$  and  $w_H[d(D)]$  is odd .



## II. APPLICATION TO TURBO CODES (cont'd)

### Implications for Turbo Codes



Because turbo codes employ two (or more) recursive convolutional codes, a turbo code is also quasi-transparent.

## II. APPLICATION TO TURBO CODES (cont'd)

Thus, on channels with ambiguities, the decoder may see the situation  $(\mathbf{u}, \sigma_0) \leftrightarrow \mathbf{c}$  (when properly locked) or the situation  $(\mathbf{u}', \sigma_0') \leftrightarrow \mathbf{c}'$  (when locked 180° out of phase).

As a result, the constituent decoders must initialize their state probabilities to account for the 50% chance of locking out of phase and the encoder appearing to start in the all-ones state:

$$\begin{aligned} D1: \alpha_0^{(1)}(s) &= 0.5, \quad \text{for } s = 0 \text{ and } 2^m - 1 \\ &= 0, \quad \text{else} \\ \beta_N^{(1)}(s) &= 0.5, \quad \text{for } s = 0 \text{ and } 2^m - 1 \\ &= 0, \quad \text{else} \end{aligned}$$

## II. APPLICATION TO TURBO CODES (cont'd)

$$\begin{aligned} \text{D2: } \alpha_0^{(2)}(s) &= 0.5, \quad \text{for } s = 0 \text{ and } 2^m - 1 \\ &= 0, \quad \text{else} \end{aligned}$$

$$\beta_N^{(2)}(s) = \alpha_N^{(2)}(s), \quad \text{for all } s$$

We can expect some degradation relative to the ideal due to decoder's uncertainty in the encoder's initial state.

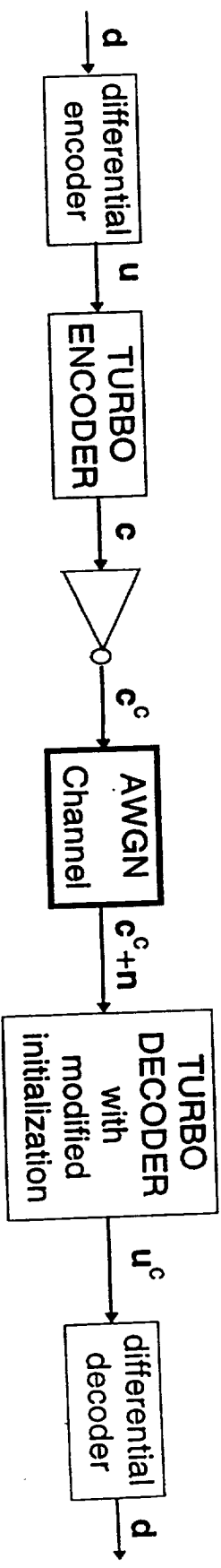
### III. SIMULATION RESULTS

A turbo code with  $w_H[n(D)]$  and  $w_H[d(D)]$  odd was selected from "Design of parallel concatenated codes," by S. Benedetto and G. Montorsi, *IEEE Trans. Comm.*, May 1996:

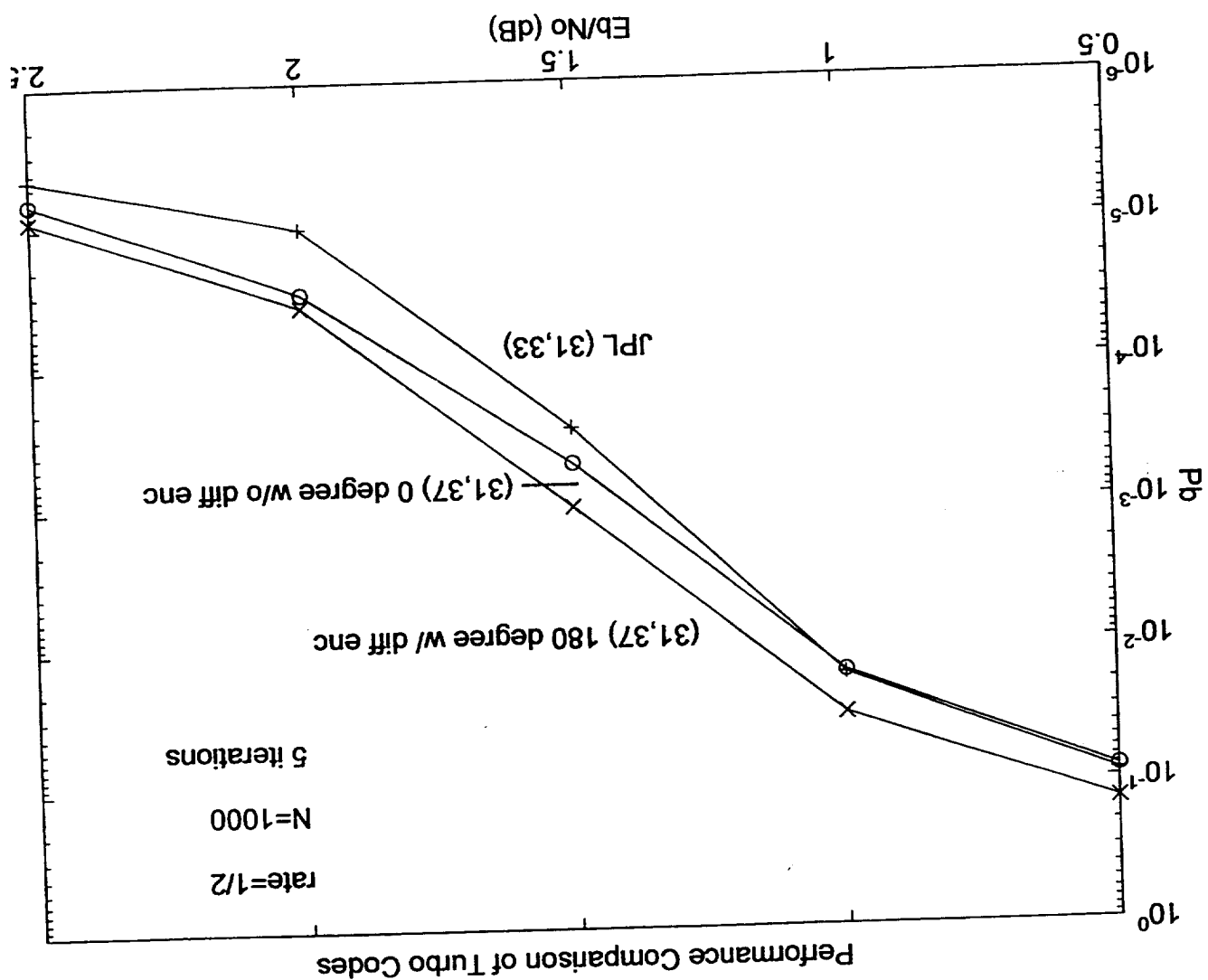
*Comm.*, May 1996:

$$d(D) = 31_{\text{octal}} \quad n(D) = 37_{\text{octal}} \quad (16 \text{ states})$$

#### *Simulation Model*

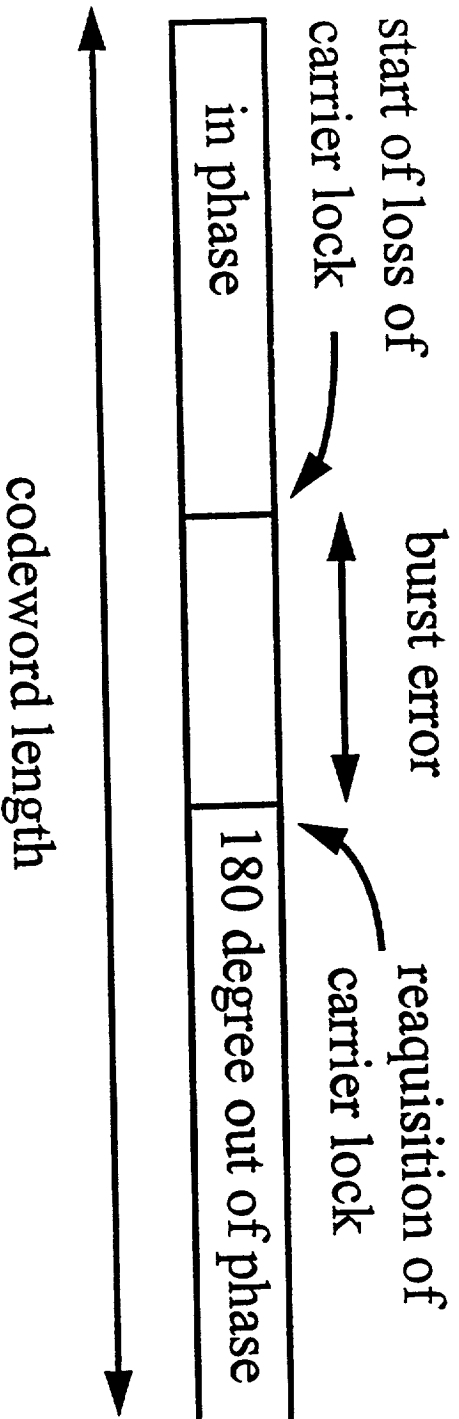








# BURST ERROR RESULTS with The (31,37) TRANSPARENT CODE



pattern1:

data(d)	parity(p)
---------	-----------

pattern2: dpdpdp.....





# BURST ERROR RESULTS with The (31,37) TRANSPARENT CODE (cont'd)

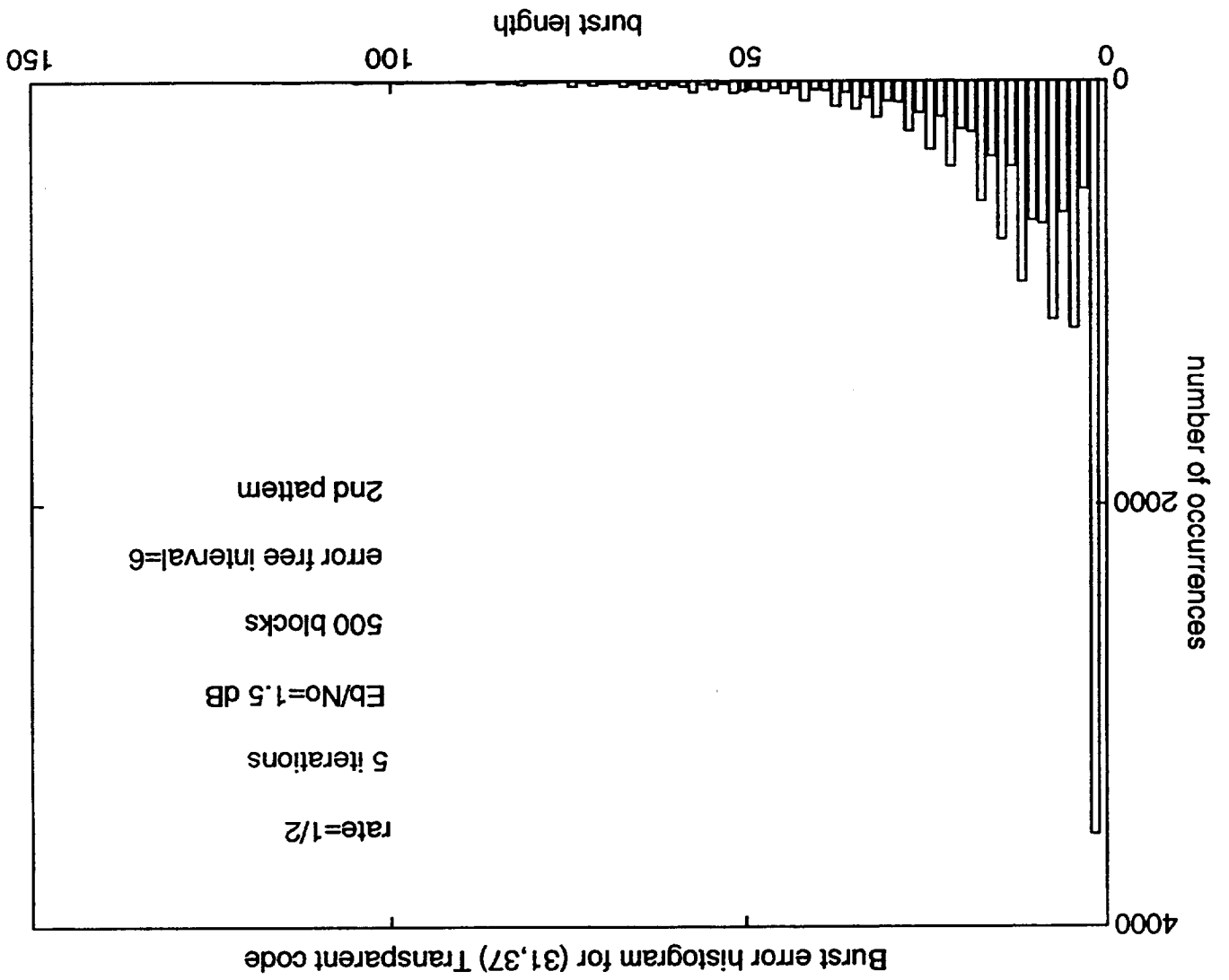
\*N=1000,

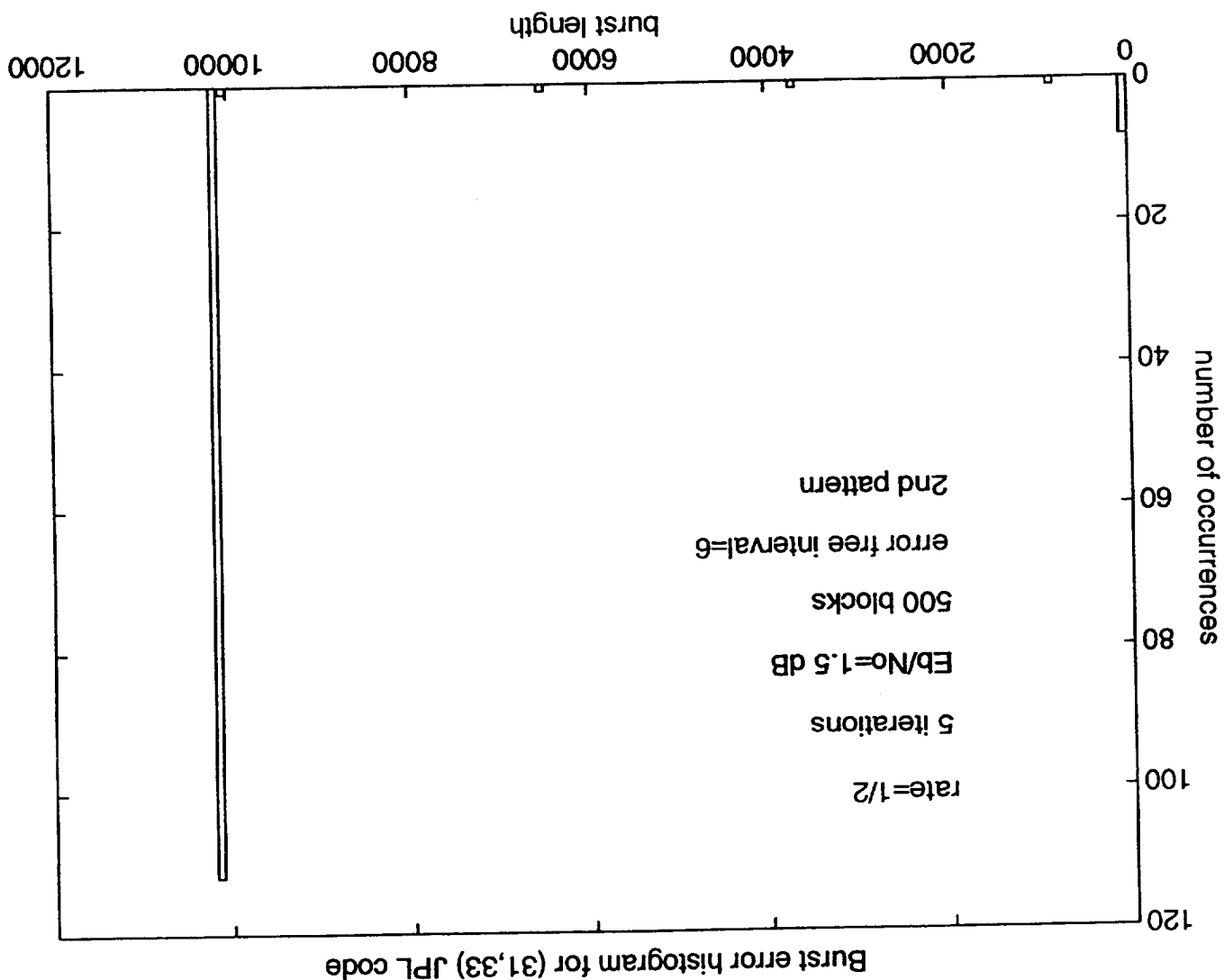
\*Eb/No=2dB,

\*5 iterations

\*for 500 error events

Pattern / # of burst err.	# of errors	Pe (bit) 1e-2	PB (sym) 1e-2	# of blocks
1st pattern / 0 burst error	106404	2.12	6.22	503
1st pattern / 100 burst errors	102801	2.31	6.57	500
2nd pattern / 0 burst error	89950	1.77	5.75	507
2nd pattern / 100 burst errors	98716	1.97	5.63	500





# **PERFORMANCE OF TURBO CODES IN PULSED CW RFI**

**William E. Ryan, Assistant Professor  
New Mexico State University**

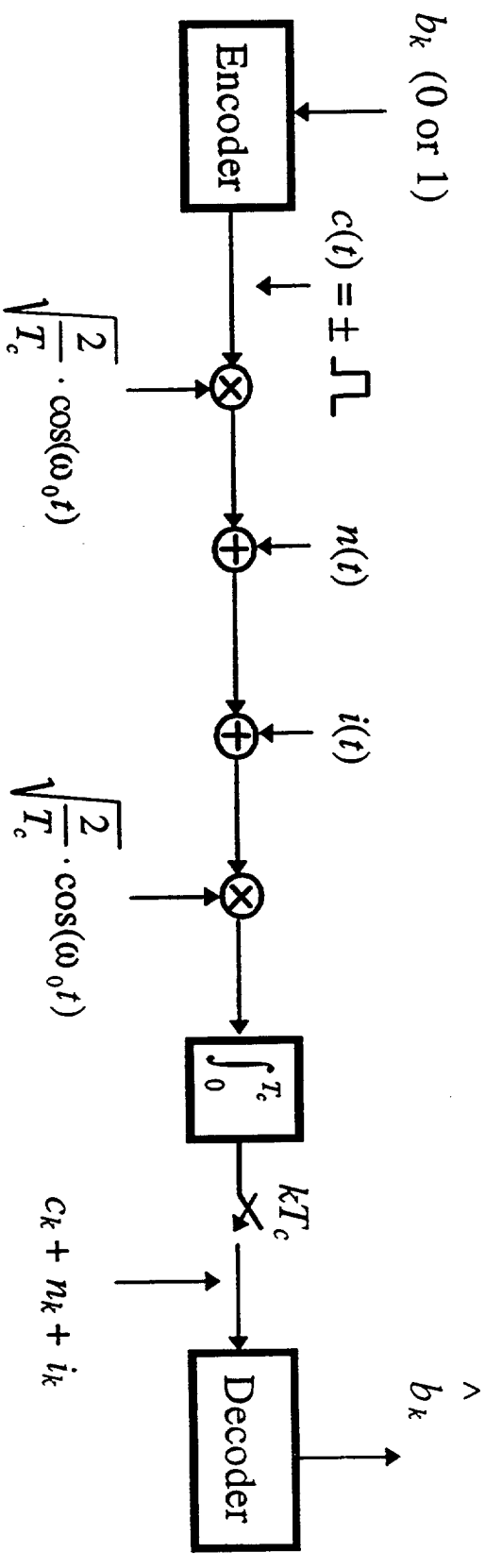
**March 1997**

## **OUTLINE**

- I. RFI Model
- II. Simulation Results
- III. Future Work

# I. RFI MODEL

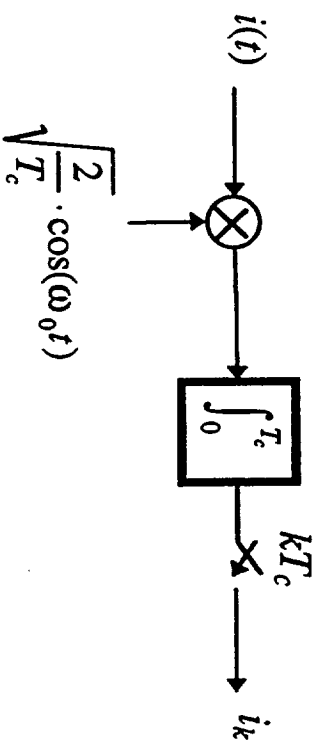
## Model of turbo-coded BPSK channel in AWGN and RFI



$$\begin{aligned} \text{where, for CW RFI, } i(t) &= a \cdot \cos(\omega_f t + \phi) & \text{for } \tau_1 < t < \tau_2 \\ &= 0 & \text{otherwise} \end{aligned}$$



## RFI component at sampler output



$$i_k = \int_{\max(\tau_1, kT_c)}^{\min(\tau_2, (k+1)T_c)} a \cdot \cos(\omega_I t + \phi) \cdot \sqrt{2/T_c} \cos(\omega_0 t) dt$$

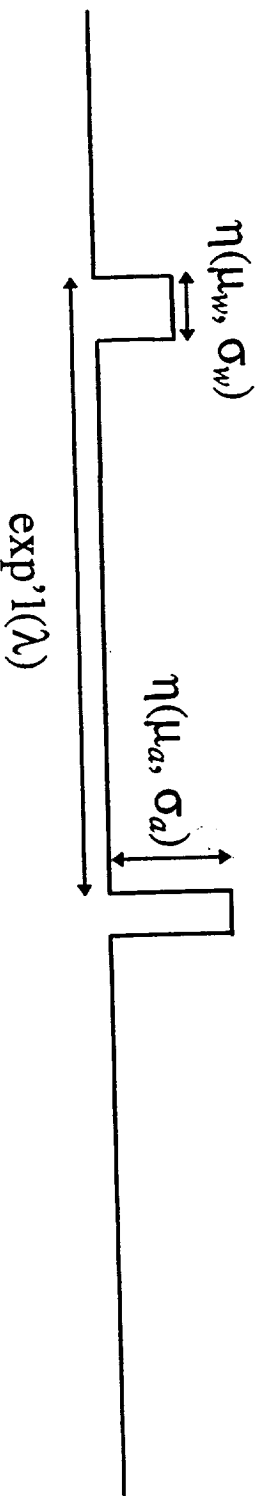
$$= \frac{a}{2\Delta\omega} \sqrt{2/T_c} \sin(\Delta\omega t + \phi) \Big|_{\max(\tau_1, kT_c)}^{\min(\tau_2, (k+1)T_c)}$$

where  $\Delta\omega \equiv (\omega_I - \omega_0)$  and  $-2\pi/T_c \leq \Delta\omega \leq 2\pi/T_c$ .

## I. RFI MODEL (cont'd)

### RFI statistics

1. Poisson arrivals  $\Rightarrow$  exponential inter-arrival times:  $\exp'1(\lambda)$
2. Gaussian widths (durations):  $\eta(\mu_w, \sigma_w)$
3. Gaussian amplitudes:  $\eta(\mu_a, \sigma_a)$
4.  $\Delta\omega$  is uniformly distributed on the interval  $(-2\pi / T_c, 2\pi / T_c)$



## II. SIMULATION RESULTS

### Turbo Code

JPL proposed (31, 33) turbo code - rate 1/2

N = 10200

5 iterations

### RFI

RFI amplitude selected for SIR = -20, -10, and 0 dB

- Ex:  $\mu_a = \sqrt{200} \Rightarrow \text{SIR}_{\text{ave}} = \frac{1}{(\sqrt{200})^2 / 2} = 0.01 = -20 \text{ dB}$

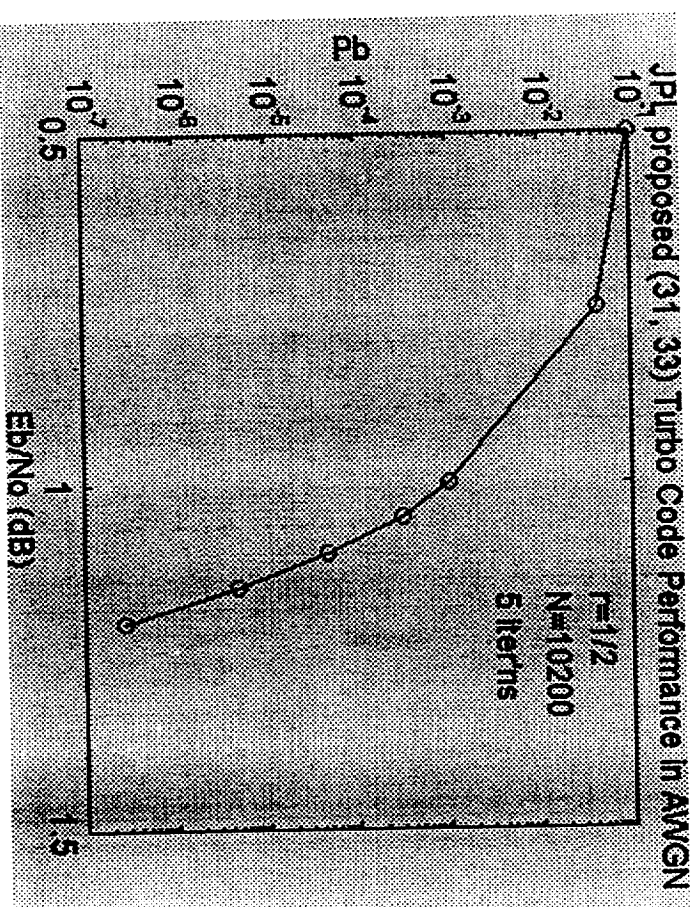
$$\sigma_a = 2$$

RFI width

$$\mu_w = 5\sigma_w \text{ so that } \Pr\{w < 0\} = Q(5) \approx 3\text{e-}7$$

## II. SIMULATION RESULTS (cont'd)

AWGN baseline: 5 iterations

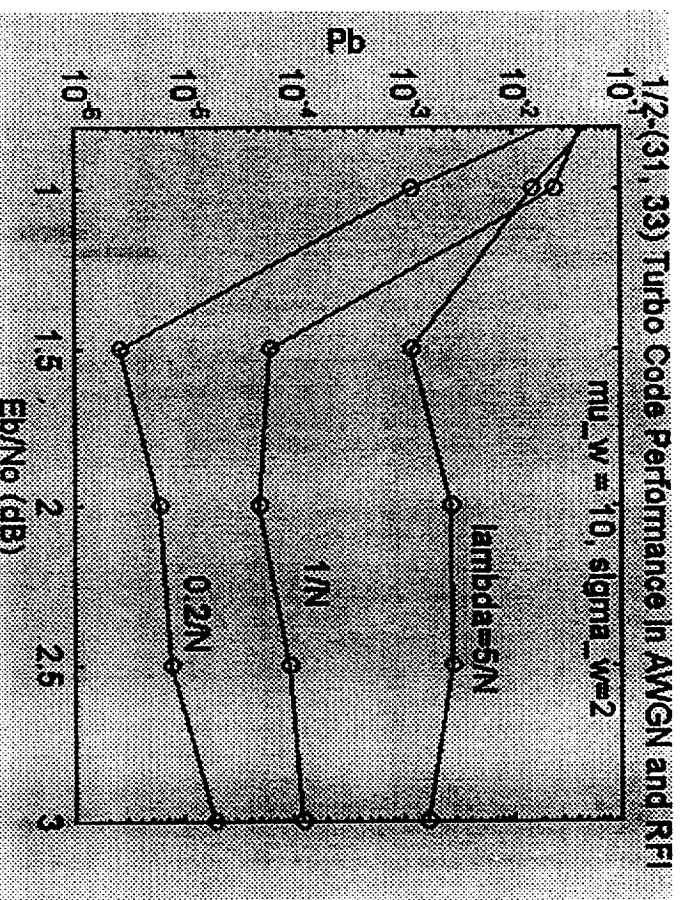


## II. SIMULATION RESULTS (cont'd)

AWGN + RFI (5 iterations): **SIR = - 20 dB**

fix  $\mu_w = N/1000 = 10$ ,  $\sigma_w = \mu_w / 5 = 2$

vary  $\lambda = 5/10000, 1/10000, 0.2/10000$

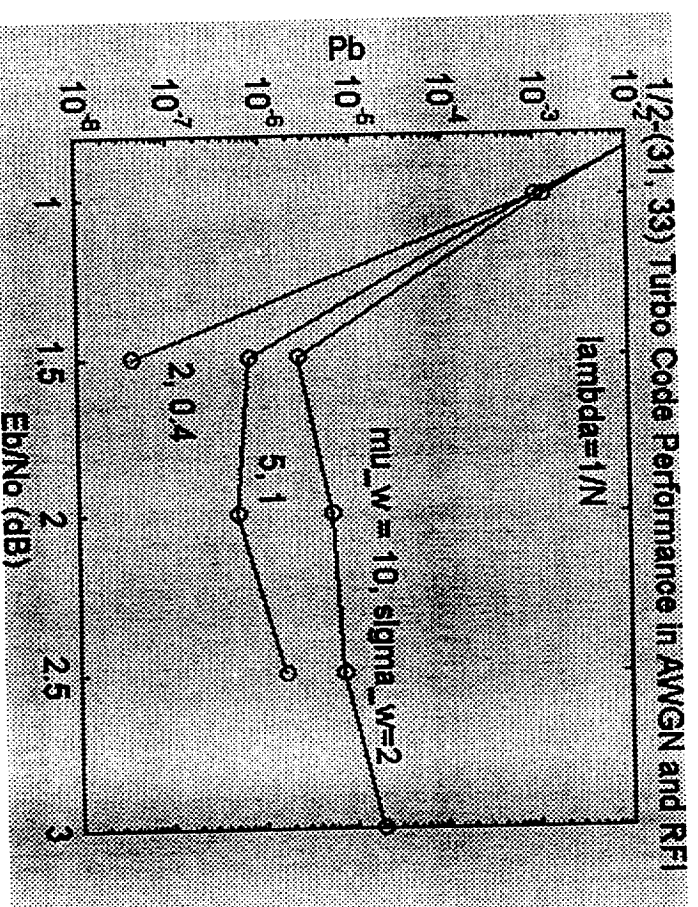


## II. SIMULATION RESULTS (cont'd)

AWGN + RFI (5 iterations): **SIR = - 20 dB**

fix  $\lambda = 1/10000$

vary  $\mu_w = 10, 5, 2$  ( $\sigma_w = \mu_w / 5$ )



## II. SIMULATION RESULTS (cont'd)

**AWGN + RFI (5 iterations): SIR = - 20 dB**

**fix  $\lambda = 0.2/10000$**

**vary  $\mu_w = 10, 2$  ( $\sigma_w = \mu_w / 5$ )**

**PERFORMANCE DEGRADATION IS NEGLIGIBLE**

## II. SIMULATION RESULTS (cont'd)

**AWGN + RFI (5 iterations): SIR = - 10 dB**

for  $\lambda = 5/10000, 1/10000, 0.2/10000$

and  $\mu_w = 10, 5, 2$  ( $\sigma_w = \mu_w / 5$ )

**PERFORMANCE DEGRADATION IS NEGLIGIBLE**

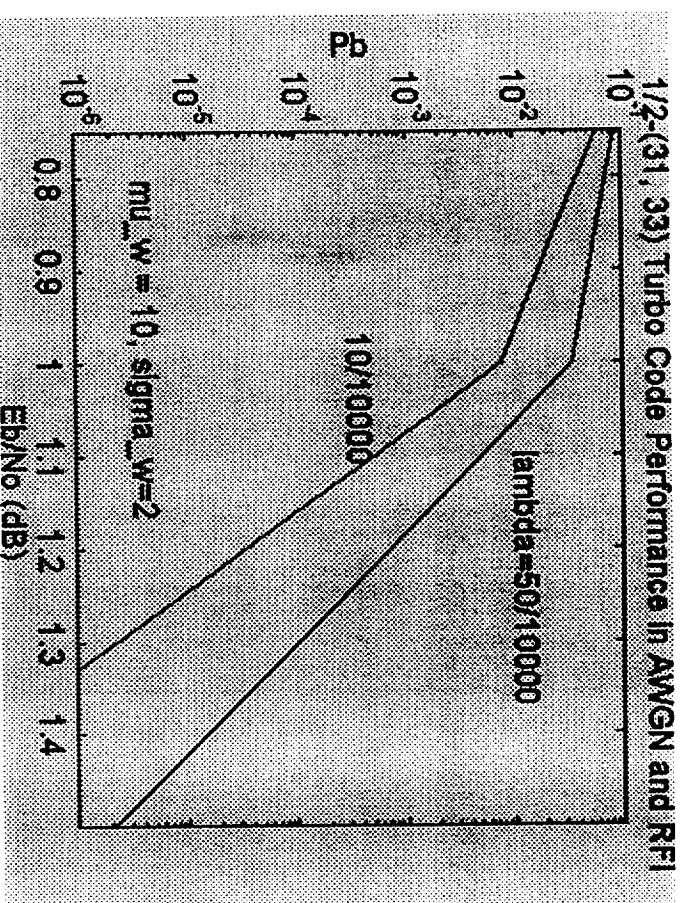


## II. SIMULATION RESULTS (cont'd)

AWGN + RFI (5 iterations): **SIR = - 10 dB**

for  $\lambda = 50/10000, 10/10000$

and  $\mu_w = 10$  ( $\sigma_w = \mu_w / 5$ )



## II. SIMULATION RESULTS (cont'd)

**AWGN + RFI (5 iterations): SIR = 0 dB**

**PERFORMANCE DEGRADATION IS NEGLIGIBLE FOR**

**for  $\lambda > 50/10000$  and  $\mu_w > 10$**



# *HIGH RATE TURBO CODES*

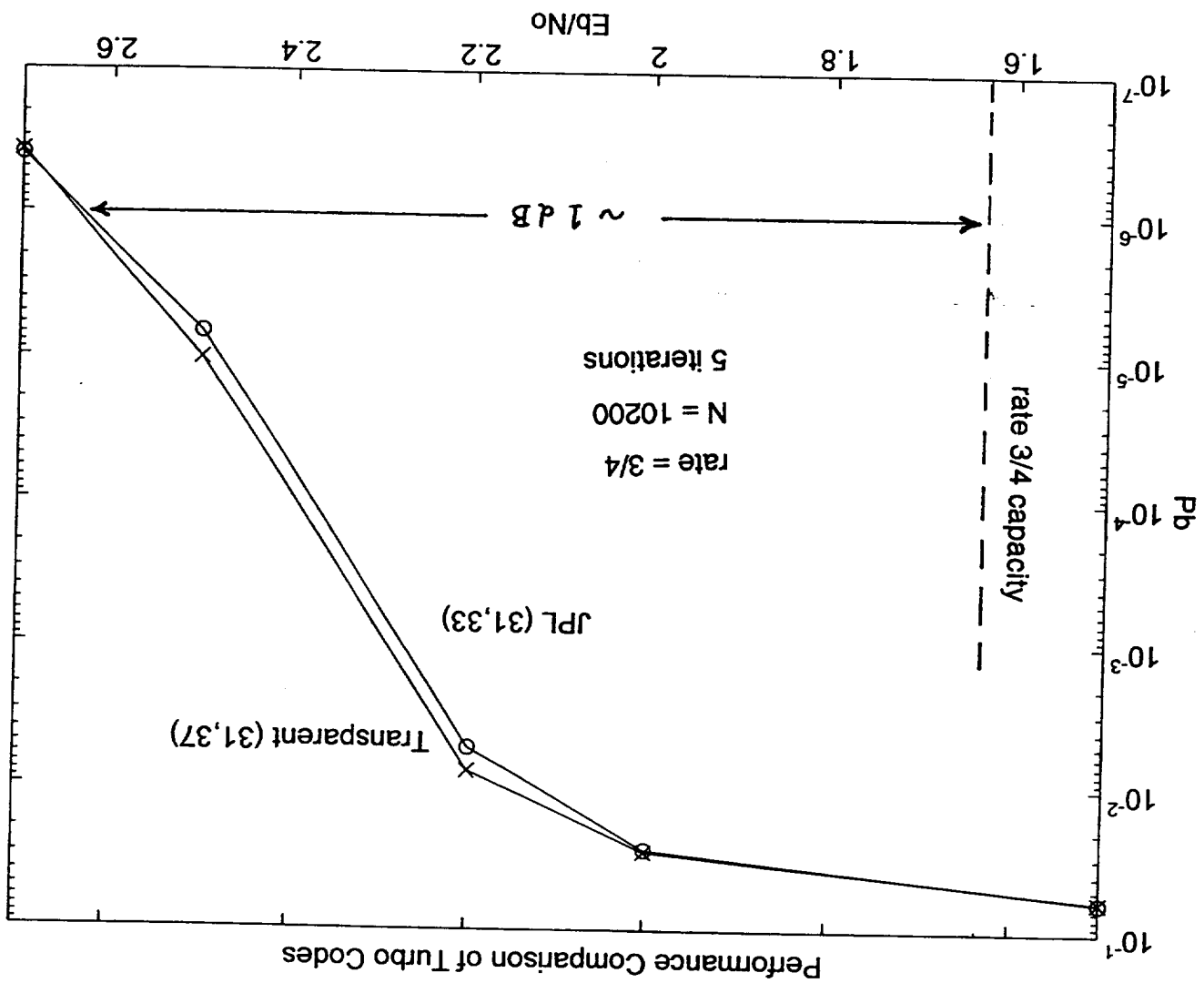
William E. Ryan, Asst. Prof., NMSU

Omer F. Acikel, Ph.D. Student, NMSU

March 11, 1997



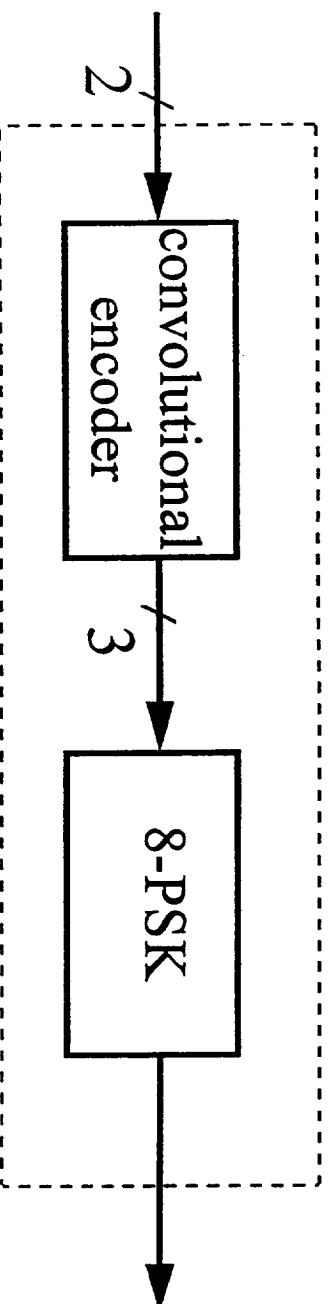
# Rate 3/4 Turbo Codes



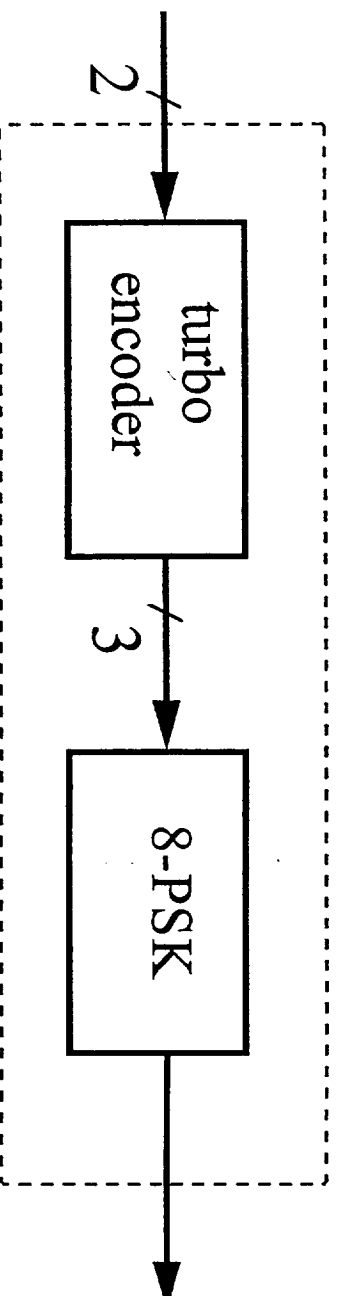


# INTRODUCTION TO TURBO-TRELLIS CODED MODULATION (T-TCM)

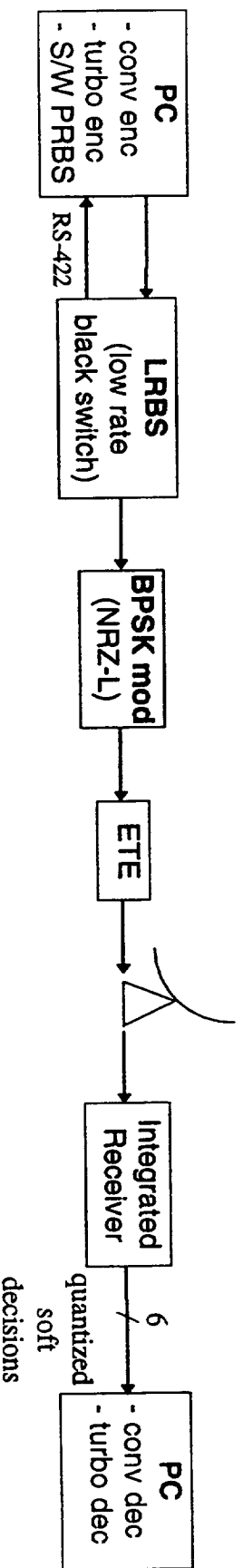
## 8-PSK TCM



## 8-PSK T-TCM



## TURBO CODE TEST THROUGH TDRS



- will run the  $r=1/2$ ,  $K=7$ , conv code for a baseline
- the turbo code will be the  $r=1/2$  (31,33) code proposed by JPL
- all encoders and decoders will be simulated in software
- software must be modified for block sync and to accept quantized soft decisions
- goal for first round of tests:  $P_b$  vs  $E_b/N_0$  curves down to about 1dB
- estimated time of completion for first round of tests is June 1997

# DAMA Carrier Acquisition

Phillip De León

*Assistant Professor*

*New Mexico State University*

*Klipsch School of Electrical and Computer Engineering  
Center for Space Telemetry and Telecommunications*

# Background of Investigation

- Simulations indicate up to a  $\pm 64\text{kHz}$  Doppler shift on Small Satellite carrier
- Current receiver can recover/demodulate carrier with up to a  $\pm 3\text{kHz}$  Doppler shift
- Nature of proposed DAMA service SN does not provide a priori knowledge of Doppler shift of carrier



# Requirements for DAMA Carrier Acquisition

- Front-end hardware unit to
  - detect DAMA carrier
  - correct for Doppler shift
- DAMA carrier detection and correction should keep pace with service request rate
- Acquisition of service request should have minimal impact to WSC facilities

# Proposed Solution

- Use windowed, discrete Fourier transform (DFT) to detect carrier (peak spectral detection)
- Pass frequency estimate to receiver for demodulation
- Implement solution using digital signal processing (DSP) hardware

# DAMA Carrier Acquisition

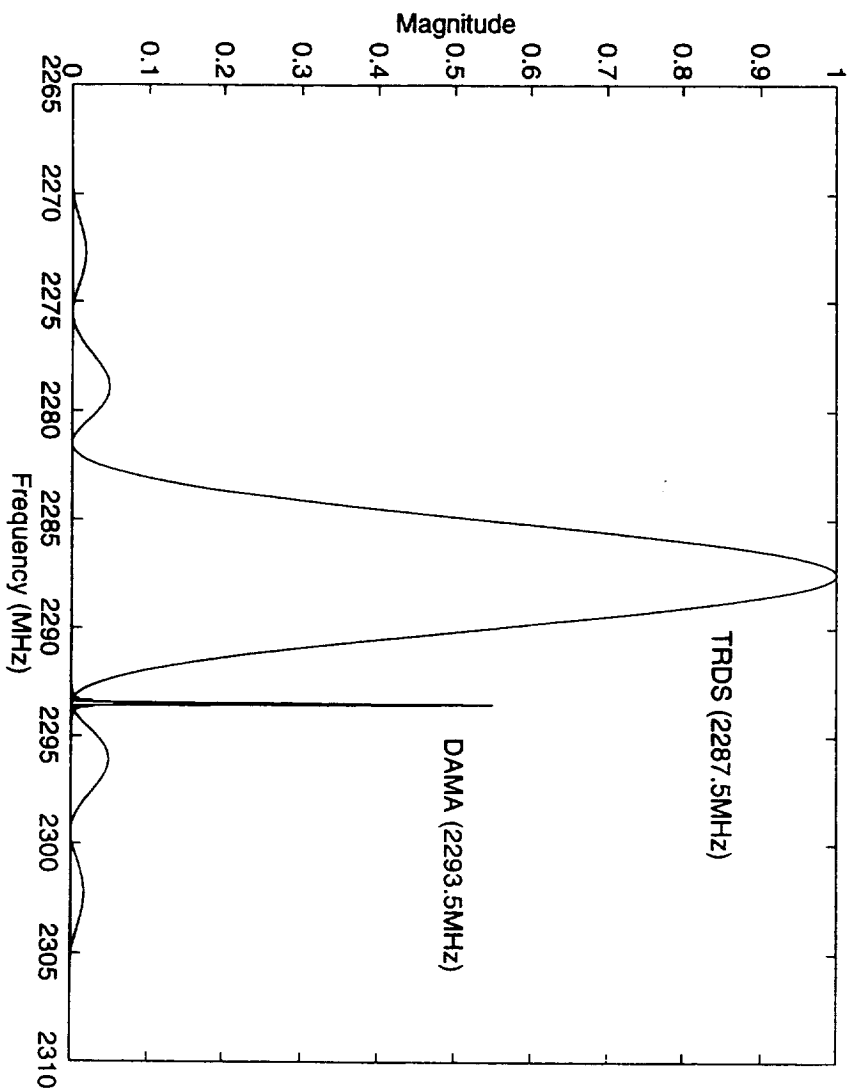
## Prototype Unit

- Hardware unit built around 40mips Motorola DSP56002 processor
- Reliable frequency estimation to within  $\pm 23\text{Hz}$  over a 24kHz bandwidth
- Reliable frequency estimation down to  $-4\text{dB SNR}$
- Approximately  $\sim 200\mu\text{s}$  for estimate (assuming prior data acquisition)

# Proposed Specifications

- DAMA carrier located in frequency at first null in TDRS spectrum (2287.5MHz + 6MHz)
- Main lobe bandwidth of 200kHz in DAMA spectrum

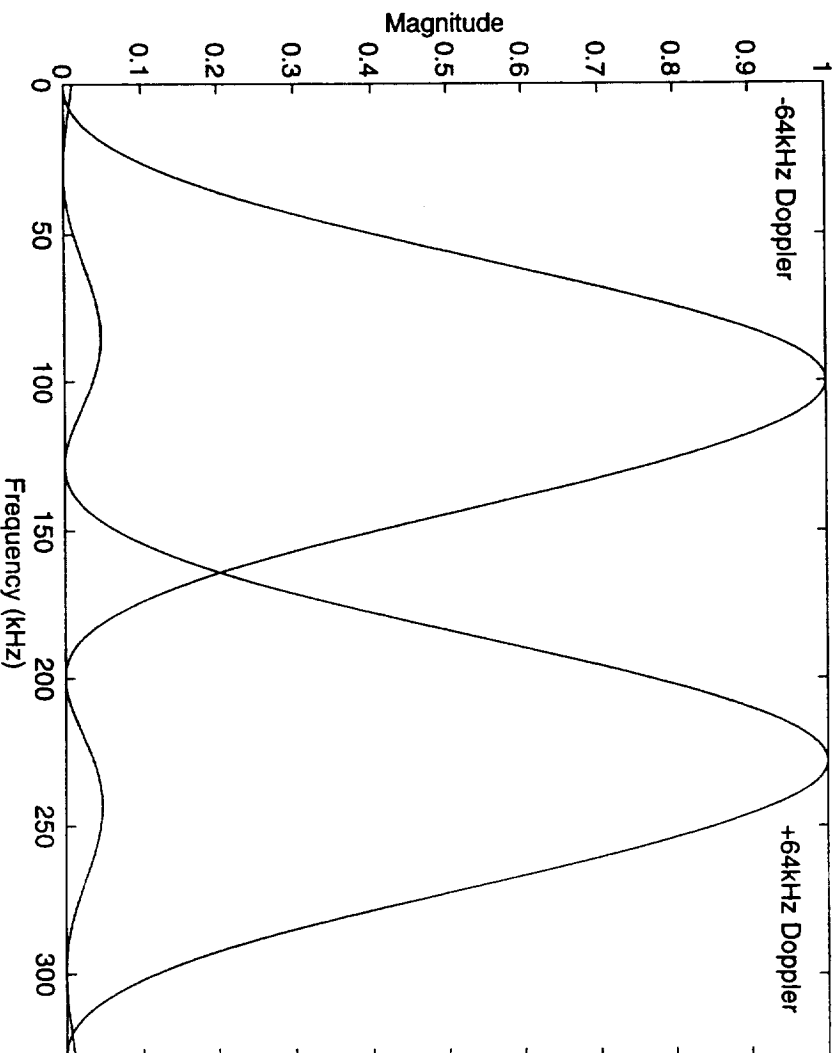
# DAMA Carrier Location



# Proposed DAMA Carrier Acquisition Unit

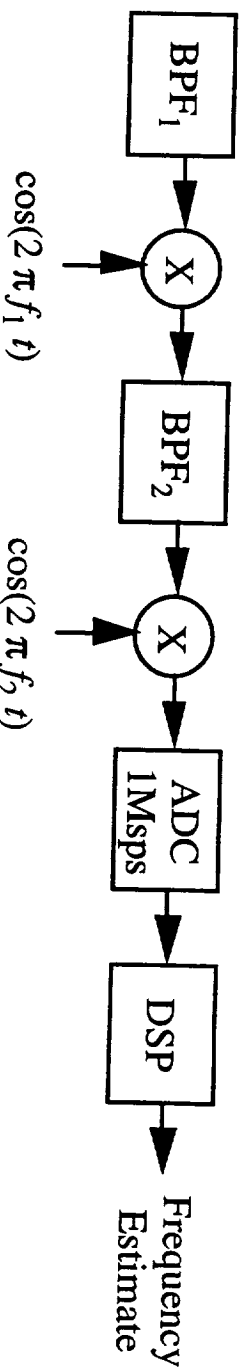
- Bandpass filter and frequency-shift DAMA signal spectrum
- Main lobe of shifted spectrum preserved in 328kHz-wide band

# Shifted DAMA Spectra



# Proposed DAMA Carrier Acquisition Unit (con't)

- $\pm 3\text{kHz}$  estimation accuracy requires 256-point DFT (1MHz A/D)
- Execution rate demands
  - $\sim 100\text{mips}$  (100MHz Motorola DSP56301)
  - 64k external SRAM





# Proposed DAMA Carrier Acquisition Unit (con't)

- Frequency estimate delivered to receiver as
  - locking tone at estimated frequency (or some fraction of it)
  - voltage proportional to estimated frequency for driving a VCO

# Further Investigations

- Exploit knowledge of DAMA signal characteristics
  - more efficient frequency estimation, i.e. “modern spectral estimation”
  - (adaptive) parametric models (AR models) of DAMA carrier
- Service multiple DAMA requests

# Conclusions

- Problem is well-suited to DSP-based implementation
- Prototype hardware provides accurate, real-time frequency estimates
- Design of DAMA Carrier Acquisition Unit based on prototype and simulation data

# **Non-linear Equalization of Non-linear Satellite Channels**

**James P. LeBlanc and William E. Ryan**

**Klipsch School of ECE  
New Mexico State University**

## Goals of Effort

- Increase data rate through TDRSS
  - Use of higher order modulation schemes
  - Higher symbol rates

## Methods

- Reduction of non-linear ISI due to TWT
- Use of adaptive non-linear equalization

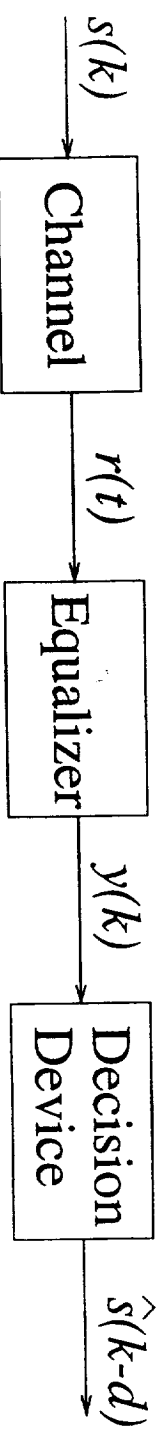
### Related Publications

- A. Gutierrez, W. Ryan *Equalization and Detection for Digital Communication over Nonlinear Bandlimited Satellite Communication Channels*, NMSU-ECE-95-008, 1995.
- T. J. Wolcott, W. P. Osborne, *Uplink-Noise Limited Satellite Communications Systems*, NMSU-ECE-95-014, 1995.
- W. P. Osborne, T. J. Wolcott, *Performance Evaluation of 8PSK at 450 MBPS Through TDRSS*, NMSU-ECE-95-004, 1995.

### Related Efforts

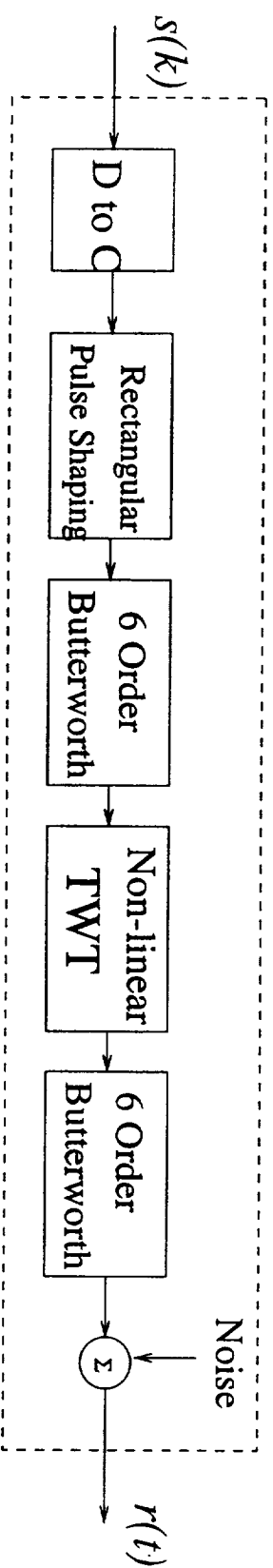
- Data acquisition (snapshots) of actual TWT channels from Loral
- In-house TWT channel implementation
- Loral has installed *advanced modem* at TDRSS ground station at WSMR incorporating nonlinear equalizer and higher order constellations

## System Block Diagram



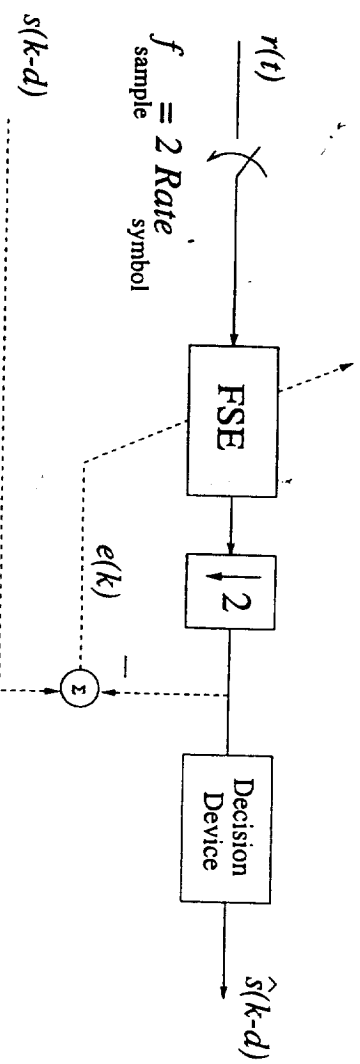
- *Channel* introduces noise and intersymbol interference (ISI)
- *Equalizer* reduces ISI (w/o excessive noise enhancement) ...
- ... allowing *Decision Device* to estimate transmitted symbol

## Baseband Channel Model

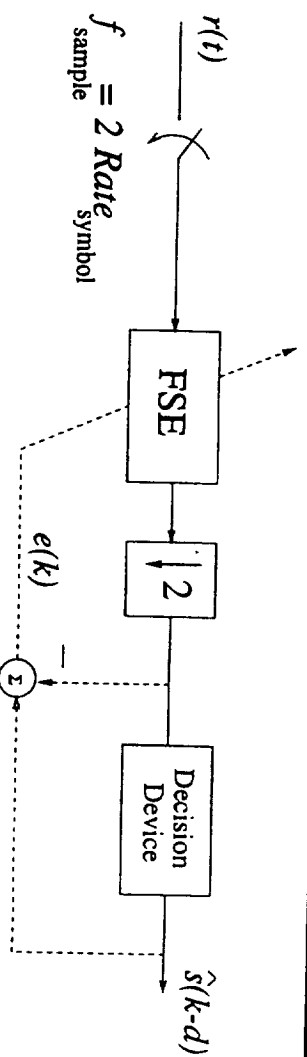




## LMS Feedforward Linear Eq. (using training)



## Decision Directed Feedforward Linear Eq.

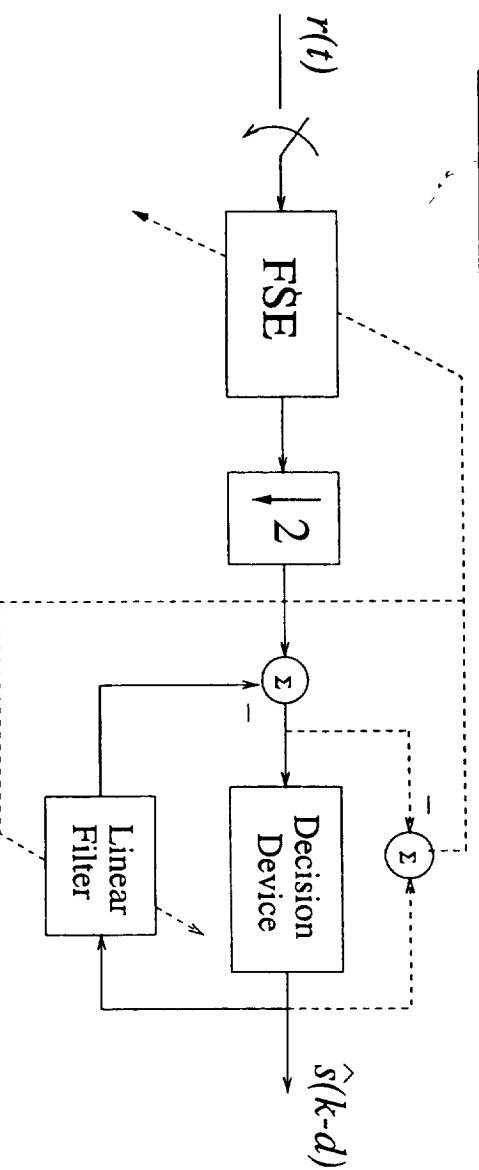


- Feedforward equalizers suffer noise enhancement
- Linear feedforward equalizers can't correct for non-linearities

### **Adaptive Non-linear Feedforward Eq.**

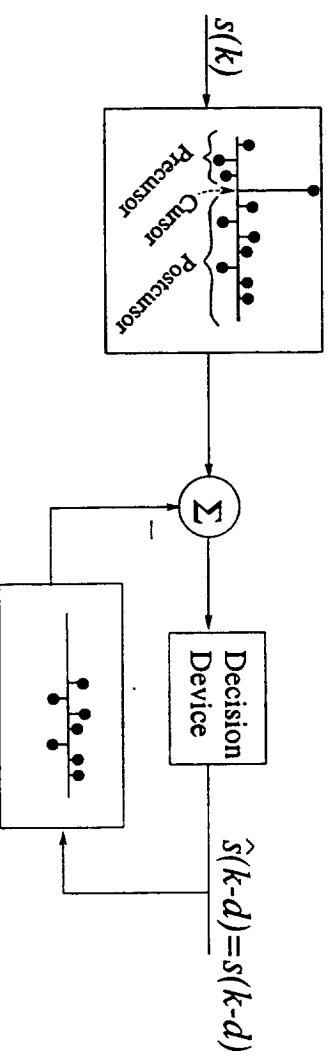
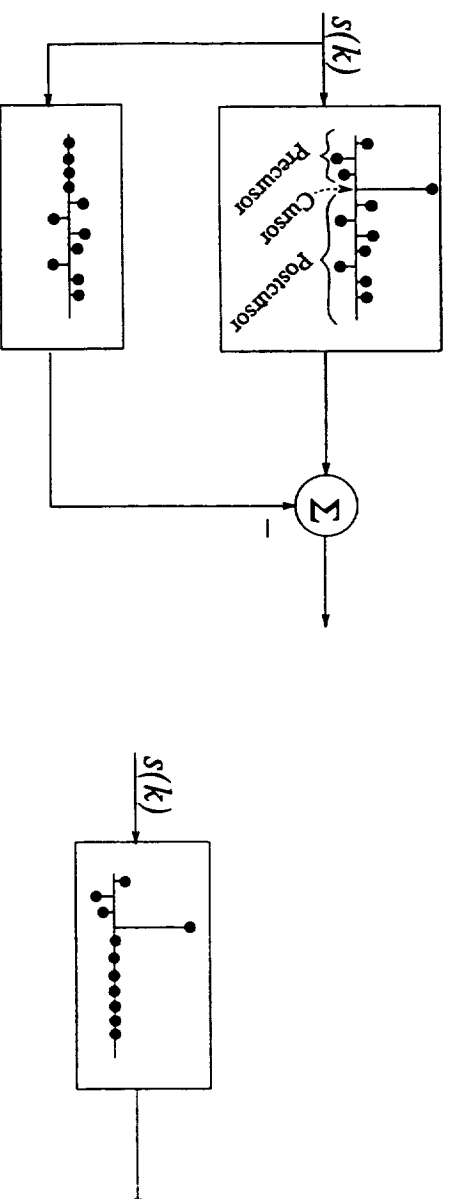
- Feedforward Non-linear equalizers may be implemented (for example, Volterra filters)
- These will exhibit non-linear noise enhancement

## Decision Feedback Equalizer (DFE)



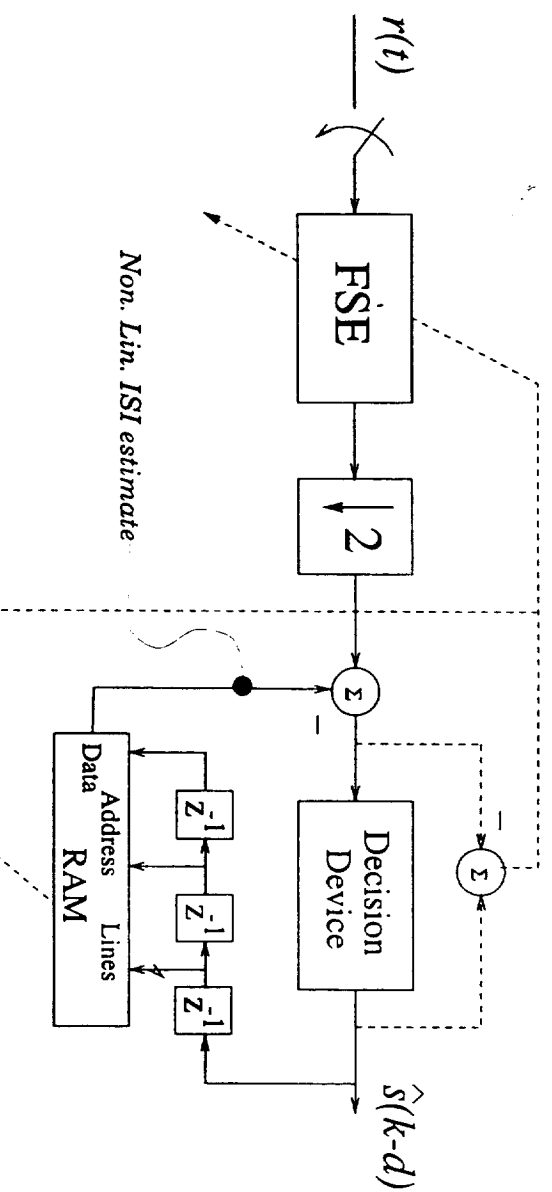
- DFEs do *not* suffer noise enhancement
- DFEs can cancel *only* post-cursor ISI (use with feedforward FSE)
- DFEs suffer from error propagation
- Linear feedback eq. can't correct for non-linearities
- Overall non-linearity makes analysis difficult

## DFE Attacks Postcursor

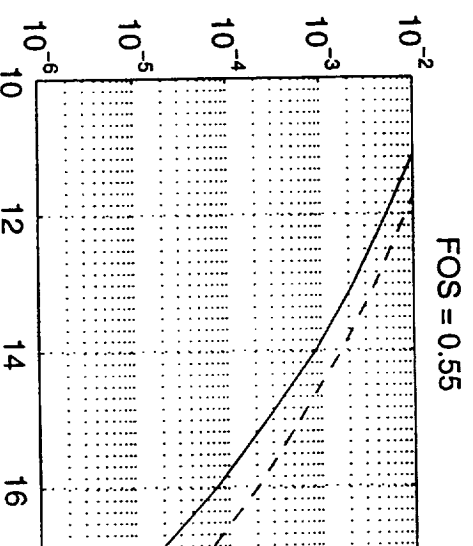
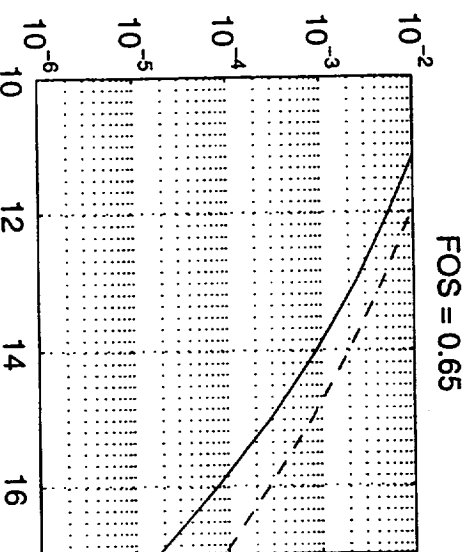
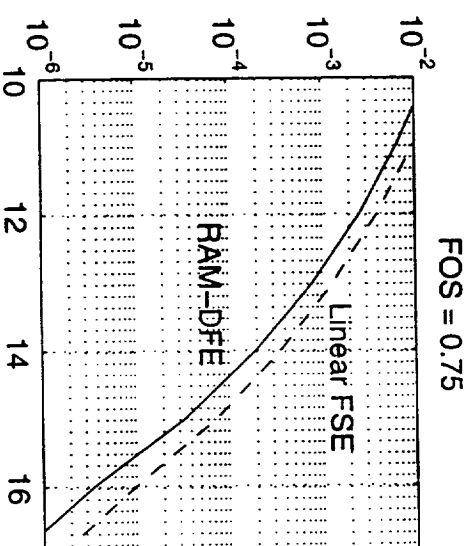
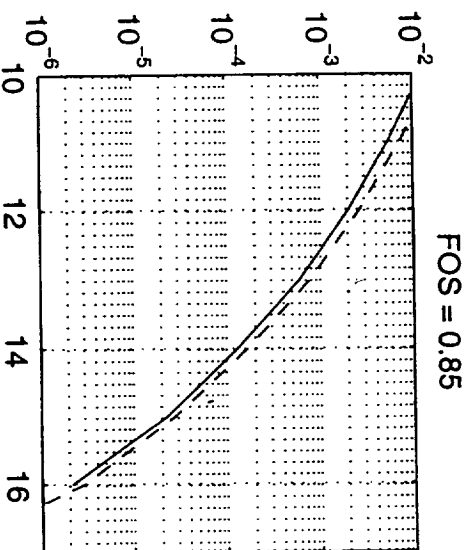


- DFEs can eliminate postcursor without noise enhancement ...
- ... if decision device output is correct

## RAM-DFE



- RAM-DFEs *can* correct for non-linearities
- RAM-DFEs can be implemented without multipliers
- RAM-DFEs typically converge slowly
- Overall non-linearity makes analysis difficult



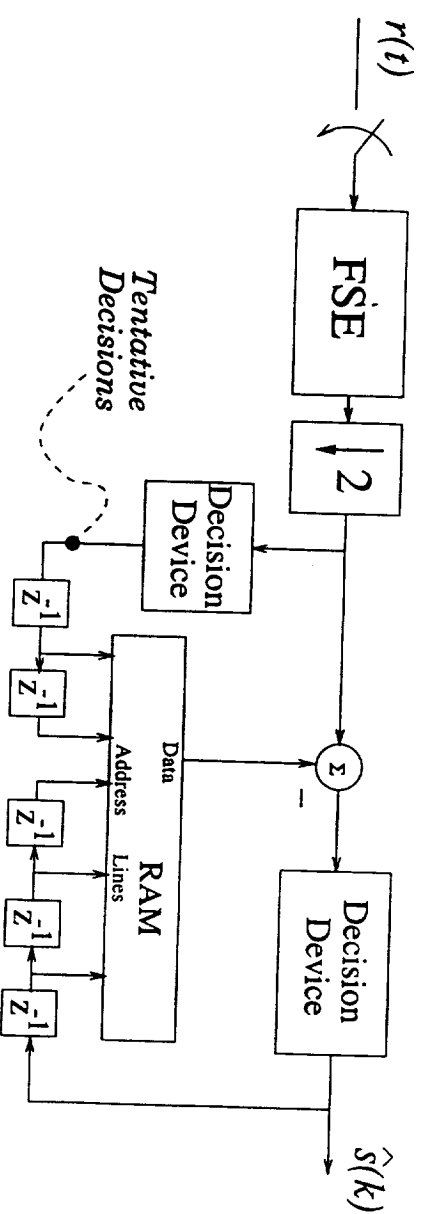
- Cutoff of Butterworth Filters is FOS (Fraction of Symbol Rate)
- 8-PSK Symbol Errors vs. SNR ( $E_s/N_0$ )

## Results

RAM-DFE performance over Linear FSE,

- is modest for low symbol rates
- improves for higher symbol rates
- *may* be further improved through use of a “RAM canceler”

## RAM Canceled



- Use of *tentative decisions* allows reduction of *precursor* non-linear ISI
- Introduces new error mechanism
- Overall performance impact under investigation



## Conclusions

- Increase in data rate through TDRSS *is* achievable using a combination of higher order modulation schemes and higher symbol rates.

## Specifically...

- Use of non-linear DFE (RAM-DFE)
  - mitigate noise enhancement,
  - reducing both linear and non-linear postcursor ISI.
- Linear Fractionally Space Equalizer
  - reduces linear precursor ISI.
- RAM-canceler with tentative decisions
  - reduces precursor nonlinear ISI components